

Third Semester B.E. Degree Examination, Dec 2020/Jan 2021

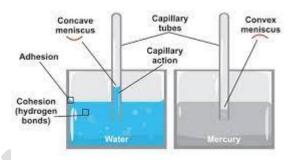
18CV33: Fluid Mechanics

MODULE 1 Define the following and mention their units: (i) Capillarity (ii) Surface tension (iii) Viscosity (06 marks)

(i) Capillarity

1a

The tendency of a liquid in a capillary tube or absorbent material to rise or fall as a result of surface tension. The more narrow the tube higher will be the capillarity rise/fall. This is due to the surface tension and properties of the medium. The surface of the fluid will be concave or convex surface based on the predominant forces in the fluid. If adhesive force is predominant, the fluid will wet the medium and water will have a concave surface. If cohesive forces are predominant, the fluid will be attracted towards itself, and the meniscus will take a concave surface.



shutterstock.com - 1643064859

Capillary rise or fall is given as

$$h = \frac{4\sigma\cos\theta}{\rho gd}$$

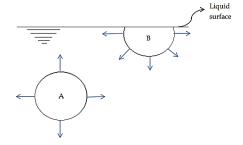
where d be the diameter of the glass tube in m, h be the capillary rise in m, σ is the surface tension of the liquid in N/m, θ is the angle of contact between liquid and glass tube, ρ is the density and g is the acceleration due to gravity

(ii) Surface tension

Surface tension is defined as the tensile force acting on the surface of a liquid in contact with gas or on the surface between two immiscible liquids such that the

contact faces behaves like a membrane under tension. Surface tension is due to unbalanced cohesive forces at the interface of liquid, gases or between two immiscible liquids. Unit of the surface tension is N/m

When a liquid possess, relatively, greater affinity for solid molecules or the liquid which have greater adhesion than



cohesion, then it will wet a solid surface in contact and tend to rise at the point of contact.

(iii) Viscosity

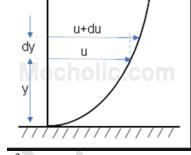
Viscosity is defined as the property of the fluid which offers resistance to the movement of one layer of fluid to the movement of adjacent layer of fluid.

Consider two layers of fluid separated by a distance of u moving at velocities v and u+du as shown.

Movement of these fluid layers induces shear stress and the shear stress is proportional to the rate of change of velocity with respect to y. Mathematically it can be written as

$$\tau \propto \frac{du}{dy}$$
Or $\tau = \mu \frac{du}{dy}$

Constant of proportionality is known as coefficient of dynamic viscosity and (du/dy) is known as velocity gradient/rate of shear strain/rate of shear deformation. Or, ratio of shear stress to shear rate is a constant, for a given temperature and pressure, and is defined as



the viscosity or coefficient of viscosity. Unit of μ is Ns/m² and CGS unit for the same is Poise.

Derive an expression for capillary rise/fall of fluid in a tube of small diameter with 1b sketches. (06 marks)

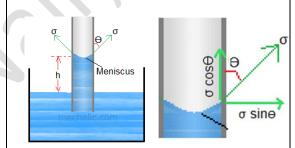
For capillary rise:

Let d be the diameter of the glass tube h be the capillary rise

 σ is the surface tension of the liquid θ is the angle of contact between liquid and glass tube

Forces acting in vertical direction are:

- 1. Weight of the liquid column
- 2. Vertical component of surface tension force



Weight of the liquid column

$$=\frac{\pi}{4} \times d^2 \times h \times \rho g$$
____(1

Vertical component of surface tension = $\sigma \times \text{Cos}\theta \times \pi d$ ____(2)

$$\frac{\pi}{4} \times d^2 \times h \times \rho g = \sigma \times \text{Cos}\theta \times \pi d$$

$$h = \frac{4\sigma\cos\theta}{\rho gd}$$

For capillary fall:

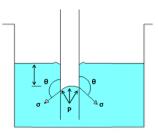
Hydrostatic force =
$$\frac{\pi}{4} \times d^2 \times p$$

= $\frac{\pi}{4} \times d^2 \times \rho g h$ _____(1)
Downward component of surface tension

$$= \sigma \times Cos\theta \times \pi d \underline{\hspace{1cm}} (2)$$

$$\frac{\frac{\pi}{4} \times d^2 \times h \times \rho g = \sigma \times \text{Cos}\theta \times \pi d}{h = \frac{4\sigma \cos \theta}{\rho g d}}$$

$$h = \frac{4\sigma\cos\theta}{\rho gd}$$



A 100 mm diameter cylinder rotates concentrically inside a 105 mm diameter fixed 1c cylinder. The length of both cylinder is 250 mm. Find the viscosity of the liquid that fills the space between the cylinders, if a torque of 1 Nm is required to maintain a rotating seed of 120 rpm. (08 marks)

Given:

$$T = F \times distance$$

 $1 = F \times r$
 $12 = F \times 0.05$
 $or F = 240 N$

$$\frac{F}{A} = \mu \frac{du}{dy}$$

$$\tau = \frac{F}{A} = \frac{240}{\pi \times 0.1 \times 0.25}$$

$$\tau = 3055.77 \text{ N/m}^2$$

$$v = r\omega$$

$$v = r\omega$$

$$v = 0.050 \times \frac{2 \times \pi \times 120}{60} = 0.628 \, m/s$$

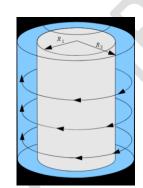
$$\tau = \mu \times \frac{du}{dy}$$

$$3055.77 = \mu \times \frac{0.628}{2.5 \times 10^{-3}}$$

$$\mu = 12.16 \text{ Ns/m}^2 = 121.6 \text{ Poise}$$

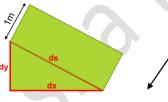
Given: $dy = 2.5 \times 10^{-3} \,\mathrm{m}$ T = 1 NmN = 120 rpm

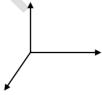
L = 0.25 m

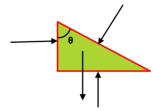


State and prove Pascal's law for the intensity of pressure at a point in a static fluid 2a 06 marks)

Pascal's law: It states that the intensity of pressure at a point in a static fluid is equal in all directions. vThe forces acting on the triangular wedge is shown below:







Weight of the wedge =
$$\rho g \times \frac{1}{2} \times (dx.dy.1)$$

Need to resolve the forces in horizontal and vertical directions.

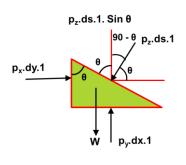
Resolving the forces in X direction:

$$p_x$$
. dy. $1 - p_z$. ds . $Cos\theta = 0$

$$p_x. \text{ dy. } 1 - p_z. \text{ ds. } \frac{dy}{ds} = 0$$

$$p_x$$
. dy = p_z . dy

$$p_x = p_z$$



Weight of the wedge = $\rho g \times 1/2 \times (dx.dy.1)$

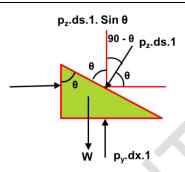
Need to resolve the forces in horizontal and vertical directions.

Resolving the forces in Y direction:

$$p_{y}. dx. 1 - p_{z}. ds. Sin\theta - \rho g \times \frac{1}{2} \times (dx. dy. 1) = 0$$

$$p_{y}. dx. 1 - p_{z}. ds. \frac{dx}{ds} - \rho g \times \frac{1}{2} \times (dx. dy. 1) = 0$$

$$p_{y}. dx. 1 - p_{z}. dx - \rho g \times \frac{1}{2} \times (dx. dy. 1) = 0$$



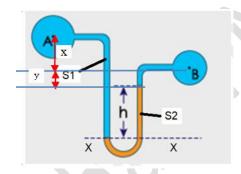
$p_{v} = p_{z} = p_{x}$

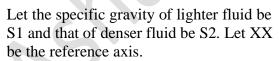
Since the element is very small (dx.dy) will be very small and can be neglected This infers that fluid pressure is same in all directions

Derive an expression for difference in pressure between two points using a U-tube differential manometer (08 marks)

Both manometers are used for measuring pressure difference between two points. Differential U tube manometer contains a manometric fluid which is denser than the fluid flowing in pipe.

Inverted U tube manometer is same in construction as like differential manometer but one major difference is that the manometric fluid is lighter than the fluid flowing through pipe because this manometer is mounted inverted in between points for measuring the pressure difference. This is used for measuring the pressure difference between underground pipe lines





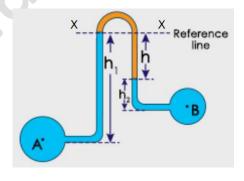
Equating the pressures for both limbs
$$P_A + S1\rho gx + S1\rho gy + S1\rho gh$$

$$= P_B + S1\rho gx + S2\rho gh$$

$$P_A + S1\rho gx + S1\rho gy + S1\rho gh$$

$$= P_B + S1\rho gx + S2\rho gh$$

$$P_A - P_B = (S2 - S1)\rho gh - S1\rho gy$$



Let the specific gravity of lighter fluid be S1 and that of denser fluid be S2. Let XX be the reference axis.

Equating the pressures for both limbs $P_A - S1\rho gh1 = P_B - S1\rho gh2 + S2\rho gh$ $P_A - P_B = S2\rho gh - S1\rho gh2 + S1\rho gh1$

Determine the pressure intensity at the bottom of a tank fille with an oil of specific gravity 0.7 to a height of 10 m. (06 marks)

Pressure intensity = $S \rho g h = 0.6 = 0.7 \times 1000 \times 9.81 \times 10 = 68.67 \ kPa$

MODULE 2

3a | Define: (i) Total pressure (ii) Center of pressure (04 Marks)

The total pressure is defined as the force exerted by a static fluid on a surface (either plane or curved) when the fluid comes in contact with the surface. This force is always

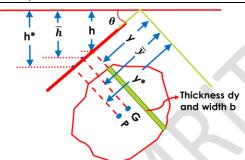
normal to the surface. The centre of pressure is defined as the point of application of the resultant pressure on the surface.

Derive an expression for total pressure and center of pressure for an inclined plane surface submerged in a liquid. (08 marks)

From the figure:

$$Sin\theta = \frac{\overline{h}}{\overline{y}} = \frac{h^*}{y^*} = \frac{h}{y}$$

$$h = ySin\theta$$



Total pressure:

Pressure intensity on the strip = ρgh

 $Area\ of\ strip = bdy$

Total force on the strip = ρgh . $bdy = \rho g$. dAh

Total force on the whole $surface = \int \rho g. \, dAh$

Total pressure:

Total force on the whole $surface = \int \rho g. \, dAh$

Total force on the whole $surface = \int \rho g. \, dAySin\theta = \rho gSin\theta. \int dAy$

$$= \rho g Sin\theta. A \frac{\bar{h}}{Sin\theta} = \rho g A \bar{h}$$

Total force on the whole surface = $\rho g A \bar{h}$

Centre of pressure:

Moment of force F about $0 - 0 = F.y^*$ ____(2)

Moment of force dF about $0 - 0 = \rho gh.bdy.y$

$$= \rho gySin\theta. dA. y = \rho gSin\theta. dA. y^2$$

Moment of force about $0 - 0 = \int \rho g Sin\theta \, dA \, y^2$

$$\int dA.\,y^2 = I_0$$

$$F. y^* = \rho g Sin \theta. I_0$$

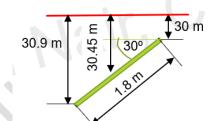
$$y^* = \frac{\rho g Sin\theta. I_0}{F} = \frac{\rho g Sin\theta. I_0}{\rho g A \overline{h}}$$

$$y^* = \frac{Sin\theta. I_0}{A\overline{h}}$$

	$\frac{h^*}{Sin\theta} = \frac{Sin\theta. I_0}{A\overline{h}}$
	$h^* = \frac{Sin^2\theta}{A\bar{h}}[I_G + A\bar{y}^2]$
	$h^* = rac{Sin^2 heta}{Aar{h}}igg[I_G + Arac{ar{h}^2}{Sin^2 heta}igg]$
	$h^* = \left[rac{I_G Sin^2 heta}{A ar{h}} + ar{h} ight]$
3c	A 1200 mm 1800 mm size rectangular plate is immersed in water with an inclination of 30° to the harizontal. The 1200 mm side of the plate is kent

A 1200 mm 1800 mm size rectangular plate is immersed in water with an inclination of 30° to the horizontal. The 1200 mm side of the plate is kept horizontal at a depth of 30 m below the water surface. Compute the total pressure on the surface and the position of center of pressure. (08 marks)

$$A = 1.2 \times 1.8 = 2.16 m^2$$
 $I_G = \frac{1.2 \times 1.8^3}{12} = 0.58 m^4$



 $Total\ pressure = \rho g A \overline{h}$ = 9810 × 2.16 × 30.45 $Total\ pressure = 645.22\ kN$

$$h^* = \left[\frac{I_G Sin^2 \theta}{A \overline{h}} + \overline{h} \right]$$

$$h^* = \left[\frac{0.58 \times 0.5^2}{2.16 \times 30.45} + 30.45 \right]$$

$$h^* = 30.452 m$$

- 4a **Differentiate between:**
 - (i) Uniform and non-uniform flow,

(ii) Steady and unsteady flow (04 Marks)

		11
	1	Uniform and non- uniform
		flow

Type of flow in which the velocity at any given point of time does not change with respect to space (along flow direction) is uniform flow.

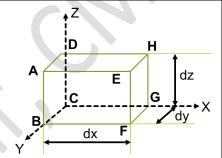
$$\left[\frac{\partial V}{\partial s}\right]_{t=constant} = 0;$$

Type of flow in which the velocity at any given point of time changes with respect to space (along flow direction) is non uniform flow.

$$\left[\frac{\partial V}{\partial s}\right]_{t=constant} \neq 0$$

Type of flow in which the fluid characteristics like velocity pressure, density etc at a point do not change with time is sterilized flow. Steady and unsteady flow Steady flow $ \begin{bmatrix} \frac{\partial V}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix} \frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\ \end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, y_0, z_0} = 0; \qquad \begin{bmatrix}\frac{\partial \rho}{\partial t} \\\\\\\\\end{bmatrix}_{x_0, $	eady ,z ₀ fluid point
---	-----------------------------------

46 Derive continuity equation for a three-dimensional flow in Cartesian coordinates. (08 Marks)



Let u, v and w be the velocity components in x, y and z directions.

Mass of fluid entering the face ABCD per second = ρu . Area of ABCD Mass of fluid entering the face ABCD per second = $\rho u.(dy.dz)$ Mass of fluid leaving the face EFGH per second

$$= \rho u. (dy. dz) + \frac{\partial (\rho u. (dy. dz). dx}{\partial x}$$

Gain of mass in
$$x$$
 – direction = $\rho u.(dy.dz) - \rho u.(dy.dz) - \frac{\partial (\rho u.(dy.dz).dx)}{\partial x}$

Gain of mass in
$$x$$
 – direction = $-\frac{\partial(\rho u). dx. dy. dz}{\partial x}$

Gain of mass in
$$x$$
 – direction = $-\frac{\partial(\rho u). dx. dy. dz}{\partial x}$
Gain of mass in y – direction = $-\frac{\partial(\rho v). dx. dy. dz}{\partial y}$
Gain of mass in z – direction = $-\frac{\partial(\rho w). dx. dy. dz}{\partial z}$

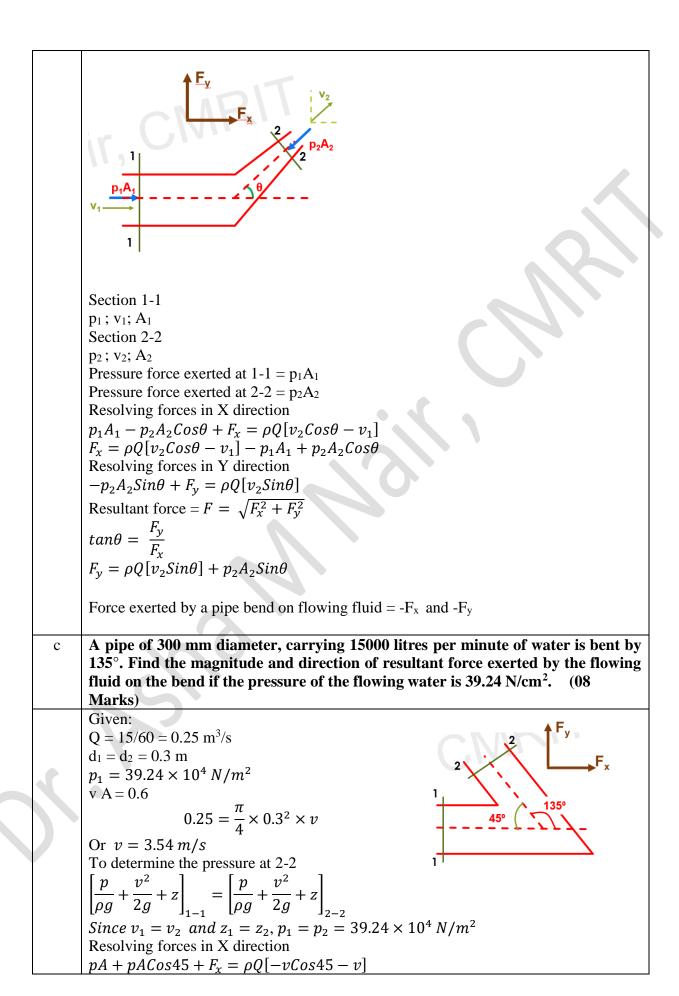
Gain of mass in
$$z$$
 – direction = $-\frac{\partial(\rho w). dx. dy. dz}{\partial x}$

Net gain of masses =
$$-\left[\frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z}\right] \cdot dx \cdot dy \cdot dz$$

Net gain in mass = rate of increase of mass of fluid in the element which is given as

$$\begin{split} &\frac{\partial \rho}{\partial t}.(dx.dy.dz) \\ &-\left[\frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z}\right].dx.dy.dz = \frac{\partial \rho}{\partial t}.(dx.dy.dz) \\ &-\left[\frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z}\right] = \frac{\partial \rho}{\partial t} \\ &\frac{\partial \rho}{\partial t} + \frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z} = 0 \; (General \; form) \end{split}$$

	$\begin{bmatrix} \partial \rho & \partial(\rho u) & \partial(\rho v) & \partial(\rho w) \end{bmatrix}_{0}$
	$\frac{\partial \rho}{\partial t} + \frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z} = 0$
	For steady state, $\frac{\partial \rho}{\partial t} = 0$; $\frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z} = 0$
	For incompressible flow, ρ is a constant
	$\begin{bmatrix} \partial u & \partial v & \partial w \\ 0 & 0 \end{bmatrix}$
	$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0$
	For 2D flow
	$\frac{\partial u}{\partial u} = \frac{\partial v}{\partial v}$
	$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$
4c	Evaluate stream function ψ and compute velocity of flow, V, for a two-dimensional
	flow field given by, $u = 4x^3$ and $v = -12x^2y$ at point (1, 2). Assume $\psi = 0$ at point (0,
	0).
	$v = \frac{\partial \psi}{\partial x} = -12x^2y \qquad (1)$ $-u = \frac{\partial \psi}{\partial y} = -4x^3 \qquad (2)$
	$v = \frac{r}{\partial x} = -12x^2y \tag{1}$
	$\frac{\partial x}{\partial u}$
	$-u = \frac{3\psi}{2} = -4x^3$ (2)
	oy (1)
	Integrating (1) wrt x
	$\psi = \frac{-12x^3y}{3} + c1 (3)$
	$\psi = \frac{1}{3} + c1 (3)$
	Differentiating (3) wrt y
	$\frac{\partial \psi}{\partial y} = \frac{-12x^3}{3} + \frac{\partial c1}{\partial y} = -4x^3 (4)$
	Comparing (4) with (2),
	$\frac{\partial c1}{\partial c} = 0$
	$\frac{\partial}{\partial y} = 0$
	Or c1 = y
	Hence
	$\psi = -4x^3y + y$
	At (1,2), $\psi = -4x^3y + y = -4 \times 1 \times 2 + 2 = -6$ MODULE 3
<u> </u>	
5a	State Impulse Momentum principle. Give fields where it is applied. (04
	Marks)
	It states that the impulse of a force F acting on a fluid of mass m in short time dt is
	equal to the rate of change of momentum in the direction of force. This is based on
	Newton's second law which states that the net force acting on a fluid mass is equal
	to the change in momentum of flow per unit time in that direction.
	F = ma
	dv
	But acceleration, $a = \frac{dv}{dt}$
	d(mv)
	$F = \frac{d(mv)}{dt} $ (1)
	F. dt = d(mv) (2)
	$\frac{1 \cdot \omega - \omega (n \omega)}{2}$
	Used to determine the resultant force exerted by a flowing fluid in a pipe bend. Used
	to estimate friction loss in pipes.
5b	Derive an expression for force exerted by a fluid on a pipe bend. (08 Marks)
טט	Derive an expression for force exerted by a finite on a pipe bend. (vo Marks)



$$39.24 \times 10^4 \times 0.071[1 + Cos45] + F_x = -1000 \times 0.25 \times 3.54[1 + Cos45]$$

 $F_x = -1510.79 - 47560.7 = -49.071 \text{ kN}$

Resolving forces in Y direction

$$-pASin\theta + F_{v} = \rho Q[v_{2}Sin\theta]$$

 $F_y = 1000 \times 0.25[3.54Sin45] + 39.24 \times 10^4 \times 0.071Sin45$

 $F_{\rm v} = 19.074 \, kN$

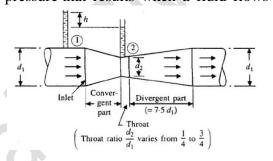
Resultant force = $F = \sqrt{F_x^2 + F_y^2} = \sqrt{49.071^2 + 19.074^2} = 52.64 \, kN$ $tan\theta = \frac{F_y}{F_x} = \frac{19.074}{-49.071}$

$$tan\theta = \frac{F_y}{F_x} = \frac{19.074}{-49.071}$$

$$\theta = 21.24^{\circ}$$

What is venture effect? Derive an expression for discharge through a 6a venturineter.(08 Marks)

The Venturi effect is the reduction in fluid pressure that results when a fluid flows through a constricted section (or choke) of a pipe. The Venturi effect is the reduction in fluid pressure that results when a fluid flows through a constricted section (or choke) of a pipe. The effect utilizes both the principle of continuity as well as the principle of conservation of mechanical energy.



Applying Bernoulli's equation between 1-1 and 2-2

$$\left[\frac{p}{\rho g} + \frac{v^2}{2g} + z\right]_{1-1} = \left[\frac{p}{\rho g} + \frac{v^2}{2g} + z\right]_{2-2}$$

$$\mathbf{z}_1 = \mathbf{z}_2$$

$$\left[\frac{p_1}{\rho g} + \frac{v_1^2}{2g}\right]_{1-1} = \left[\frac{p_2}{\rho g} + \frac{v_2^2}{2g}\right]_{2-2}$$

$$\frac{p_1 - p_2}{\rho g} = \frac{v_2^2 - v_1^2}{2g}$$

From the figure,

$$\frac{p_1 - p_2}{\rho g} = h = \frac{v_2^2 - v_1^2}{2g}$$

Or $v_2^2 - v_1^2 = 2gh$

Or
$$v_2^2 - v_1^2 = 2gh$$

$$a_1v_1 = a_2v_2$$

$$v_1 = \frac{a_2}{a_1} v_2$$

$$a_1 v_1 = a_2 v_2$$

$$v_1 = \frac{a_2}{a_1} v_2$$
Or $v_2^2 - v_1^2 = 2gh$

$$v_2^2 - \left[\frac{a_2}{a_1}\right]^2 v_2^2 = 2gh$$

$$v_2^2[a_1^2 - a_2^2] = a_1^2 2gh$$

$$v_2^2[a_1^2 - a_2^2] = a_1^2 2gh$$

$$v_2^2 = \frac{a_1^2}{[a_1^2 - a_2^2]} 2gh$$

$$v_2 = \sqrt{\frac{a_1^2}{[a_1^2 - a_2^2]} 2gh}$$

b	A pitot tube fixed in a pipe of 300 mm diameter is used to measure the velocity and rate of flow. If the stagnation and static pressure heads are 6.0 m and 5.0 m respectively, compute the velocity and rate of flow. Assume Cv= 0.98 for the pitot tube. (06 Marks)
	h = 6 - 5 = 1 m
	$c_{\rm v} = 0.98$
	$v = c_v \sqrt{2gh}$
	$v = 0.98 \times \sqrt{2 \times 9.81 \times 1} = 4.34 m/s$
	$Q = a v = \frac{\pi}{4} \times 0.3^2 \times 4.34 = 0.31 m^3/s$
С	A 20 cm x 10 cm venturimeter is used to measure the flow of water in a horizontal pipe. The pressure at the inlet of venturimeter is 17.658 N/cm ² and the vacuum pressure at the throat is 30 cm of mercury. Find the discharge of water through
	the venturimeter assuming Cd=0.98. (06 Marks)
	Given: $d_1 = 0.2 \text{ m}; d_2 = 0.1 \text{ m}$ $c_d = 0.98$
	$Q_{act} = c_d a_1 a_2 \sqrt{\frac{2gh}{[a_1^2 - a_2^2]}}$
	$a_1 = \frac{\pi}{4} \times 0.2^2 = 31.42 \times 10^{-3} m^2$ $a_2 = \frac{\pi}{4} \times 0.1^2 = 7.85 \times 10^{-3} m^2$
	$\sqrt{a_1^2 - a_2^2} = \sqrt{31.42 \times 10^{-3^2} - 7.85 \times 10^{-3^2}} = 30.42 \times 10^{-3} m^2$
	$a_1 a_2 = 31.42 \times 10^{-3} \times 7.85 \times 10^{-3} = 246.65 \times 10^{-6} m^2$
	$\frac{p_1}{\rho g} - \frac{p_2}{\rho g} = \frac{17.658 \times 10^4}{9810} + \frac{0.30 \times 13.6 \times 9810}{9810}$
	$\begin{array}{cccc} \rho g & \rho g & 9810 & 9810 \\ n_t & n_s & \end{array}$
	$\frac{p_1}{q_2} - \frac{p_2}{q_3} = 22.08 m$
	$\rho g \rho g$
	$Q_{act} = c_d a_1 a_2 \sqrt{\frac{2gh}{[a_1^2 - a_2^2]}}$
	19 62 × 22 08
	$Q_{act} = 0.98 \times 246.65 \times 10^{-6} \times \sqrt{\frac{19.62 \times 22.08}{30.42 \times 10^{-3}}}$
1	$Q_{act} = 28.85 \times 10^{-3} \text{ m}^3/\text{s}$
	MODULE 4
7a	Define hydraulic coefficients for an orifice and give the relation between them. (06 Marks)
	Hydraulic coefficients are:
	Coefficient of velocity, c _v
	It is the ratio of the velocity of the jet at the vena contracta to the theoretical
	velocity of the jet.
	Coefficient of contraction, cc It is the ratio of the area of the jet at the year contracts to the area of the crifice
	It is the ratio of the area of the jet at the vena contracta to the area of the orifice.

$$c_c = \frac{a_2}{a_0}$$

> Coefficient of discharge, cd

It is the ratio of the actual discharge from an orifice to theoretical discharge from orifice.

$$c_d = \frac{Q_{act}}{Q_{th}}$$

Prove that
$$[c_c \times c_v] = c_d$$

$$c_d = \frac{Q_{act}}{Q_{th}}$$

$$Q_{act} = c_d \times Q_{th} = a_{act} \times v_{act}$$

$$Q_{act} = c_d \times Q_{th} = [c_c \times a_{th}] \times [c_v \times v_{th}]$$

$$Q_{act} = c_d \times Q_{th} = [c_c \times c_v] \times [v_{th} \times a_{th}]$$

$$[c_c \times c_v] \times [v_{th} \times a_{th}] = [c_c \times c_v] \times Q_{th} = c_d \times Q_{th}$$

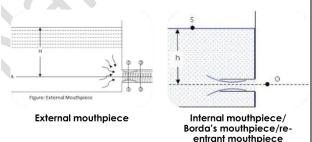
Hence
$$[c_c \times c_v] = c_d$$

b Give classification of mouth pieces with suitable sketches. (06 Marks)

Mouthpiece - It is the short length of a pipe which is two to three times its diameter in length, fitted in a tank/vessel containing the fluid. It is used to measure the rate of flow of fluid

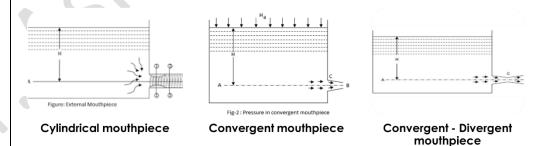
Mouthpieces are classified on the basis of their position with respect to the tank

On the basis of their position with respect to the tank or vessel to which they are fitted, mouthpieces are classified as external mouthpieces and internal mouthpieces.



Mouthpieces are classified on the basis of their shape

On the basis of their shape, mouthpieces are classified as cylindrical mouthpieces, convergent mouthpieces and convergent-divergent mouthpieces.

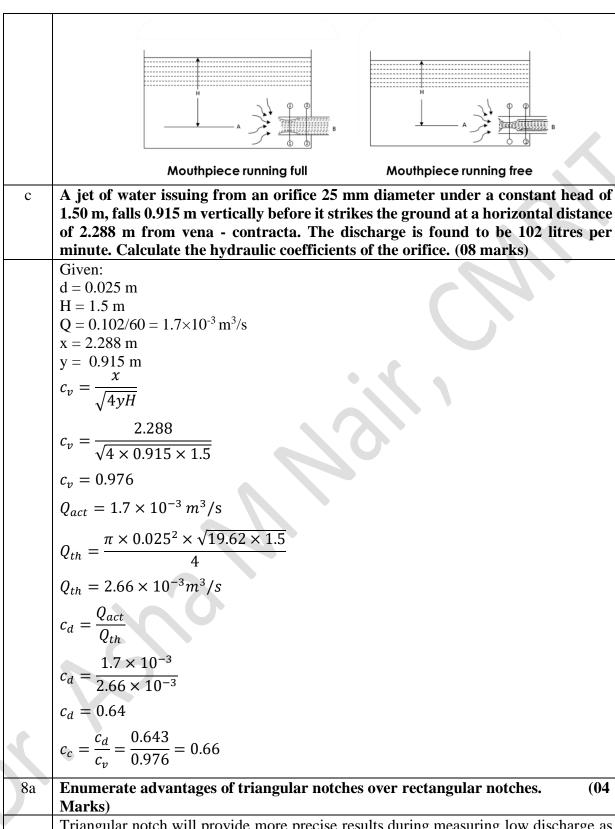


Mouthpieces are classified on the basis of nature of discharge

On the basis of nature of discharge at the outlet of mouthpiece, mouthpieces are classified as mouthpieces running full and mouthpieces running free.

If the jet of liquid after contraction does not touch the sides of mouthpiece, mouthpiece will be termed as running free.

If the jet of liquid after contraction expands and fills the whole mouthpiece, mouthpiece will be termed as running full.



Triangular notch will provide more precise results during measuring low discharge as compare to result obtained from rectangular notch. This is due to the space for the water to flow through the notch. Only one reading of H will be required for determination of

discharge in case of triangular notch. For rectangular notch, the space of water it is

constant for every depth.

b	Derive the expression for discharge through a triangular notch. (08
	Marks)
	$\tan(\theta/2) = \frac{AC}{QC} = \frac{AC}{H-h}$
	$AC = (H - h) \cdot \tan \theta / 2$
	Width of the strip = $2(H - h)$. tan $\theta/2$
	Area of strip = $2(H - h)$. tan $\theta/2$.dh
	Theoretical velocity = $\sqrt{2gh}$
	$dQ = c_d \times a_{th} \times v_{th}$
	$dQ = c_d \times (2(H - h) \cdot \tan \theta / 2 \cdot dh) \sqrt{2gh}$
	$dQ = c_d \times (2(H - h). \tan \theta / 2 . dh) \sqrt{2gh}$
	$\int dQ = \int c_d \times (2(H - h)) \tan \theta / 2 \cdot (dh) \sqrt{2gh}$
	$\int dQ = 2c_d \times \tan \theta / 2 \times \sqrt{2g} \times \int \left[H h^{\frac{1}{2}} - h^{\frac{3}{2}} \right] dh$
	$Q = 2c_d \times \tan \theta / 2 \times \sqrt{2g} \times \left[\frac{Hh^{3/2}}{3/2} - \frac{h^{5/2}}{5/2} \right]_0^H$
	$Q = 2c_d \times \tan \theta / 2 \times \sqrt{2g} \times \left[\frac{H^{5/2}}{3/2} - \frac{H^{5/2}}{5/2} \right]_0^H$
	$Q = 2c_d \times \tan \theta / 2 \times \sqrt{2g} \times \frac{2 \times 2 \times H^{5/2}}{15}$
	$Q = \frac{8}{15} c_d \times \tan \theta / 2 \times \sqrt{2g} \times H^{5/2}$
	For a right angled notch, $\theta = 90^{\circ}$, $c_d = 0.6$
	$\tan \theta/2 = 1$
	$Q = 1.417 \times H^{5/2}$
С	A river 60 m wide has vertical banks and 1.50 m depth of flow. The velocity of flow is 1.20 m/s. A broad crested weir 2.40 m high is constructed across the river.
	Find the head on the weir crest considering the velocity of approach. Assume Cd
	- 0.90. (08 Marks)
	Given:
	L = 60 m H = 1.5 m
	$c_{\rm d} = 0.9$
	$A = 60 \times 1.5 = 90 \text{ m}^2$
	$Q = 90 \times 1.2 = 108 \ m^3/s$
	Without considering velocity of approach
4	$Q_{act} = 1.705 \times c_d \times L \times H^{3/2}$
	$108 = 1.705 \times 0.9 \times 60 \times H^{3/2}$
	$H = 1.11 \mathrm{m}$
	By considering velocity of approach
	$h_a = \frac{v_a^2}{2g} = \frac{1.2^2}{19.62} = 0.073 m$
	$108 = 1.705 \times 0.9 \times 60 \times \left[((H + h_a)^{3/2}) - (h_a)^{\frac{3}{2}} \right]$
	$108 = 1.705 \times 0.9 \times 60 \times \left[((H + 0.073)^{3/2}) - (0.073)^{\frac{3}{2}} \right]$

	$\left[((H + 0.073)^{3/2}) - (0.073)^{\frac{3}{2}} \right] = 1.17$
	Let $H = 1.05$
	LHS = RHS
	Hence H = 1.05 m
	MODULE 5
9a	Derive Darcy-Weisbach equation for head loss due to friction in a pipe. (08)
<i>γ</i> α	Marks)
	Consider uniform horizontal flow of nine
	Applying Bernoulli's equation between (1) and
	(2) Resistance to flow (F)
	$\left \frac{p}{\rho g} + \frac{v^2}{2g} + z \right _{1=1}$
	1 9
	$= \left[\frac{p}{\rho g} + \frac{v^2}{2g} + z \right]_{0.00} + loss$ Direction of flow
	$=\left \frac{1}{\rho g}+\frac{1}{2g}+2\right _{0}+toss$
	$z_1 = z_2 = 0; v_1 = v_2$
	$\begin{bmatrix} z_1 = z_2 = 0; v_1 = v_2 \\ \left[\frac{p_1}{\rho g} + \frac{v_1^2}{2g} + z_1 \right] = \left[\frac{p_2}{\rho g} + \frac{v_2^2}{2g} + z_2 \right] + Losses$
	$\left \frac{p_1}{p_1} + \frac{v_1}{2p_1} + z_1 \right = \left \frac{p_2}{p_2} + \frac{v_2}{2p_1} + z_2 \right + Losses$
	$\left[egin{array}{ccccc} ho g & 2g & floor & ho g & 2g & floor \end{array} ight]$
	$\frac{p_1}{\rho g} - \frac{p_2}{\rho g} = h_f$
	$p_1 - p_2 = h_f \rho g_{\perp}(1)$
	According to Froude
	Frictional force, F is given as
	F = frictional resistance/wetted area/unit velocity × wetted area × square of velocity
	According to Froude
	Frictional force, F is given as
	$F = \text{frictional resistance/wetted area/unit velocity} \times \text{wetted area} \times \text{square of velocity}$
	$F = f' \times \pi dL \times v^2$
	Considering equilibrium of forces:
	$p_1 a_1 - p_2 a_2 = F = f' \times \pi dL \times v^2$
	$p_1 - p_2 = \frac{f' \times \pi dL \times v^2}{A} $ (2)
	A = C
	Substituting (2) in (1) $f(x) = dx + x^2$
	$p_1 - p_2 = h_f \rho g = \frac{f' \times \pi dL \times v^2}{A}$
	D and A
	$\frac{P}{A} = \frac{\pi d}{\pi d^2 / 4} = \frac{4}{d}$
	$A \pi d^2/4 d$
	$h_c = \frac{f' \times 4L \times v^2}{}$
	$h_f = \frac{f' \times 4L \times v^2}{\rho g d}$ $\frac{f'}{\rho} = \frac{f}{2}$
	$\left \frac{f'}{f} - \frac{f}{f} \right $
	$\frac{1}{\rho} - \frac{1}{2}$
	$\int_{L} 4fL v^2$
	$h_f = \frac{4fL v^2}{2gd}$
	Called as Darcy Weisbach equation
b	List major and minor losses in a pipe flow. (04 Marks)
	Friction loss

	Darcy Weishbach eqn
	Chezy's equation
	Minor losses:
	Loss of head due to sudden enlargement
	Loss of head due to sudden contraction
	Loss of head at the entrance of a pipe
	➤ Loss of head at the exit of a pipe
	Loss of head due to sudden obstruction in a pipe
	➤ Loss of head due to bend in the pipe
	➤ Loss of head in various pipe fittings
c	Water is required to be supplied to a colony of 4000 residents at a rate of 180 litres
	per person from a source 3 km away. If half the daily requirement needs to be
	pumped in 8 hours against a friction head of 18 in, find the size of the main pipe
	supplying water. Assume friction factor as 0.028. (08 Marks)
	$Q = 4000 \times 0.180/(2 \times 8 \times 3600)$
	$Q = 12.5 \times 10^{-3} m^3 / s$
	$0.028 \times 3000 \times (12.5 \times 10^{-3})^2$
	$18 = \frac{0.028 \times 3000 \times (12.5 \times 10^{-3})^2}{2g \times \left(\frac{\pi}{4}\right)^2 \times d^5}$
	$2g \times \left(\frac{\pi}{4}\right) \times d^5$
	d = 0.143 m
10 a	What is an equivalent pipe? Derive an expression for diameter of an equivalent
	pipe.
	(08 Marks)
	This is defined as a pipe of uniform diameter having loss of head and discharge equal
	to the loss of head and the discharge of a compound pipe consisting of several pipes
	of different lengths and diameters. The uniform diameter of the equivalent pipe is
	called as Equiavlent Size of the Pipe.
	Neglecting minor losses, Total head loss = $\frac{4f_1L_1v_1^2}{2gd_1} + \frac{4f_2L_2v_2^2}{2gd_2} + \frac{4f_3L_3v_3^2}{2gd_3}$
	Neglecting minor losses, Total head loss = $\frac{1}{2ad_1} + \frac{1}{2ad_2} + \frac{1}{2ad_3} + \frac{1}{2ad_3}$
	$\begin{bmatrix} q - a_1v_1 - a_2v_2 - a_3v_3 \\ \pi \\ 1 - 2 \end{bmatrix} = \pi$
	$Q = a_1 v_1 = a_2 v_2 = a_3 v_3 Q = \frac{\pi}{4} \times d_1^2 \times v_1 = \frac{\pi}{4} \times d_2^2 \times v_2 = \frac{\pi}{4} \times d_3^2 \times v_3 Q$
	Q
	$v_1 = \frac{1}{\pi r}$
	$v_1 = \frac{Q}{\frac{\pi}{4} \times d_1^2}$
	$H = \frac{4f_1L_1Q^2}{2g \times \left(\frac{\pi}{4}\right)^2 \times d_1^5} + \frac{4f_2L_2Q^2}{2g \times \left(\frac{\pi}{4}\right)^2 \times d_2^5} + \frac{4f_3L_3Q^2}{2g \times \left(\frac{\pi}{4}\right)^2 \times d_3^5}$
	$2a \times \left(\frac{\pi}{2}\right)^2 \times d^5 2a \times \left(\frac{\pi}{2}\right)^2 \times d^5 2a \times \left(\frac{\pi}{2}\right)^2 \times d^5$
	(4)
	If f is uniform,
	6450 ² [x x x]
	$H = \frac{64fQ^2}{2g \times \pi^2} \left[\frac{L_1}{d_1^5} + \frac{L_2}{d_2^5} + \frac{L_3}{d_3^5} \right]$
	$2g \times \pi^2 \left[d_1^5 \cdot d_2^5 \cdot d_3^5 \right]$
	Considering an equivalent pipe of length L and diamter d
	$64fQ^2$ $\begin{bmatrix} L_1 & L_2 & L_3 \end{bmatrix}$ $64fQ^2$ $\begin{bmatrix} L \end{bmatrix}$
	$\left \frac{2g\times\pi^2}{d_1^5}\right \frac{d_2^5}{d_2^5}+\frac{1}{d_2^5}\right =\frac{2g\times\pi^2}{2g\times\pi^2}\left[\frac{1}{d_2^5}\right]$
	$\begin{bmatrix} L_1 & L_2 & L_3 & L \end{bmatrix}$
	$\frac{64fQ^2}{2g \times \pi^2} \left[\frac{L_1}{d_1^5} + \frac{L_2}{d_2^5} + \frac{L_3}{d_3^5} \right] = \frac{64fQ^2}{2g \times \pi^2} \left[\frac{L}{d^5} \right]$ $\frac{L_1}{d_1^5} + \frac{L_2}{d_2^5} + \frac{L_3}{d_3^5} = \frac{L}{d^5}$
	a_1 a_2 a_3 a_4 This equation is called as Dupit's equation
b	Explain phenomenon of water hammer in pipes. (04 Marks)
υ	Explain phenomenon of water nammer in pipes. (04 Marks)

Water hammer is a phenomenon that can occur in any piping system where valves are used to control the flow of liquids or steam. Water hammer is the result of a pressure surge, or high-pressure shockwave that propagates through a piping system when a fluid in motion is forced to change direction or stop abruptly.

The hammer effect (or water hammer) can harm valves, pipes, and gauges in any water, oil, or gas application. It occurs when the liquid pressure is turned from an on position to an off position abruptly. When water or a liquid is flowing at full capacity there is a normal, even sound of the flow.

Pressure rise is dependent upon:

- Velocity of flow of water in the pipe
- > Length of the pipe
- > Time taken to close the valve
- Elastic properties of the material of the pipe
- Water is flowing in a pipe of 150 mm diameter with a velocity of 2.5 m/s, when it is suddenly brought to rest by closing the valve. Find the pressure rise in the pipe assuming it to be elastic with $E = 206 \text{ GN/m}^2$ and Poisson's ratio of 0.25. The bulk modulus of water, $K = 206 \text{ GN/m}^2$. Thickness of pipe wall is 5 mm. (08 Marks)

Elastic pipe:

$$p = v \sqrt{\frac{\rho}{\left[\frac{D}{Et} + \frac{1}{K}\right]}}$$

$$p = 2.5 \sqrt{\frac{1000}{\left[\frac{0.15}{206 \times 10^9 \times 5 \times 10^{-3}} + \frac{1}{206 \times 10^9}\right]}}$$

$$= 2.5 \sqrt{\frac{1000}{\left[1.46 \times 10^{-10} + 4.85 \times 10^{-12}\right]}} = 6.44 MPa$$