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Internal Assessment Test 1 – Sept. 2019

Sub:	Energy Engineering				Sub Code:	15ME71	Branch:	Mechanical		
Date:	07/09/2019	Duration:	90 mins	Max Marks:	50	Sem / Sec:	7 A & B			OBE
<u>Answer all Questions</u>							MARKS	CO	RBT	
1	Derive an expression for power coefficient for a horizontal axis wind turbine.					[10]	CO4	L2		
2	An aero-generator, installed at sea shore generates an output of 1200W at wind speed of 6m/s at one atmosphere pressure and a temperature of 27°C. What will be the output, if the same aero-generator is installed on the top of a hill where the temperature is 15°C, pressure is 0.85 atmosphere and wind speed is 8 m/s.					[10]	CO4	L3		
3	Following data were measured for a HAWT: Speed of wind = 8 m/s at 1 atmosphere and 20°C, Diameter of rotor = 120m, Speed of rotor = 40 rpm. Calculate the maximum possible torque produced at the shaft.					[10]	CO4	L3		
4	Wind blow with a velocity of 15 m/s at 15°C and 1 standard atm. pressure. The turbine diameter is 120 m with operating speed 40 rpm at maximum efficiency. Propeller type wind turbine is considered. Calculate the following: (i) Total power density in the wind stream, (ii) The maximum obtainable power density, (iii) A reasonably obtainable power density, (iv) Total power, (v) Maximum axial thrust, (vi) Torque at maximum efficiency Assume $R = 0.287 \text{ kJ/kgK}$, $\eta = 35\%$.					[20]	CO4	L3		

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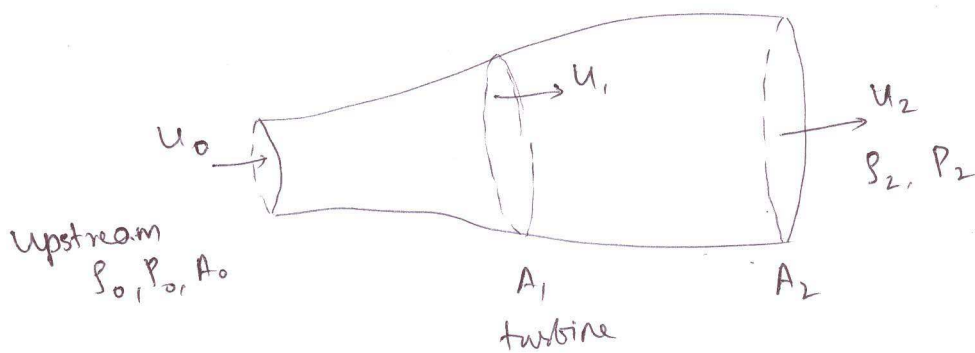
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1) Expression for power coefficient



The rotor of the turbine is replaced as disk (actuator disk theory). The control volume for analysis is shown in figure. The control volume here is a streamtube, and hence all the mass entering the control volume upstream converts downstream. i.e.

$$\dot{m} = \rho A_0 U_0 = \rho A_1 U_1 = \rho A_2 U_2$$

The reduction in momentum of the fluid results in thrust on the turbine. i.e.

$$F = \dot{m} (U_0 - U_2)$$

The power extracted by the turbine

$$P_T = F U_1 = \dot{m} (U_0 - U_2) U_1 \rightarrow \textcircled{1}$$

The power extracted is also equal to loss in KE

$$P_w = \frac{1}{2} \dot{m} (U_0^2 - U_2^2) \rightarrow \textcircled{2}$$

Equation $\textcircled{1}$ and $\textcircled{2}$

$$\dot{m} (U_0 - U_2) U_1 = \frac{1}{2} \dot{m} (U_0^2 - U_2^2)$$

$$\Rightarrow U_1 = \left(\frac{U_0 + U_2}{2} \right) \rightarrow \textcircled{3}$$

The ~~star~~ interference factor 'a' is defined as

$$a = \frac{\Delta U}{U_0} = \left(\frac{U_0 - U_1}{U_0} \right)$$

This can be rewritten as

$$a U_0 = U_0 - U_1 \Rightarrow U_1 = U_0 (1 - a) \rightarrow \textcircled{4}$$

Equation (3) and (4)

$$u_1 = \left(\frac{u_0 + u_2}{2} \right) = u_0(1-a)$$

$$u_0 + u_2 = 2u_0(1-a)$$

$$u_0 + u_2 = 2u_0 - 2au_0$$

$$\Rightarrow 2au_0 = (u_0 - u_2)$$

$$\Rightarrow a = \left(\frac{u_0 - u_2}{2u_0} \right) \rightarrow (5)$$

The expression for power can be recast as

$$P_T = \frac{1}{2} \rho A_1 (u_0^2 - u_2^2)$$

$$= \frac{1}{2} \rho A_1 u_1 (u_0 - u_2)(u_0 + u_2)$$

$$= \frac{1}{2} \rho A_1 u_0 (1-a)(2au_0)(2u_1)$$

$$= \frac{1}{2} \rho A_1 u_0 (1-a) 2au_0 (2u_0)(1-a)$$

$$P_T = 4a(1-a)^2 \left(\frac{1}{2} \rho A_1 u_0^3 \right)$$

The power in wind is $P_0 = \frac{1}{2} \rho A_1 u_0^3$

$$P_0 = \frac{1}{2} (\rho A_1 u_0) u_0^2$$

$$P_0 = \frac{1}{2} \rho A_1 u_0^3$$

power coefficient $C_p = \left(\frac{P_T}{P_0} \right) = \frac{4a(1-a)^2 \left(\frac{1}{2} \rho A_1 u_0^3 \right)}{\left(\frac{1}{2} \rho A_1 u_0^3 \right)}$

$$\boxed{C_p = 4a(1-a)^2}$$

2) Given for case 1: $P = 1200 \text{ W}$ $U_0 = 6 \text{ m/s}$
 $P = 1 \text{ atm} = 1 \times 101.325 \times 10^3 \text{ N/m}^2$
 $T = 27^\circ\text{C} = 300 \text{ K}$

$$P = C_p P_0 \quad \rho = \frac{P}{RT}$$

$$1200 = C_p \left(\frac{1}{2} \rho A U_0^3 \right)$$

$$= (C_p A) \left(\frac{1}{2} \times \frac{101.325 \times 10^3}{R \times 300} \times 6^3 \right) \rightarrow \textcircled{1}$$

For case 2: $P = ?$ $U_0 = 8 \text{ m/s}$
 $P = 0.85 \text{ atm} = 0.85 \times 101.325 \times 10^3 \text{ N/m}^2$
 $T = 15^\circ\text{C} = 288 \text{ K}$

$$P = C_p P_0$$

$$? = C_p \left(\frac{1}{2} \times \frac{0.85 \times 101.325 \times 10^3 \times A \times 8^3}{R \times 288} \right) \rightarrow \textcircled{2}$$

$\textcircled{2}/\textcircled{1}$ gives

$$\frac{?}{1200} = \frac{1 \times 6^3}{300} \times \frac{288}{0.85 \times 8^3} = \underline{\underline{2518 \text{ W}}}$$

It is ~~assumed~~ assumed that the turbine operates at the same power coefficient under both conditions.

3) Given. $U_0 = 8 \text{ m/s}$

$$D = 120 \text{ m}$$

$$N = 40 \text{ rpm}$$

$$\text{Density } \rho = \frac{P}{RT}$$

$$= \frac{101.325 \times 10^3}{287 \times 288}$$

$$= \underline{\underline{1.204 \text{ kg/m}^3}}$$

$$P = 101.325 \times 10^3 \text{ N/m}^2$$

$$R = 287 \text{ J/kg K}$$

$$T = 20^\circ\text{C} = 293 \text{ K}$$

max Power produced by turbine

$$P_T = C_p P_0$$

at max power output $C_p = \left(\frac{16}{27}\right)$

$$P_T = \left(\frac{16}{27}\right) P_0$$

$$= \left(\frac{16}{27}\right) \left(\frac{1}{2} \rho A U_0^3\right)$$

$$= \left(\frac{16}{27}\right) \left(\frac{1}{2} \times 1.204 \times \pi \times \frac{120^2}{4} \times 8^3\right)$$

$$= 2.065 \times 10^6 \text{ W}$$

$$= \underline{\underline{2.065 \text{ MW}}}$$

$$P = \frac{2\pi N T}{60}, \text{ where } T \text{ is the torque}$$

$$T = \frac{60 P}{2\pi N}$$

$$= \frac{60 \times 2.065 \times 10^6}{2\pi \times 40}$$

$$= 0.492 \times 10^6 \text{ Nm}$$

$$T = \underline{\underline{492 \text{ kNm}}}$$

4) Given $u_0 = 15 \text{ m/s}$

$$T = 15^\circ\text{C} = 288 \text{ K}$$

$$P = 1 \text{ atm} = 101.325 \times 10^3 \text{ N/m}^2$$

$$D = 120 \text{ m}$$

$$N = 40 \text{ rpm}$$

$$\rho = \frac{P}{RT}$$

Power density in wind stream

$$\frac{P_0}{A} = \left(\frac{1}{2} \rho u_0^3 \right)$$

$$\rho = \frac{P}{RT} = \frac{101.325 \times 10^3}{287 \times 288} = \underline{\underline{1.226 \text{ kg/m}^3}}$$

(i) Power density in wind stream

$$\left(\frac{P_0}{A} \right) = \frac{1}{2} \rho u_0^3$$

$$= \frac{1}{2} \times 1.226 \times 15^3$$

$$= \underline{\underline{2068 \text{ W/m}^2}}$$

(ii) Max obtainable power density

$$= \left(\frac{P_{\text{max}}}{A} \right) = (C_p)_{\text{max}} \left(\frac{P_0}{A} \right)$$

$$= \left(\frac{16}{27} \right) \left(\frac{P_0}{A} \right)$$

$$= \frac{16}{27} \times 2068 = \underline{\underline{1225 \text{ W/m}^2}}$$

(iii) Reasonably obtainable power density

$$= (C_p)_{\text{reasonable}} \times \left(\frac{P_0}{A} \right)$$

$$= 0.35 \times 2068$$

$$= \underline{\underline{723.8 \text{ W/m}^2}}$$

(iv) Total power

Assume $\epsilon_p = 0.35$

$$\begin{aligned} P_{\text{total}} &= A \times \left(\frac{P_{\text{max}}}{A} \right) \\ &= \pi \times \frac{120^2}{4} \times 1225 \times 0.35 \\ &= 8.185 \times 10^6 \text{ W} \\ &= \underline{\underline{8.185 \text{ MW}}} \end{aligned}$$

(v) Max axial thrust

$$\begin{aligned} F &= \frac{1}{2} \rho A U^2 \\ &= \frac{1}{2} \times 1.226 \times \pi \times \frac{120^2}{4} \times 15^2 \\ &= \underline{\underline{1.559 \times 10^6 \text{ N}}} \end{aligned}$$

(vi) Power at max efficiency

$$P_{\text{max}} = A \left(\frac{P_{\text{max}}}{A} \right) = 1225 \times \pi \times \frac{120^2}{4}$$

$$P = \frac{2\pi NT}{60}$$

$$T = \frac{60 P}{2\pi N}$$

$$= \frac{60 \times 1225 \times \pi \times \frac{120^2}{4}}{2\pi \times 60}$$

$$= \underline{\underline{3.307 \text{ kNm}}}$$