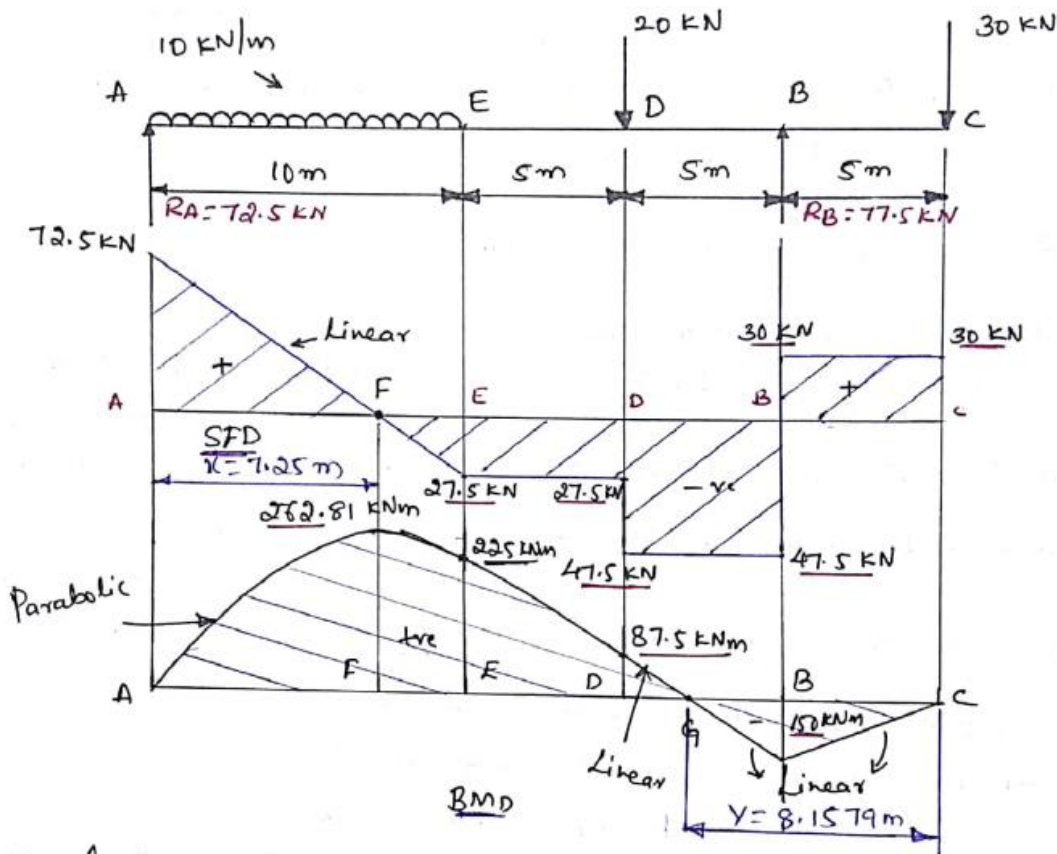


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Internal Assessment Test – 2

Sub: Mechanics of Materials				Code: 18ME52		
Date: 25/01/2022	Duration: 90 mins	Max Marks: 50	Sem: 3	Branch (sections): ME		
Answer one question from Part A and 3 questions from Part B						
PART A				Marks	OBE	
					CO	RBT
1	<p>A beam 25 m long is supported at A and B and is loaded as shown in fig. 1. Draw SFD and BMD for the beam. Find the position and magnitude of the maximum bending moment. Also determine the point of contraflexure.</p>			[20]	CO4	L3
<p style="text-align: center;">Fig. 1</p>						
2	<p>Draw SFD and BMD for the beam show in fig.2. Indicate all principal values.</p>			[20]	CO4	L3
<p style="text-align: center;">Fig. 2</p>						
PART B						
3	<p>With usual notations derive Lamé's equation.</p>			[10]	CO2	L2
4	<p>(a) Derive an expression to find longitudinal stress in thin cylinder. (b) Derive an expression to find circumferential stress in thin cylinder.</p>			[10]	CO2	L2
5	<p>A thin cylindrical shell is 3m long and 1m in internal diameter. It is subjected to internal pressure of 1.2 MPa. If the thickness of the sheet is 12mm, find the circumferential stress, longitudinal stress, changes in diameter, length and volume. Take <math>E=200</math> GPa and <math>\mu=0.3</math>.</p>			[10]	CO2	L2
6	<p>A thick cylindrical pipe of internal diameter 500 mm and 75 mm thick is subjected to an internal fluid pressure of <math>6 \text{ N/mm}^2</math> and external pressure of <math>5 \text{ N/mm}^2</math>. Determine the maximum hoop and minimum stress developed and draw the variation of hoop stress and radial stress across the thickness of the cylinder.</p>			[10]	CO2	L2

1



To find reactions at supports ( $R_A$  &  $R_B$ ):

Take moments about A

$$R_B \times 20 = (10 \times 10) \left(\frac{10}{2}\right) + 20 \times 15 + 30 \times 25$$

$$\therefore R_B = 77.5 \text{ kN (Ans)}$$

$$\text{Total load} = (10 \times 10) + 20 + 30 = R_A + R_B$$

$$\therefore R_A + 77.5 = 150; \therefore R_A = 72.5 \text{ kN (Ans)}$$

Shear force diagram (SFD)

$$\text{Shear force at A } F_A = R_A = +72.5 \text{ kN}$$

$$\text{Shear force at E, } F_E = +72.5 - (10 \times 10) = -27.5 \text{ kN}$$

Between E and D, shear force is constant at  $-27.5 \text{ kN}$

$$F_D = -27.5 \text{ kN (without point load)}$$

$$\text{Shear force at D, } F_D = -27.5 - 20 = -47.5 \text{ kN}$$

(Sudden variation due to point load)

Between D and B, Shear force is constant at  $-47.5 \text{ kN}$

$$F_B = -47.5 \text{ kN (without point load)}$$

$$\text{S.F at B, } F_B = -47.5 + 77.5 = +30 \text{ kN}$$

[Sudden variation due to point load]

Between 'B' and 'C' Shear force is constant at  $+30 \text{ kN}$

$$\text{S.F at C, } F_C = +30 \text{ kN}$$

Bending moment diagram (BMD)

Bending moment at A,  $M_A = 0$

$$\text{B.M at E, } M_E = 72.5 \times 10 - (10 \times 10) \left(\frac{10}{2}\right) = +225 \text{ kNm}$$

(Parabolic)

$$\text{B.M at D, } M_D = +72.5 \times 15 - (10 \times 10) \left(\frac{10}{2} + 5\right) = +87.5 \text{ kNm}$$

$$\text{B.M at B, } M_B = +72.5 \times 20 - (10 \times 10) \left(\frac{10}{2} + 10\right) - 20 \times 5$$
$$= -150 \text{ kNm}$$

$$\text{B.M at C, } M_C = 72.5 \times 25 - (10 \times 10) \left(\frac{10}{2} + 15\right) - 20 \times 10 + 77.5 \times 5 = 0$$

To find maximum bending moment.

At 'F', Shear force is changing its sign.

$$\text{Shear force at F, } F_F = 72.5 - 10x$$

$$\therefore x = 7.25 \text{ m (Ans)}$$

$$\text{Bending moment at F, } M_F = 72.5 \times 7.25 - (10 \times 7.25) \left(\frac{7.25}{2}\right)$$
$$= +262.8125 \text{ kNm (Ans)}$$

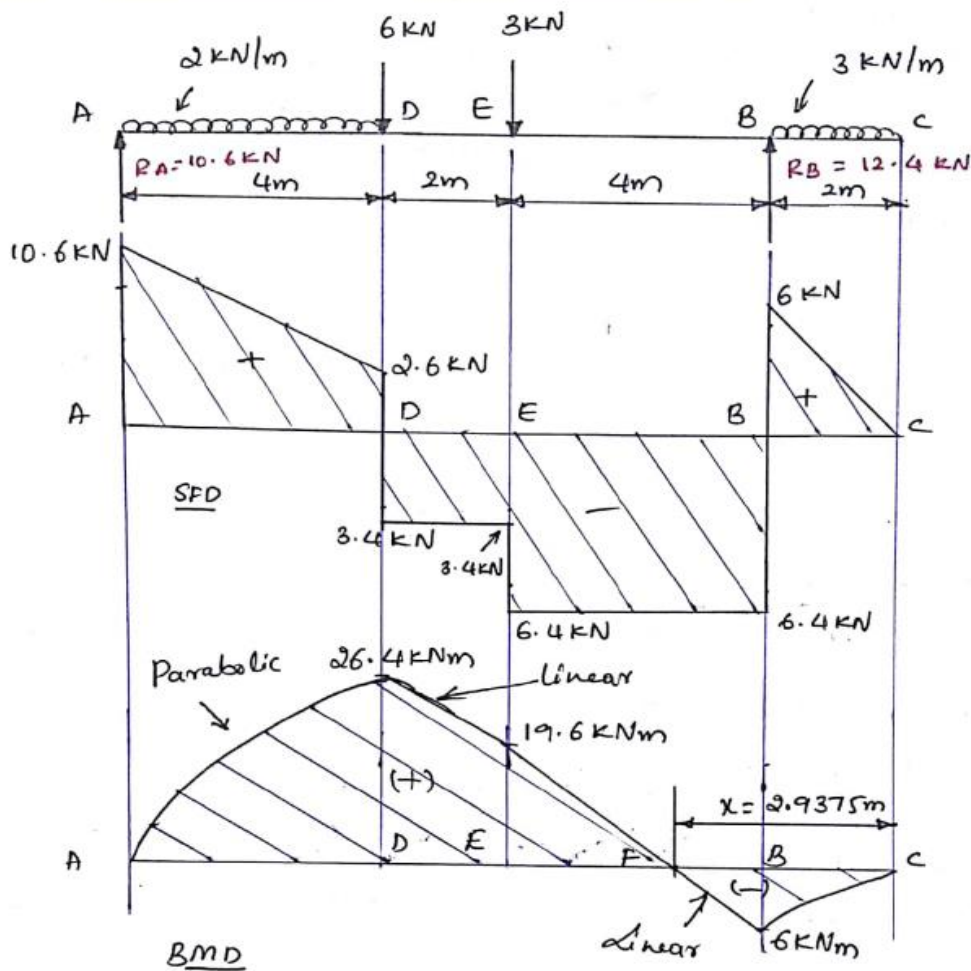
Point of contraflexure (It is the point where the BM changes its sign)

$$\text{B.M at G, } M_G = -304 + 77.5 (y - 5)$$

$$-304 + 77.5y - 387.5 = 0$$

$$\therefore y = 8.1579 \text{ m (Ans)}$$

$\therefore$  Point of contraflexure at 'G' is 8.1579 m from C.



To find Reactions at supports ( $R_A$  &  $R_B$ ).

Take moments about A

$$R_B \times 10 = (2 \times 4) \left( \frac{4}{2} \right) + 6 \times 4 + 3 \times 6 + (3 \times 2) \left( \frac{2}{2} + 10 \right) \Rightarrow$$

$$\therefore R_B = 12.4 \text{ kN (Ans)}$$

$$\text{Total load} = R_A + R_B$$

$$R_A + 12.4 = (2 \times 4) + 6 + 3 + (3 \times 2)$$

$$\therefore R_A = 10.6 \text{ kN (Ans)}$$

Shear force diagram (SFD)

$$\text{S.F at A, } F_A = +R_A = +10.6 \text{ kN}$$

$$\text{S.F at D, } F_D = R_A - 2 \times 4 = 2.6 \text{ kN}$$

$$\text{S.F at D, } F_D = 2.6 - 6 = -3.4 \text{ kN}$$

[Sudden variation due to Point load at D]

Shear force remains constant between D and E

$$S.F \text{ at } E, F_E = -3.4 \text{ kN}$$

$$S.F \text{ at } E, F_E = -3.4 - 3 = -6.4 \text{ kN}$$

[Sudden variation due to point load  
at E]

Shear force remains constant between E and B

$$S.F \text{ at } B, F_B = -6.4 \text{ kN}$$

$$S.F \text{ at } B, F_B = -6.4 + 12.4 = 6 \text{ kN}$$

[Sudden variation due to  $R_B$ ]

$$S.F \text{ at } C, F_C = 6 - 3 \times 2 = 0$$

Bending moment diagram (BMD)

$$B.M \text{ at } A, M_A = 0$$

$$B.M \text{ at } B, M_B = R_A \times 4 - (2 \times 4) \left(\frac{4}{2}\right) = +26.4 \text{ kNm}$$

$$B.M \text{ at } E, M_E = R_A \times 6 - (2 \times 4) \left(\frac{4}{2} + 2\right) - 6 \times 2 = +19.6 \text{ kNm}$$

$$B.M \text{ at } B, M_B = R_A \times 10 - (2 \times 4) \left(\frac{4}{2} + 6\right) - 6 \times 6 - 3 \times 4 = -6 \text{ kNm}$$

$$B.M \text{ at } C, M_C = 0$$

$$R_A \times 12 - (2 \times 4) \left(\frac{4}{2} + 8\right) - 6 \times 8 - 3 \times 6 + R_B \times 2 - (3 \times 2) \left(\frac{2}{2}\right) = 0$$

Point of contraflexure:

$$\text{Bending moment at } F, M_F = -(3 \times 2) \left(\frac{x-2}{2}\right) + R_B (x-2)$$

$$0 = -6x + 6 + 12.4x - 24.8$$

$$0 = +6.4x - 18.8$$

$$6.4x = 18.8$$

$$\therefore x = \frac{18.8}{6.4}$$

$$\therefore x = \underline{\underline{2.9375 \text{ m}}}$$

## 18.2. STRESSES IN A THICK CYLINDRICAL SHELL

Fig. 18.1 (a) shows a thick cylinder subjected to a internal fluid pressure.

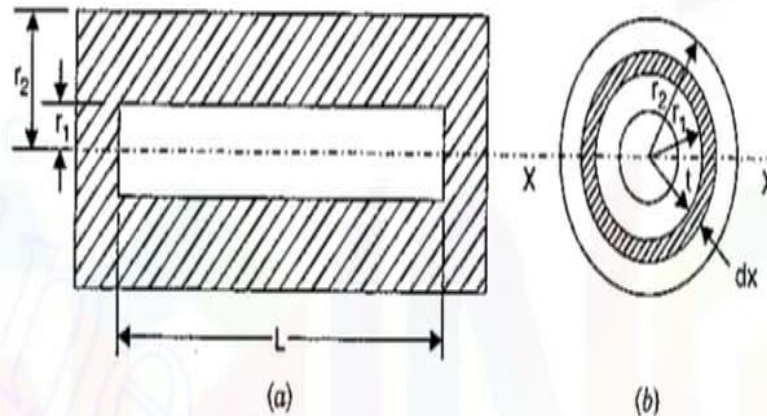


Fig. 18.1

Let  $r_2$  = External radius of the cylinder,  
 $r_1$  = Internal radius of the cylinder, and  
 $L$  = Length of cylinder.

Consider an elementary ring of the cylinder of radius  $x$  and thickness  $dx$  as shown in Fig. 18.1 (b) and 18.2.

Let  $p_x$  = Radial pressure on the inner surface of the ring  
 $p_x + dp_x$  = Radial pressure on the outer surface of the ring  
 $\sigma_x$  = Hoop stress induced in the ring.

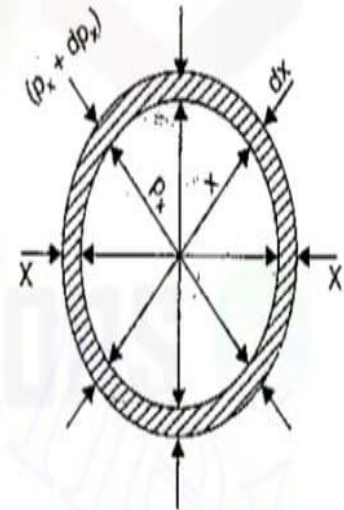


Fig. 18.2

Take a longitudinal section  $x-x$  and consider the equilibrium of half of the ring of Fig. 18.2 as shown in Fig. 18.2 (a) or in Fig. 18.2 (b).

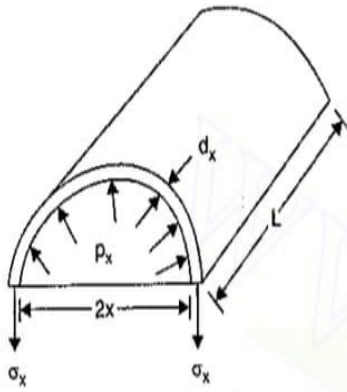


Fig. 18.2 (a)

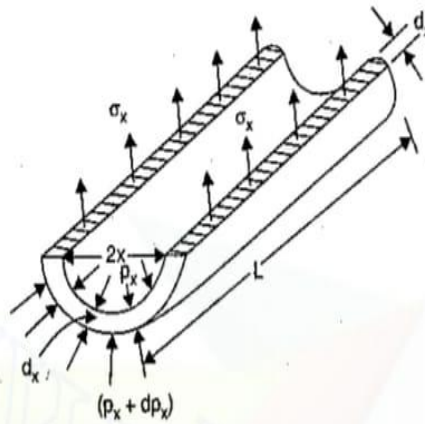


Fig. 18.2 (b)

Bursting force

$$\begin{aligned}
 &= p_x (2xL) - (p_x + dp_x) \times 2(x + dx) \cdot L \\
 &= 2L [p_x \cdot x - (p_x \cdot x + p_x \cdot dx + x dp_x + dp_x \cdot dx)] \\
 &= 2L [-p_x \cdot dx - x \cdot dp_x] \quad \text{(Neglecting } dp_x \cdot dx \text{ which is a small quantity)} \\
 &= -2L (p_x dx + x \cdot dp_x) \quad \dots(i)
 \end{aligned}$$

$$\text{Resisting force} = \text{Hoop stress} \times \text{Area on which it acts} = \sigma_x \times 2dx \cdot L \quad \dots(ii)$$

Equating the resisting force to the bursting force, we get

$$\sigma_x \times 2dx \cdot L = -2L (p_x \cdot dx + x \cdot dp_x)$$

$$\text{or} \quad \sigma_x = -p_x - x \frac{dp_x}{dx} \quad \dots(iii)$$

The longitudinal strain at any point in the section is constant and is independent of the radius. This means that cross-sections remain plane after straining and this is true for sections, remote from any end fixing. As longitudinal strain is constant, hence longitudinal stress will also be constant.

Let  $\sigma_2$  = Longitudinal stress.

Hence at any point at a distance  $x$  from the centre, three principal stresses are acting :  
They are :

- (i) the radial compressive stress,  $p_x$
- (ii) the hoop (or circumferential) tensile stress,  $\sigma_x$
- (iii) the longitudinal tensile stress  $\sigma_2$ .

The longitudinal strain ( $e_2$ ) at this point is given by,

$$e_2 = \frac{\sigma_2}{E} - \frac{\mu\sigma_x}{E} + \frac{\mu p_x}{E}$$

But longitudinal strain is constant.

$$\therefore \frac{\sigma_2}{E} - \frac{\mu\sigma_x}{E} + \frac{\mu p_x}{E} = \text{constant}$$

But  $\sigma_2$  is also constant, and for the material of the cylinder  $E$  and  $\mu$  are constant.

$$\begin{aligned}
 \therefore \sigma_x - p_x &= \text{constant} \\
 &= 2a \text{ where } a \text{ is constant}
 \end{aligned}$$

$$\therefore \sigma_x = p_x + 2a \quad \dots(iv)$$

Equating the two values of  $\sigma_x$  given by equations (iii) and (iv), we get

$$p_x + 2a = -p_x - x \frac{dp_x}{dx}$$

or  $x \cdot \frac{dp_x}{dx} = -p_x - p_x - 2a = -2p_x - 2a$

or  $\frac{dp_x}{dx} = -\frac{2p_x}{x} - \frac{2a}{x} = -\frac{2(p_x + a)}{x}$

or  $\frac{dp_x}{(p_x + a)} = -\frac{2dx}{x}$

Integrating the above equation, we get

$$\log_e (p_x + a) = -2 \log_e x + \log_e b$$

where  $\log_e b$  is a constant of integration.

The above equation can also be written as

$$\log_e (p_x + a) = -\log_e x^2 + \log_e b$$

$$= \log_e \frac{b}{x^2}$$

$$\therefore p_x + a = \frac{b}{x^2}$$

or  $p_x = \frac{b}{x^2} - a \quad \dots(18.1)$

Substituting the values of  $p_x$  in equation (iv), we get

$$\sigma_x = \frac{b}{x^2} - a + 2a = \frac{b}{x^2} + a \quad \dots(18.2)$$

The equation (18.1) gives the radial pressure  $p_x$  and equation (18.2) gives the hoop stress at any radius  $x$ . These two equations are called *Lame's equations*. The constants 'a' and 'b' are obtained from boundary conditions, which are :

(i) at  $x = r_1$ ,  $p_x = p_0$  or the pressure of fluid inside the cylinder, and

(ii) at  $x = r_2$ ,  $p_x = 0$  or atmosphere pressure.

After knowing the values of 'a' and 'b', the hoop stress can be calculated at any radius.

4(a)

### EXPRESSION FOR LONGITUDINAL STRESS

Consider a thin cylindrical vessel subjected to internal fluid pressure. The longitudinal stress will be set up in the material of the cylinder, if the bursting of the cylinder takes place along the section AB of Fig. (a).

The longitudinal stress ( $\sigma_2$ ) developed in the material is obtained as :

Let  $p$  = Internal pressure of fluid stored in thin cylinder

$d$  = Internal diameter of cylinder

$t$  = Thickness of the cylinder

$\sigma_2$  = Longitudinal stress in the material.

The bursting will take place if the force due to fluid pressure acting on the ends of the cylinder is more than the resisting force due to longitudinal stress ( $\sigma_2$ ) developed in the material as shown in Fig. 17.4 (b). In the limiting case, both the forces should be equal.

Force due to fluid pressure =  $p \times$  Area on which  $p$  is acting

$$= p \times \frac{\pi}{4} d^2$$

Resisting force =  $\sigma_2 \times$  Area on which  $\sigma_2$  is acting

$$= \sigma_2 \times \pi d \times t$$

$\therefore$  Hence in the limiting case

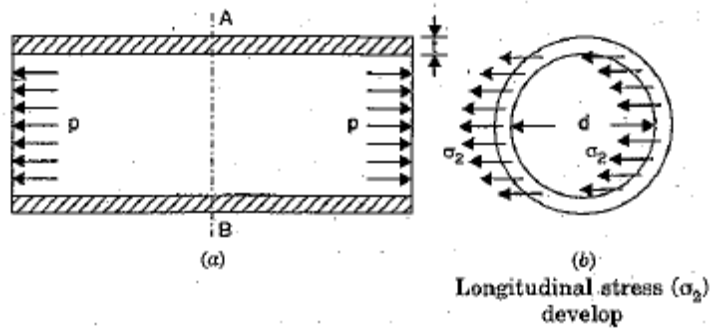
Force due to fluid pressure = Resisting force

$$p \times \frac{\pi}{4} d^2 = \sigma_2 \times \pi d \times t$$

$$\therefore \sigma_2 = \frac{p \times \frac{\pi}{4} d^2}{\pi d \times t} = \frac{pd}{4t}$$

The stress  $\sigma_2$  is also tensile.





4(b)

### EXPRESSION FOR CIRCUMFERENTIAL STRESS (OR HOOP STRESS)

Consider a thin cylindrical vessel subjected to an internal fluid pressure. The circumferential stress will be set up in the material of the cylinder, if the bursting of the cylinder takes place as shown in Fig. (a).

The expression for hoop stress or circumferential stress ( $\sigma_1$ ) is obtained as given below.

Let  $p$  = Internal pressure of fluid

$d$  = Internal diameter of the cylinder

$t$  = Thickness of the wall of the cylinder

$\sigma_1$  = Circumferential or hoop stress in the material.

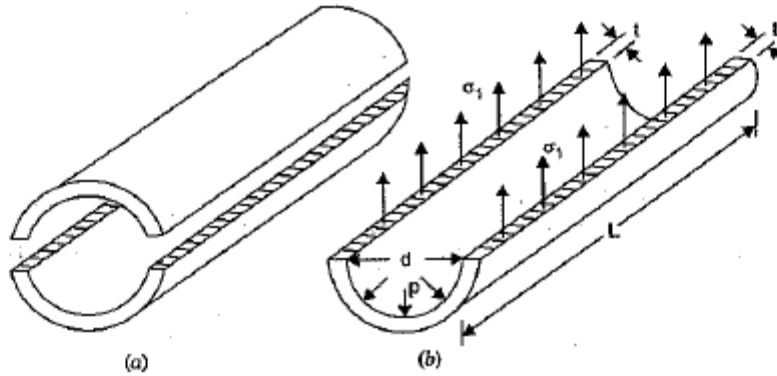


Fig. 17.3

The bursting will take place if the force due to fluid pressure is more than the resisting force due to circumferential stress set up in the material. In the limiting case, the two forces should be equal.

$$\begin{aligned} \text{Force due to fluid pressure} &= p \times \text{Area on which } p \text{ is acting} \\ &= p \times (d \times L) \end{aligned} \quad \dots(i)$$

( $\because$   $p$  is acting on projected area  $d \times L$ )

$$\begin{aligned} \text{Force due to circumferential stress} &= \sigma_1 \times \text{Area on which } \sigma_1 \text{ is acting} \\ &= \sigma_1 \times (L \times t + L \times t) \\ &= \sigma_1 \times 2Lt = 2\sigma_1 \times L \times t \end{aligned} \quad \dots(ii)$$

Equating (i) and (ii), we get

$$p \times d \times L = 2\sigma_1 \times L \times t$$

$$\sigma_1 = \frac{pd}{2t} \quad (\text{cancelling } L)$$

1. Circumferential stress,  $\sigma_c$ :

$$\begin{aligned}\sigma_c &= (p \times d) / (2 \times t) \\ &= (1.2 \times 1000) / (2 \times 12) \\ &= \underline{50 \text{ N/mm} = 50 \text{ MPa}} \text{ (Tensile)}.\end{aligned}$$

2. Longitudinal stress,  $\sigma_L$ :

$$\begin{aligned}\sigma_L &= (p \times d) / (4 \times t) \\ &= \sigma_c / 2 = 50 / 2 \\ &= \underline{25 \text{ N/mm} = 25 \text{ MPa}} \text{ (Tensile)}.\end{aligned}$$

3. Circumferential strain,  $\epsilon_c$ :

$$\begin{aligned}\epsilon_c &= \frac{(p \times d)}{(4 \times t)} \times \frac{(2 - \mu)}{E} \\ &= \frac{(1.2 \times 1000)}{(4 \times 12)} \times \frac{(2 - 0.3)}{200 \times 10^3} \\ &= \underline{2.125 \times 10^{-04}} \text{ (Increase)}\end{aligned}$$

Change in diameter,  $\delta d = \epsilon_c \times d$

$$= 2.125 \times 10^{-04} \times 1000 = \underline{0.2125 \text{ mm}} \text{ (Increase)}.$$

4. Longitudinal strain:  $\epsilon_L = \frac{(p \times d)}{(4 \times t)} \times \frac{(1 - 2 \times \mu)}{E}$

$$\begin{aligned}&= \frac{(1.2 \times 1000)}{(4 \times 12)} \times \frac{(1 - 2 \times 0.3)}{200 \times 10^3} \\ &= \underline{5 \times 10^{-05}} \text{ (Increase)}\end{aligned}$$

Change in length =  $\epsilon_L \times L = 5 \times 10^{-05} \times 3000 = \underline{0.15 \text{ mm}}$  (Increase).

Volumetric strain,  $\frac{dv}{V}$  :

$$\begin{aligned}\frac{dv}{V} &= \frac{(p \times d)}{(4 \times t) \times E} \times (5 - 4 \times \mu) \\ &= \frac{(1.2 \times 1000)}{(4 \times 12) \times 200 \times 10^3} \times (5 - 4 \times 0.3) \\ &= \underline{4.75 \times 10^{-4}} \text{ (Increase)}\end{aligned}$$

$$\begin{aligned}&= 4.75 \times 10^{-4} \times \frac{\pi}{4} \times 1000^2 \times 3000 \\ &= 1.11919 \times 10^6 \text{ mm}^3 = 1.11919 \times 10^{-3} \text{ m}^3 \\ &= \underline{1.11919 \text{ Litres}}.\end{aligned}$$

Data:

$$d_2 = 500 \text{ mm}; \quad \therefore r_2 = 250 \text{ mm}; \quad t = 75 \text{ mm} \quad \therefore r_1 = 250 + 75 \\ = \underline{325 \text{ mm}}$$

$$P_x = 6 \text{ N/mm}^2$$

Soln: Internal pressure at any radius,  $x$   $P_x = \frac{b}{x^2} - a$

$$\text{when } x = r_2 = 250 \text{ mm}, \quad P_x = 6 \text{ N/mm}^2$$

$$\therefore 6 = \frac{b}{250^2} - a \quad \text{--- (i)}$$

$$\text{when } x = r_1 = 325 \text{ mm}, \quad P_x = 0.$$

$$0 = \frac{b}{325^2} - a \quad \text{--- (ii)}$$

$$\text{Eqn (i) - (ii) gives, } 6 = \frac{b}{250^2} - \frac{b}{325^2}$$

$$\therefore b = \underline{918478.261}$$

$$\text{Substituting 'b' in eqn (i), } a = \underline{8.696}$$

$$\text{Hoop stress at any radius, } x, \quad f_x = \frac{b}{x^2} + a$$

$$f_x = \frac{918478.261}{x^2} + 8.696$$

$$\text{Mean radius, } r_m = \frac{r_2 + r_1}{2} = \frac{250 + 325}{2} = \underline{287.5 \text{ mm}}$$

when  $x = r_2 = 250 \text{ mm}$ ;  $f_x = f_{250}$

i.e., hoop stress at the inner circumference

$$f_{250} = \frac{918478.261}{250^2} + 8.696 = 23.4 \text{ N/mm}^2$$

when  $x = r_m = 287.5 \text{ mm}$ ,  $f_x = f_{287.5}$

i.e., hoop stress at the mean radius

$$f_{287.5} = \frac{918478.261}{287.5^2} + 8.696 = 19.81 \text{ N/mm}^2$$

when  $x = r_1 = 325 \text{ mm}$ ,  $f_x = f_{325}$

i.e., hoop stress at the outer circumference

$$f_{325} = \frac{918478.261}{325^2} + 8.696 = 17.4 \text{ N/mm}^2$$

Radial pressure at the mean circumference

$$P_{287.5} = \frac{918478.261}{287.5^2} - 8.696 = 2.416 \text{ N/mm}^2$$

