

Internal Assessment Test 1 – May 2022

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Sub:	Finite Element Methods					Code:	18ME61		
Date:	06/ 05 / 2022	Duration:	90 mins	Max Marks:	50	Sem:	VI	Branch:	MECH

Note: Answer all questions.

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			OBE	
			CO	RBT
1	Explain the basic steps involved in FEM.	8	CO1	L2
2	Write the equilibrium equation for 2D state of stress.	2	CO1	L1
3	Derive stiffness matrix for one dimensional bar element.	10	CO3	L2
4	A cantilever beam of span 'L' is subjected to a point load at free end. Derive an equation for the deflection at free end by using RR method. Assume polynomial displacement	10	CO2	L3



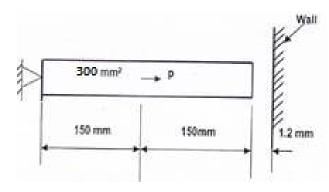
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- A compound bar 800 mm long is made of steel of 500 mm length with an area of 400 mm² with the remaining length made of brass having an area of 300 mm². At the junction it is subjected to an axial load of 200 KN which is in compression to steel. Both the ends are fixed. *E*_{steel}=200 *GPa*, *E*_{Brass}=70 *GPa*. Find nodal displacements, stress in each material.
- 10 CO4 L3
- 6 A load of P = 60 KN is applied as shown. Determine the following, a) Nodal Displacement, b) Stress in each member. Given Data: E = 200 GPa

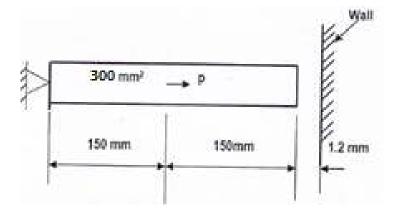




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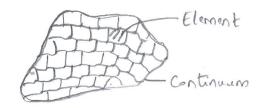


1st IAT Solution

1 STEPS IN FEA

1. Discritization of given Continuum

In this step, the Continuum is Subdivided into number of pouts Called finite elements. The type of element Selected depends on the Kind of analysis namely 1D, 2D or 3D.



2. Sclection of displacement model for each eliment is

The displacement variation for each eliment is

unknown. Hence a mathematical model is represented for

each finite element. They can be either polynomial or

trignometric function.

3. Generation of stiffness matrix for each finite element

4. To derive global Stiffness matrix $K = K_1 + K_2 - \cdots$

5) Imposing the equilibrium equation [K][2] = [F]

Enforcing the boundary Conditions of the given problem

There are two methods to enforce the boundary

Condition

Description

Llimination method - He eliminate the Componding hows of

Columns were we have Constraints

Columns were we have Constraints

The penalty Method - Add a Constant C= | max kij | X 104

The first of last element of diagonal me

Determination of unknowns.

In this step, the unknowns such as stesses,

displacements, strain etc are found.

2. Equilibrium equation for 2D state of stress $\frac{\partial G_{N}}{\partial n} + \frac{\partial C_{N}y}{\partial y} + X = 0$ $\frac{\partial C_{yx}}{\partial x} + \frac{\partial G_{y}}{\partial y} + Y = 0$

3. Stiffness matrix for one dimensional bar element

Strain energy for 3 D element $U_e = \frac{1}{2} \int_V \sigma^T \epsilon \, dv$ Strain energy for 1D element $U_e = \frac{1}{2} \int_{l_e} \sigma^T \epsilon \, A \, dx$ Strain $\epsilon = Bq$

where
$$B = \frac{1}{l_e} \begin{bmatrix} -1 & 1 \end{bmatrix}$$

$$q = \begin{bmatrix} q_1 \\ q_2 \end{bmatrix}$$

Strain energy for 1D element $U_e = \frac{1}{2} \int_{l_e} [EBq]^T Bq A dx$

Relation between natural and Cartesian coordinate is

$$\xi = \frac{x - x_1}{x_2 - x_1} - 1$$

$$\frac{d\xi}{dx} = \frac{2}{l_e}$$

$$U_e = \frac{1}{2} \int_{l_e} q^T B^T E B q A \frac{l_e}{2} d\xi$$

$$U_e = \frac{1}{2} q^T \left[EA \, \frac{l_e}{2} \int_{l_e} \, B^T \, Bd\xi \, \right] q$$

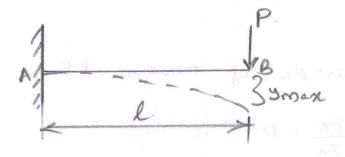
$$U_e = \frac{1}{2} q^T k_e \ q$$

Where $k_e = stiffness\ matrix$

$$k_e = EA \frac{l_e}{2} \int_{l_e} B^T B d\xi$$

$$K = \frac{EA}{l_e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

A Cantilever beam Subjected to point load at free end. Derive an ean for max' deflection using R.R method.



Sol 1) Potential Energy functional

T = S.E + W.P

(+ve) (-ve)

$$S.E = \frac{EI}{2} \int_{0}^{1} \left(\frac{d^2y}{dn^2}\right)^2 dn$$

W.P = P. Ymax

Assume displacement function $y = a_0 + a_1 x + a_2 x^2 + a_3 x^3$

B.C At A, x=0, y=0 AtA,x=0, dy 20

 $\frac{dy}{dx} = \frac{a_1 + 2a_2x + 3a_3x^2}{4x}$

a =0

 $y = a_2 x^2 + a_3 x^3$

$$\frac{dy}{dx} = 2a_{2}x + 3a_{3}x^{2}$$

$$\frac{d^{2}y}{dx^{2}} = 2a_{2} + 6a_{3}x$$

$$y = y_{max} \quad \text{at} \quad x = 1$$

$$y = x_{2}x^{2} + a_{3}x^{3}$$

$$y_{max} = a_{2}l^{2} + a_{3}t^{3}$$

$$P \in \text{ functional}$$

$$\pi = \frac{ET}{2} \int_{0}^{1} (2a_{2} + 6a_{3}x)^{2} dx - P(a_{2}l^{2} + a_{3}l^{3})$$

$$= \frac{ET}{2} \int_{0}^{1} [4a_{2}^{2} + 36a_{3}^{2}x^{2} + 24a_{2}a_{3}x] dx - P(a_{2}l^{2} + a_{3}l^{3})$$

$$= \frac{ET}{2} \left[4a_{2}^{2}x + 36a_{3}^{2}x^{3} + 24a_{2}a_{3}x^{2} \right]^{1} - P(a_{2}l^{2} + a_{3}l^{3})$$

$$\pi = \frac{ET}{2} \left[4a_{2}^{2}l + 36a_{3}^{2}x^{3} + 24a_{2}a_{3}x^{2} \right]^{1} - P(a_{2}l^{2} + a_{3}l^{3})$$

$$V = \frac{ET}{2} \left[4a_{2}^{2}l + 36a_{3}^{2}x^{3} + 24a_{2}a_{3}x^{2} \right]^{1} - P(a_{2}l^{2} + a_{3}l^{3})$$

$$\sqrt{2} = \frac{ET}{2} \left[4a_{2}^{2}l + 36a_{3}^{2}l + 24a_{2}a_{3}l^{2}l \right] - P(a_{2}l^{2} + a_{3}l^{3})$$

$$\sqrt{2} = \frac{ET}{2} \left[4a_{2}^{2}l + 36a_{3}^{2}l + 24a_{2}a_{3}l^{2}l \right] - P(a_{2}l^{2} + a_{3}l^{3})$$

$$\sqrt{2} = \frac{ET}{2} \left[4a_{2}^{2}l + 36a_{3}^{2}l + 24a_{2}a_{3}l^{2}l \right] - P(a_{2}l^{2} + a_{3}l^{3}l + 24a_{2}a_{3}l^{2}l \right] - P(a_{2}l^{2} + a_{3}l^{3}l + 24a_{2}a_{3}l^{2}l + 24a_{2}a_{3}l^{2}l - P(a_{2}l^{2} + a_{3}l^{3}l + 24a_{2}a_{3}l + 24a_{2}a_{3}l^{2}l - P(a_{2}l^{2} + a_{3}l^{3}l + 24a_{2}a_{3}l + 24a_{2}$$

 $\frac{\partial R}{\partial a_3} = 0 \quad ; \quad \frac{EI}{2} \left[24 a_3 l^3 + 12 a_2 l^2 \right] - Pl^3 = 0 \quad \Rightarrow 2$

$$\frac{EI}{2} \left[16a_2l^2 + 24a_3l^3 \right] - 2Pl^3 = 0$$

$$\frac{EI}{2} \left[12a_2 l^2 + 24a_3 l^3 \right] - Pl^3 = 0$$

$$a_2 = \frac{2Pl^3}{4l^2 \cdot EI}$$

$$a_2 = \frac{Pl}{2EI}$$

Sub. az in egn 2

$$\frac{EI}{2} \left[12 \cdot \frac{Pl}{2EI} \cdot l^2 + 2493 l^3 \right] - Pl^3 = 0$$

$$\frac{6Pl^3}{EI} + 24q_3l^3 = \frac{2Pl^3}{EI}$$

$$24a_3l^3 = \frac{2Pl^3}{EI} = \frac{6Pl^3}{EI} = -\frac{4Pl^3}{EI}$$

$$a_3 = -4Pl^3 = -Pl$$

$$6ET$$

$$24l^3.ET$$

Man' deflection of the thempalges small se

$$y_{\text{max}} = a_2 l^2 + a_3 l^3$$

$$= \frac{P \cdot l^3}{2EI} - \frac{P l^3}{6EI}$$

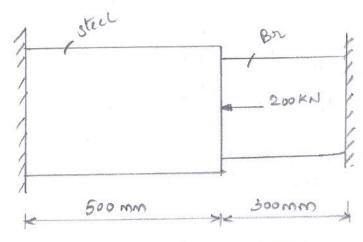
$$\frac{y_{\text{max}}}{3EI} = \frac{p_{2}^{3}}{3EI}$$

= 1 1 1 4 2 2 4 36 2 1 + 24 2 3 2

of 500mm length with an area of 400 mm² with the remaining length made of brass having an area of 300 mm². At the junction it is subjected to an axial load of 200 km which is in Compression to steel. Both the ends are fixed.

Est = 200 GPa; EBI = 70 GPa, Using Penalty method find the foll. nodal displacement, Shess in each element, reaction at Supports.

501



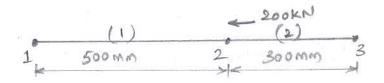
Est = 2004Pa

EBIR 709Pa

A1 = 400 mm2

A2 = 300 mm2

F. E. Model



Staffners matrix for each eliment

For element 1

$$K_{1} = \frac{E_{gt}A_{1}}{l_{1}} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$= 200 \times 10^{3} \times 400 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$= 10^{5} \begin{bmatrix} 1.6 & -1.6 \\ -1.6 & 1.6 \end{bmatrix}^{1}$$

$$= 10^{5} \begin{bmatrix} 1.6 & -1.6 \\ -1.6 & 1.6 \end{bmatrix}^{1}$$

$$= 70 \times 10^{3} \times 300 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$= 70 \times 10^{3} \times 300 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$= 10^{5} \begin{bmatrix} 0.7 & -0.7 \\ -0.7 & 0.7 \end{bmatrix}^{2}$$

$$= 10^{5} \begin{bmatrix} 0.7 & -0.7 \\ -0.7 & 0.7 \end{bmatrix}^{2}$$

$$= 10^{5} \begin{bmatrix} 1.6 & -1.6 & 0 \\ -1.6 & 2.3 & -0.7 \\ 0 & -0.7 & 0.7 \end{bmatrix}$$

Using Penalty method
$$C = |\max Kij| \times 10^4$$

$$= 2.3 \times 10^5 \times 10^4$$

$$C = 2.3 \times 10^9$$

'C' Should be added to first 8 last element of the diagonal.

$$K = 10^{5} \begin{cases} 1.6 & -1.6 & 0 \\ +c & \\ -1.6 & 2.3 & -0.7 \\ 0 & -0.7 & 0.7 \\ +c & \\ \end{cases}$$

$$K = \begin{bmatrix} 2.30016 \times 10^{9} & -1.6 \times 10^{5} & 0 \\ -1.6 \times 10^{5} & 2.3 \times 10^{5} & -0.7 \times 10^{5} \\ 0 & -0.7 \times 10^{5} & 2.30007 \times 10^{5} \end{bmatrix}$$

Equilibrium egn.

$$\begin{bmatrix} 2.30016 \times 10^{9} & -1.6 \times 10^{5} & 0 \\ -1.6 \times 10^{5} & 2.3 \times 10^{5} & -0.7 \times 10^{5} \end{bmatrix} \begin{bmatrix} q_{1} \\ q_{2} \\ q_{3} \end{bmatrix} = \begin{bmatrix} 0 \\ -200 \times 10^{3} \\ 0 \end{bmatrix}$$

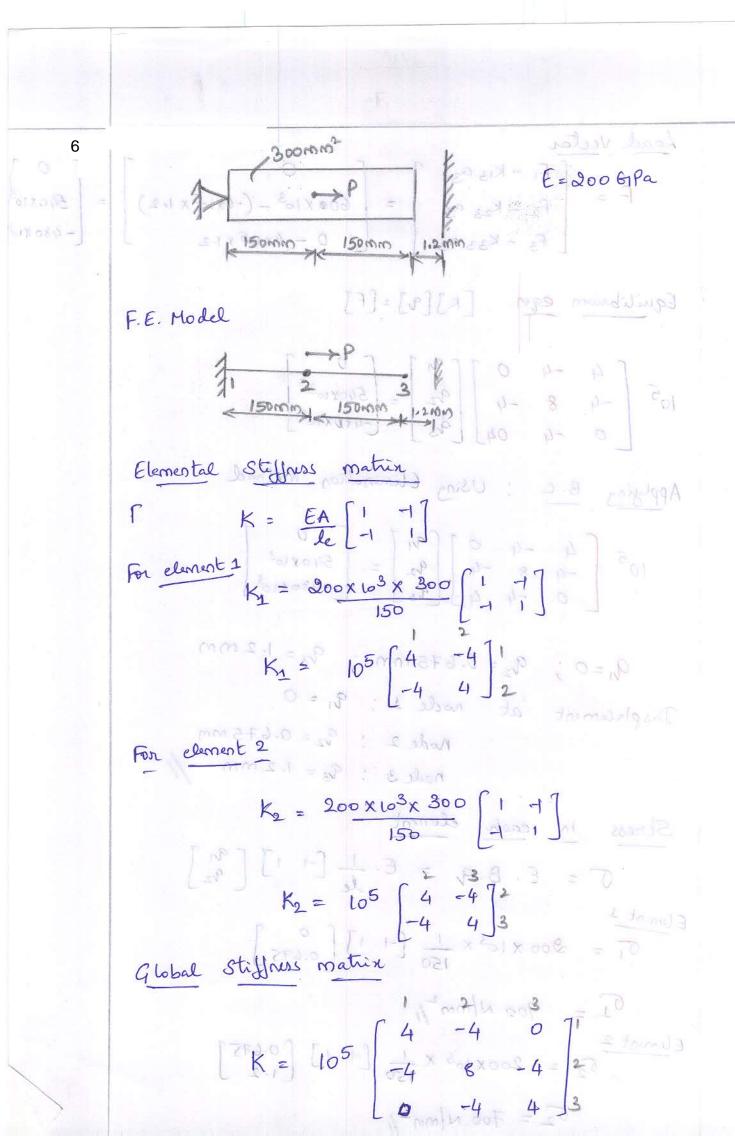
Solving Equilibrium eqn

$$q_1 = -6.0491 \times 10^{-5} \, \text{mm} \qquad q_2 = -0.8696 \, \text{mm}$$

$$\overline{O}_{2} = EB. Q$$

$$= EB. \frac{1}{l_{2}} \begin{bmatrix} -1 & 1 \end{bmatrix} \begin{bmatrix} q_{3} \\ q_{4} \end{bmatrix}$$

$$= 70 \times 10^{3} \times \frac{1}{300} \left[-1 \right] \left[-0.8696 -2.6466 \times 10^{5} \right]$$



Load Vector

$$F = \begin{bmatrix} F_1 - K_{15} \alpha_3 \\ F_2 - K_{23} \alpha_3 \\ F_3 - K_{23} \alpha_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 600 \times 10^3 - (-4x_{15} \times 1.2) \end{bmatrix} = \begin{bmatrix} 0 \\ 540 \times 10^3 \\ -480 \times 10^3 \end{bmatrix}$$

Equilibrium eqn $\begin{bmatrix} KJ[Y] = [F] \end{bmatrix}$

$$\begin{bmatrix} 4 & -4 & 0 \\ -4 & 8 & -4 \\ 0 & -4 & 04 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ q_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 540 \times 10^3 \\ -480 \times 10^3 \end{bmatrix}$$

Applying B.C: Using Eliminating Method

$$\begin{bmatrix} 10^5 & 4 & -4 & 0 \\ 4 & 8 & -4 \\ 0 & -4 & 4 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ -6 & 4 \end{bmatrix} = \begin{bmatrix} 0 \\ 540 \times 10^3 \\ -480 \times 10^3 \end{bmatrix}$$

$$Q_1 = 0; \quad Q_2 = 0.675 \text{ mm}; \quad Q_3 = 1.2 \text{ mm}$$

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Stress in each element

$$0 = 0; \quad Q_2 = 0.675 \text{ mm}; \quad Q_3 = 1.2 \text{ mm}$$

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$$0 = 0; \quad Q_3 = 0.675 \text{ mm}; \quad Q_3 = 0.67$$