



# CBCS SCHEME

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18CV72

## Seventh Semester B.E. Degree Examination, Jan./Feb. 2023 Design of RCC and Steel Structures

Time: 3 hrs.

Max. Marks: 100

- Note: 1. Answer any TWO full questions, choosing ONE full question from each module.  
2. Use of IS-456, IS-800, SP-16, SP(6) – steel tables are permitted.  
3. Assume missing data suitably.*

### Module-1

- 1 Design slab and beam type combined footing for two columns of size 300mm × 300mm and 400 × 400mm subjected to 500kN and 700kN respectively. The centre to centre spacing between columns is 3.50m. The width of the footing is restricted to 1.5m. Take SBC of soil = 150kN/m<sup>2</sup>. Use M<sub>25</sub> and Fe415 grades. Also show reinforcement in L/S and C/S.  
(50 Marks)

OR

- 2 Design a cantilever retaining wall to retain an earth embankment 4m high above ground level. The density of earth is 18kN/m<sup>3</sup> and its angle of repose is 30°. The embankment is horizontal at top. The S.B.C. of soil is 200kN/m<sup>2</sup>. The coefficient of friction between soil and concrete is 0.5. Adopt M-20 and Fe415 grades. Draw C/S elevation of retaining wall.  
(50 Marks)

### Module-2

- 3 A line diagram of a roof truss with internal loads and forces in each members are shown in Fig.Q.3. Design the various members of the roof truss along with their end connection with bolt using property class 5.6 black bolts. Also design the bearing plate at the support for the reaction and anchor bolts for an uplift force of 15kN. Draw elevation of truss greater than half span.  
(50 Marks)

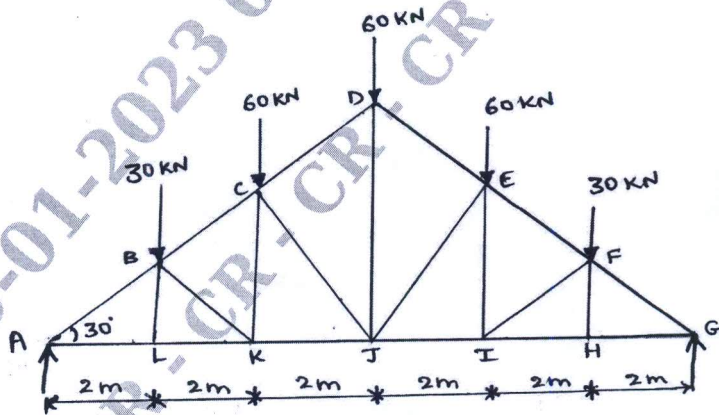


Fig.Q.3

Important Note : 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.  
2. Any revealing of identification, appeal to evaluator and /or equations written eg. 42+8 = 50, will be treated as malpractice.

Tabulation of member forces

Members	Length (m)	Force (kN)	Nature of Force
AB, GF	2.31	240.00	Compression
BC, FE	2.31	210.00	Compression
CD, ED	2.31	160.04	Compression
AL, GH	2.00	207.84	Tension
LK, HI	2.00	207.84	Tension
KJ, IJ	2.00	181.82	Tension
BL, FH	1.154	0.00	-
BK, FI	2.31	30.00	Compression
CK, EI	2.31	15	Tension
CJ, EJ	3.05	66.05	Compression
DJ	3.46	66.00	Compression

OR

- 4 Design a simply supported gantry girder to carry an electrically operated travelling crane with the following data:  
 Span of crane bridge = 25m  
 Column spacing = span of gantry girder = 8m  
 Wheel Base = 3.5m  
 Crane capacity = 200kN  
 Weight of crane bridge = 150kN  
 Weight of Trolley = 75kN  
 Min Hook Distance = 1.0m  
 Weight of Rail = 0.30kN/m  
 Height of Rail = 105mm  
 Also draw sectional elevation.

(50 Marks)

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## Seventh Semester B.E. Degree Examination, Jan./Feb. 2023 Design of RCC and Steel Structures

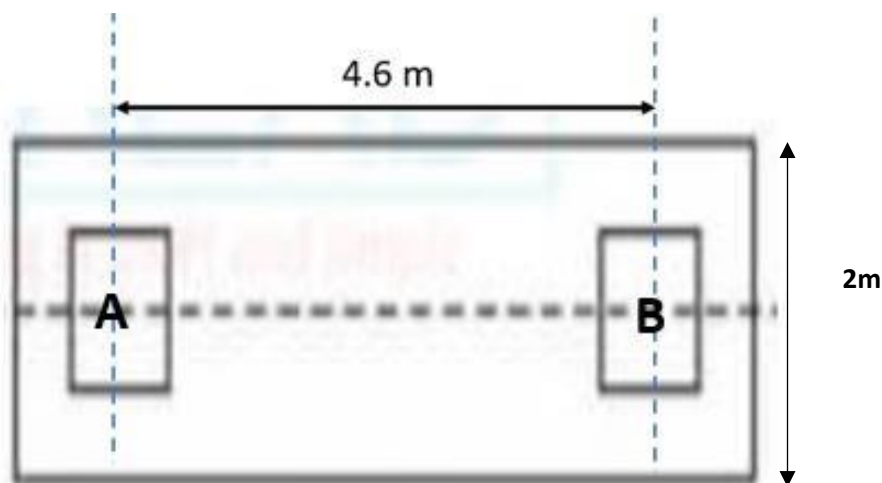
Time: 3 hrs.

Max. Marks: 100

- Note: 1. Answer any TWO full questions, choosing ONE full question from each module.  
2. Use of IS-456, IS-800, SP-16, SP(6) – steel tables are permitted.  
3. Assume missing data suitably.*

Design slab and beam type combined footing for two columns of size 300mm × 300mm and 400 × 400mm subjected to 500kN and 700kN respectively. The centre to centre spacing between columns is 3.50m. The width of the footing is restricted to 1.5m. Take SBC of soil = 150kN/m<sup>2</sup>. Use M<sub>20</sub> and Fe415 grades. Also show reinforcement in L/S and C/S.

Solutions



- **Footing base dimensions**

Assuming  $\Delta P$ , the self-weight of the combined footing plus backfill to constitute 10 or 15 percent of the total column loads,

$$\Delta P = (700 + 1200) \times 15/100 = 285 \text{ kN}$$

$$P_1 + P_2 = 700 + 1200 = 1900 \text{ kN}$$

Allowable soil pressure or safe bearing capacity,  $q_a = 130 \text{ kN/m}^2$

$$\text{Area of the footing, } A_{req} = \frac{P_1 + P_2 + \Delta P}{q_a} = 16.8 \text{ m}^2$$



Width of footing,  $B = 2\text{m}$  ( Given in question)

$$\text{Total Length of footing, } L = \frac{A_{req}}{B} = \frac{16.8}{2} = 8.4 \text{ m}$$

- **Locate the point of application of the column loads**

In order to obtain a uniform soil pressure distribution, the line of action or point of application of the resultant column load must pass through the centroid of the footing.

Assuming a load factor of 1.5, the factored column loads are:

- $P_{u1} = 700 \times 1.5 = 1050 \text{ kN}$ ;  $P_{u2} = 1200 \times 1.5 = 1800 \text{ kN} \Rightarrow P_{u1} + P_{u2} = 2850 \text{ kN}$

Let  $\bar{x}$  be the centroid of the column loads, where  $s = 4.6 \text{ m}$

$$\Rightarrow \bar{x} = \frac{P_{u2} s}{P_{u1} + P_{u2}} =$$

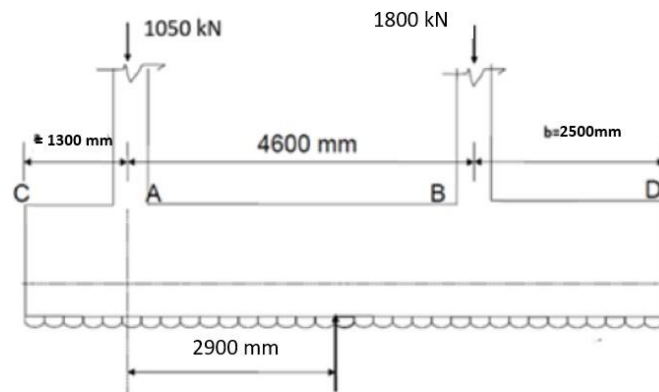
$$= \frac{2400 \times 4.6}{1050 + 2400} = 2.9 \text{ m}$$

If the cantilever projection of footing beyond column A is 'a' then,

$$a + 2.9 = L / 2 \Rightarrow a = 8.4 / 2 - 2.9 = 1.3 \text{ m}$$

Similarly, if the cantilever projection of footing beyond Column B is 'b' then,

$$b = 8.4 - 1.3 - 4.6 = 2.5 \text{ m}$$

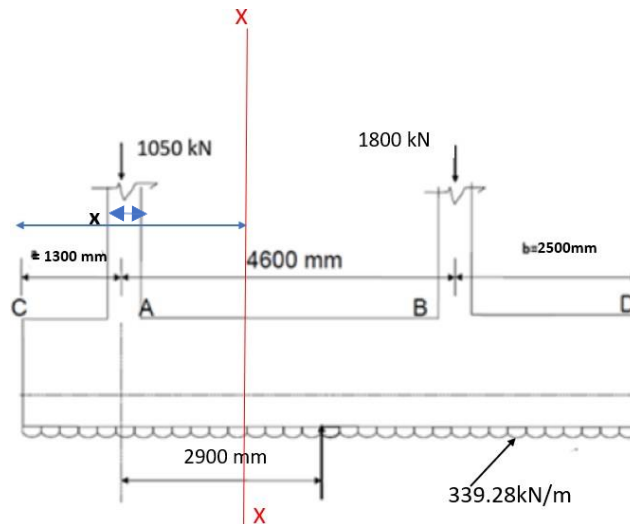


- **Uniformly distributed load acting in upward direction (soil pressure)**

Treating the footing as a wide beam ( $B = 2000 \text{ mm}$ ) in the longitudinal direction, the uniformly distributed load (acting upward) is given by  $q_{uB}$

$$q_{uB} = \frac{P_{u1} + P_{u2}}{L} = \frac{1050 + 1800}{8.4} = 339.28 \text{ kN/m}$$

- **Shear force calculations**

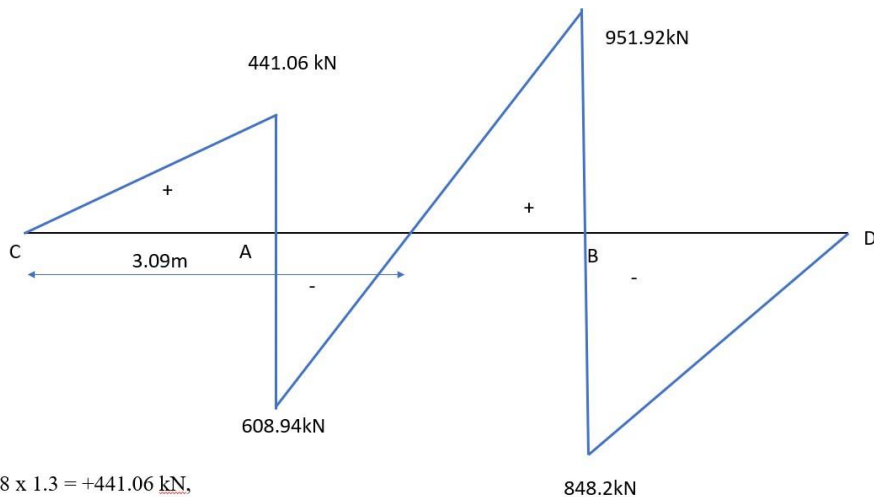


- Shear force at A, just before 1050 kN, left of section XX,  $V_{AC} = +339.28 \times 1.3 = +441.06$  kN
- Shear force at A, just after 1050 kN, left of section XX,  $V_{AB} = -1050 + 339.28 \times 1.3 = -608.94$  kN
- Shear force at B just after 1800kN, right of section XX,  $V_{BA} = +1800 - 339.28 \times 2.5 = +951.92$  kN
- Shear force at B just before 1800kN, right of section XX,  $V_{BD} = 339.28 \times 2.5 = -848.2$  kN
- **Location of zero shear, Left of section XX**

$$339.28 \times X - 1050 = 0, \text{ location of zero shear.}$$

$$339.28 \times X = 1050, 1050/339.28 = X, X = 3.09\text{m}$$

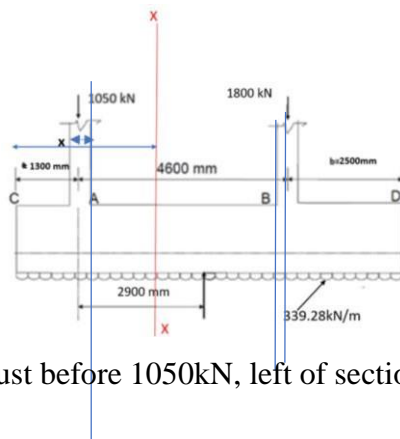
$$\underline{\underline{X = 3.09 \text{ m from C}}}$$



- $V_{AC} = 339.28 \times 1.3 = +441.06 \text{ kN}$ ,
- $V_{AB} = -1050 + 339.28 \times 1.3 = -608.94 \text{ kN}$
- $V_{BA} = 1800 - 339.28 \times 2.5 = +951.92 \text{ kN}$
- $V_{BD} = 339.28 \times 2.5 = -848.2 \text{ kN}$

### Shear force diagram

- **Bending moment calculations**

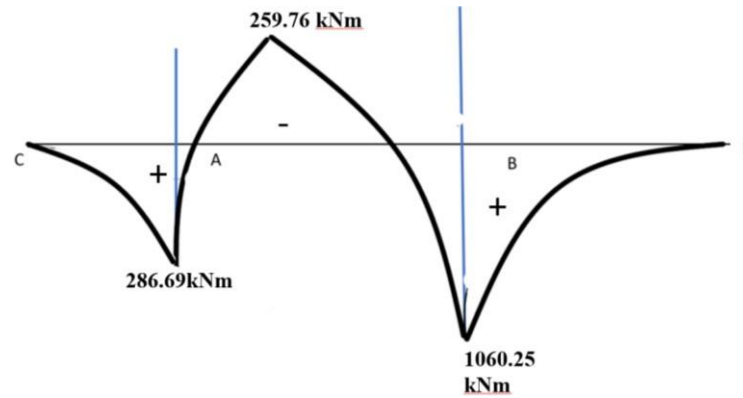


- BM at A, just before 1050kN, left of section XX,  $M_{AC} = 339.28 \times 1.3 \times 1.3 / 2 = +286.69 \text{ kNm}$
- BM at just at the inner face of Column A (1050kN), left of section XX,  

$$M_{AB} = -1050 \times 0.35/2 + 339.28 \times (1.3 + 0.35/2) \times (1.3 + 0.35/2)/2$$

$$= -1050 \times 0.35/2 + 339.28 \times (1.3 + 0.172) \times (1.3 + 0.172)/2 = +185.32 \text{ kNm}$$
- Negative Bending moment at  $X = 3.09 \text{ m}$  (Location of zero shear)  

$$M_u = 339.28 \times (3.09)^2/2 - 1050 \times (3.09 - 1.3) = -259.76 \text{ kNm}$$
- BM at B, just before 1800 kN, right of section XX =  $+339.28 \times 2.5^2/2 = +1060.25 \text{ kNm}$
- BM at B, just after the inner face of Column B (1800 kN), right of section XX =
- $339.28 \times (2.5 + 0.4/2)^2/2 - 1800 \times 0.4/2 = +876.67 \text{ kNm}$

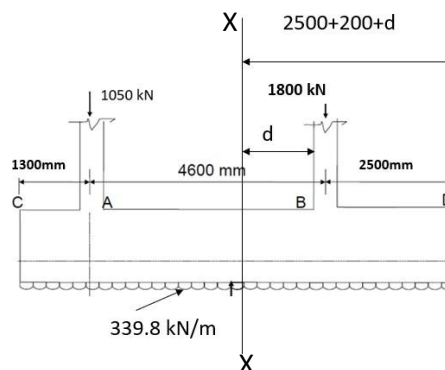


### Bending moment diagram

- **Thickness of footing or effective depth of footing based on shear**

*One-way shear* (longitudinal):  $V_{u1}$  calculate it at a distance “d” from the edge of the heavier column, where “d” is the effective depth of the footing.

The critical section (**always for column with heavier load**) for one-way shear is located at a distance  $d$  from the (**inner**) face of column B, and has a value



*Critical One-way shear force,  $V_{u1}$  at section XX (just right of XX section) =*

Column load (B) - Uniformly distributed upward load intensity  $\times$  (2500 + 200 +  $d$ )

$$= (1800 - 339.28 \times (2.5 + 0.200 + d)) = (882.54 - 339.28 \times d) \text{ kN} \dots (1)$$

Take  $\tau_c = 0.48 \text{ N/mm}^2$  (for M 20 concrete, **Assuming Percentage of steel** as,  $p_t = 0.50$ ) IS 456 2000, page 73, table 19

*Design shear strength of concrete,  $V_{uc} = \tau_c \times B \times d = 0.48 \times B \times d$*

Equate  $V_{uc}$  and  $V_{u1}$

*B is width of footing = 2000 mm*

$$V_{uc} = 0.48 \times 2000 \times d = (960d) \text{ N} \dots (2)$$



Equating one-way shear force and design shear strength of concrete, (1) = (2)

$$V_{u1} = V_{uc} \Rightarrow (882.54 \times 10^3 - 339.28 \times d) = 960d, 882.54 \times 10^3 = 1299.8 d$$

⇒ Effective depth of footing,  $d = 679.25$  mm **Rounded to 680 mm**

Use 20 mm  $\phi$  bars with a clear cover of **75 mm**, **Taking an overall depth or thickness of the footing**

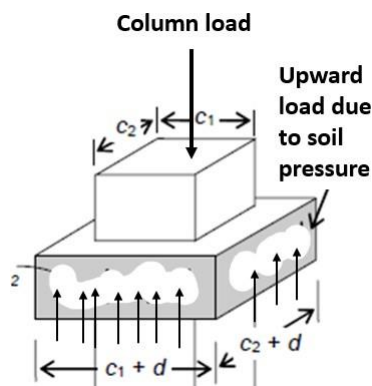
$$D = d + 75 + 20/2 = 680 + 75 + 20/2 = 765 \text{ mm}$$

- **Two-way shear force for columns A and B (Punching shear)**

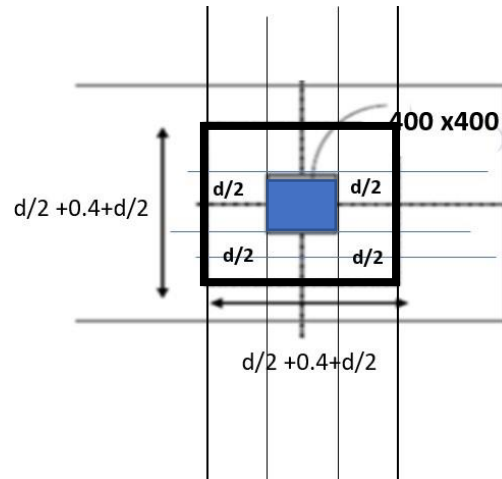
*Two-way shear or punching shear (we need to consider the upward soil pressure not upward soil intensity) \* Since it is acting on an area.*

Factored soil pressure or Upward soil pressure,  $q_u = (339.28) / (B \times 1) = (339.8/2) = 169.64 \text{ kN/m}^2$

The critical section is located  $d/2$  from the periphery of columns A and B.



**Shear stresses in footing slab due to punching shear**

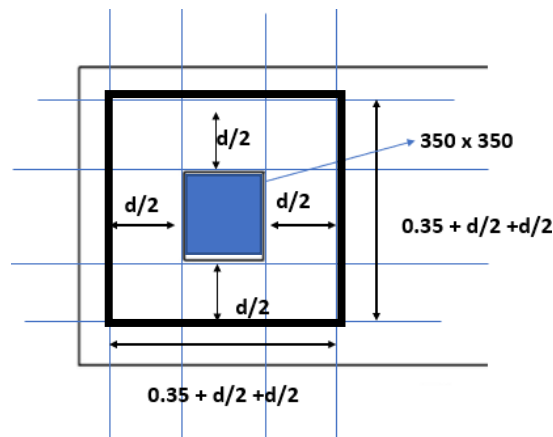


Punching shear or Two- way shear calculations for heavier Column B

$$V_{u2} = 1800 - 169.64 (0.4 + 0.680/2 + 0.680/2) \times (0.4 + 0.680/2 + 0.680/2)$$

$$= 1602.13 \text{ kN @ B (Heavier column)}$$

Punching shear or Two-way shear for Column A (350 mm x 350 mm)



*Punching shear or Two-way shear @ A,*

$$Two\text{-}way\ shear\ V_{u2} = (\text{Column load at A})\ 1050 - 169.64 \times (0.35 + 0.680/2 + 0.680/2) \times (0.35 + 0.680/2 + 0.680/2)$$

$$= 870 \text{ kN @ A (Lighter column)}$$

If no shear reinforcement is provided, **Page 58, IS 456, Clause 31.6.3.1**, calculated shear stress

at critical section shall not exceed  $k_s (0.25 \sqrt{f_{ck}})$

where

$k_s = (0.5 + \beta_c)$  but not greater than 1,  $\beta_c$  being the ratio of short side to long side of the column/capital; and

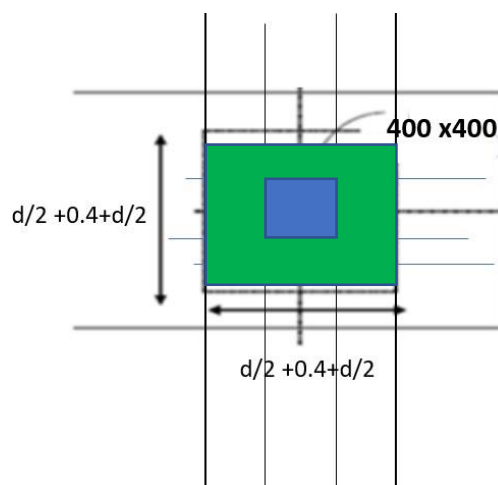
$\tau_c = 0.25 \sqrt{f_{ck}}$  in limit state method of design, and  $0.16 \sqrt{f_{ck}}$  in working stress method of design.

For square columns,  $k_s = (0.5 + \beta_c)$ ,  $\beta_c = 350/350 = 400/400 = 1.0$ ,  $k_s = (0.5 + 1)$  but it should not be greater than 1, hence  $k_s = 1$

Permissible shear stress,  $\tau_{c2} = k_s (0.25 \sqrt{f_{ck}}) = 1.0 \times 0.25 \times \sqrt{20} = 1.118 \text{ N/mm}^2$

Permissible two-way shear force for column B (*heavier column*)

Permissible two-way shear force,  $V_{uc} = \text{Permissible shear stress} \times (\text{Area of the footing slab enclosed by the perimeter of the critical section})$



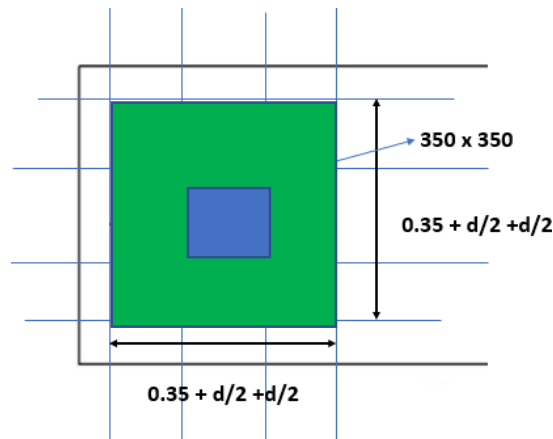
Perimeter of critical section (Green coloured area) =  $4 \times (400 + 680/2 + 680/2)$

$V_{uc} = 1.118 \times [4 \times (400 + 680/2 + 680/2)] \times 680 = 3284.24 \text{ kN @ B}$

In the similar way lets calculate for Column A

Permissible two-way shear force for Column A

$V_{uc} = 1.118 \times [(350 + 680/2 + 680/2) \times 4] \times 680 = 3132.18 \text{ kN @ A}$



Compare whether permissible two way shear force is greater than two shear way ( Actual) force

$$V_{uc} = 3284.23 \text{ kN} > V_{u2} = 1602.82 \text{ kN} \quad @ \text{ B It is safe.}$$

$$V_{uc} = 3132.18 \text{ kN} > V_{u2} = 870.00 \text{ kN} @ \text{ A . It is Safe.}$$

Hence safe against two way or punching shear, ( if not provide shear reinforcement- stirrups or bent up bars)

- **Design of longitudinal flexural reinforcement**

Maximum 'negative' moment: $M_u =$ - 259.76 kNm	Maximum 'positive' moment: $M_u = +$ 1060.25 kNm
$M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)$ <p> <math>M_u = 259.76 \times 10^6 \text{ N mm}</math>  <math>B = b = 2000 \text{ mm}, f_{ck} = 20 \text{ N/mm}^2,</math>  <math>f_y = 415 \text{ N/mm}^2</math>  <math>d = 680 \text{ mm}</math>  <math>D = 765 \text{ mm}</math>  <math>A_{st} \text{ provided} = 1075.67 \text{ mm}^2</math>            Check for <math>(A_{st})_{min} = 0.0012 BD =</math>  <math>0.0012 \times 2000 \times 765 = 1836 \text{ mm}^2</math> </p>	$M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)$ <p> <math>M_u = 1060.25 \times 10^6 \text{ N mm}</math>  <math>B = b = 2000 \text{ mm}</math>  <math>d = 680 \text{ mm}</math>  <math>A_{st} \text{ provided} = 4648.12 \text{ mm}^2</math>            Check for <math>(A_{st})_{min} = 0.0012 BD =</math>  <math>0.0012 \times 2000 \times 765 = 1836 \text{ mm}^2</math> </p>

$A_{st} \text{ provided} < (A_{st})_{min}$ , Hence provide  $(A_{st})_{min}$

But we have assumed  $p_t = 0.5$

$$p_t = 100 A_{st, req} / (B \times d)$$

Choose 16 mm diameter bars, calculate no of bars =  $1836 / (\pi/4 \times 16^2) = 10$

**Provide 10 # 16 mm diameter bars at top**

- **Development length  $L_d = 47 \times \text{dia of bar}$   
 $= 47 \times 16 = 752 \text{ mm}$**

Since  $p_t$  was assumed as 0.5 %, we need to calculate  $A_{st, req}$  by using

$$p_t = 100 A_{st, req} / (B \times d)$$

$$A_{st, req} = 0.50 \times 2000 \times 680 / 100 = 6800 \text{ mm}^2$$

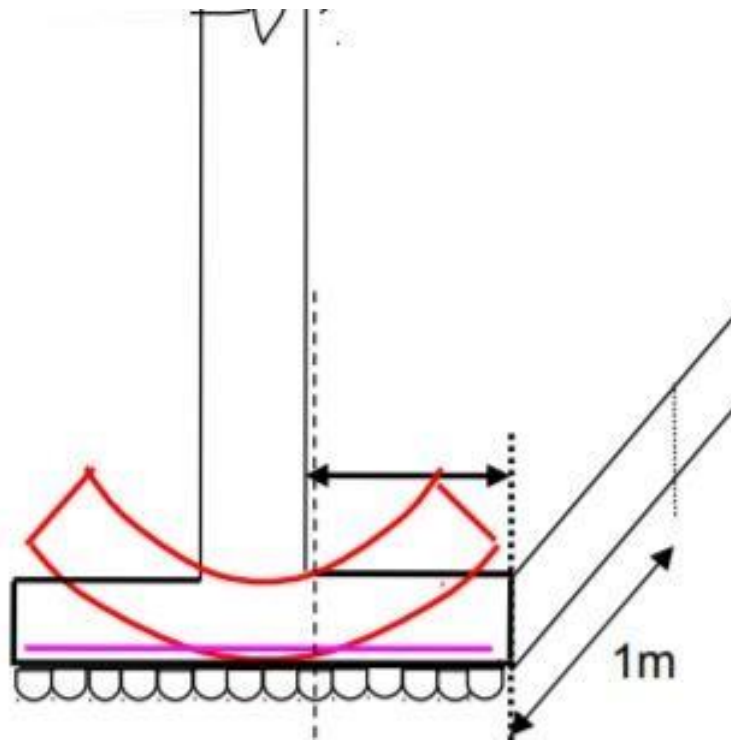
Since Moment is more, we are choosing 6800  $\text{mm}^2$  and going for larger diameter bars ie 20 mm

$$\begin{aligned} \text{No of 20 mm dia bars} &= 6800 / (\pi/4 \times 20^2) \\ &= 22 \end{aligned}$$

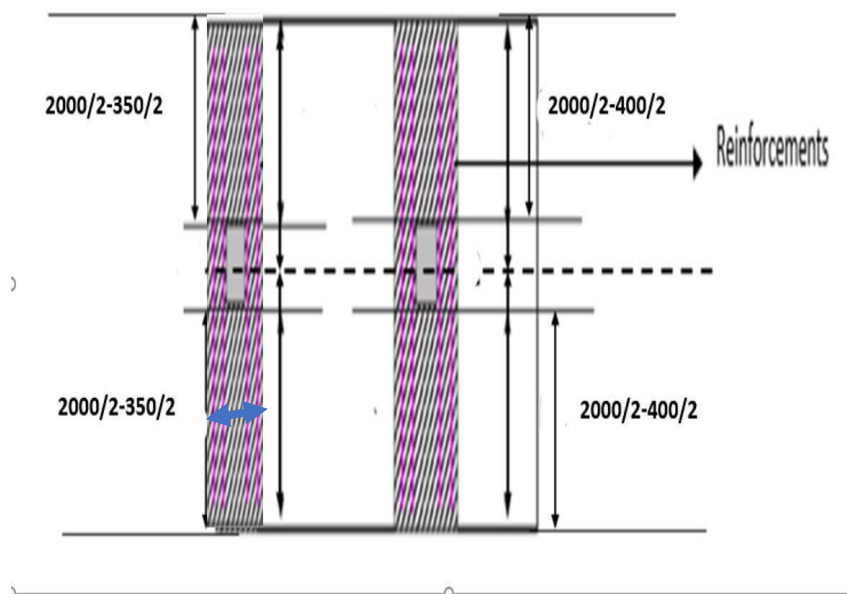
**Provide 22 # 20mm diameter bars at bottom**

- **Development length  $L_d = 47 \times \text{dia of bar}$   
 $= 47 \times 20 = 940 \text{ mm}$**

## Design of column strips as transverse beams



Transverse bending of footing





<i>Transverse beam under column A</i>	<i>Transverse beam under column B</i>
<ul style="list-style-type: none"> <li>• Factored Column load A per width of footing = <math>1050/2.0 = 525</math> kN/m</li> <li>• Cantilever Projection of beam beyond column face = <math>(2000 - 350)/2 = 825</math> mm = 0.825 m</li> <li>• Maximum transverse moment at column face A : <math>M_u = 525 \times 0.825^2/2 = 178.66</math> kNm</li> <li>• Effective depth for transverse beam (Assume 16 mm <math>\phi</math> bars as transverse reinforcement which is placed above the 20 mm <math>\phi</math> longitudinal bars): <math>d = 680 - 75 - 20 - 16/2 = 577</math> mm</li> <li>• Assume width of transverse beam, b = <math>\text{width of column} + 2 \times 0.75d</math> <math>b = 350 + 2 \times 0.75 \times 577 = 1215.5</math> mm <math>M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)</math></li> </ul>	<ul style="list-style-type: none"> <li>• Factored Column load B per width of footing = <math>1800/2.0 = 900</math> kN/m</li> <li>• Cantilever Projection beyond column face = <math>(2000 - 400)/2 = 800</math> mm = 0.800m</li> <li>• Moment at column face B = <math>900 \times 0.80^2/2 = 288</math> kNm</li> <li>• Width of transverse beam, b = <math>\text{width of column} + 2 \times 0.75d</math> <math>400 + 0.75 \times 577 + 0.75 \times 577 =</math> mm = 1265.5mm <math>M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)</math> <math>M_u = 384 \times 10^6</math> N mm b = 1265.5mm d = 577 mm A<sub>st</sub> = 1956.2 mm<sup>2</sup></li> </ul>

<p><math>b = 1215.5 \text{ mm}</math>, <math>d = 577 \text{ mm}</math>  <math>M_u = 178.6 \times 10^6 \text{ N mm}</math>  <math>A_{st} = 880.23 \text{ mm}^2</math></p>	
<p>Page 48, CL No 5.2.1</p> <ul style="list-style-type: none"> <li>Minimum <math>A_{st} = 0.0012 bD =</math>  <math>A_{st \text{ min}} = .0012 \times 1215.5 \times 765 = 1115.83 \text{ mm}^2</math></li> <li>Use 12mm dia bars (Your wish!!)</li> <li>Number of 12 mm <math>\phi</math> bars required =  <math>A_{st} / \text{area of one bar} = 1115.83 / (\pi/4 \times 12^2) = 10</math></li> </ul> <p><b>Provide 10 nos 12 mm <math>\phi</math> bars</b></p> <p>Check for development length = <math>47 \times 12 = 564 \text{ mm}</math></p>	<ul style="list-style-type: none"> <li>Provide <math>(A_{st})_{\text{min}} = 0.0012 \times 1265.5 \times 765 = 1161.73 \text{ mm}^2</math></li> </ul> <p>Use 12 mm dia bars</p> <p>Number of 12 mm <math>\phi</math> bars required  <math>= 1956.2 / (\pi/4 \times 12^2) = 17.29 = 18</math></p> <p><b>Provide 18 nos 12 mm <math>\phi</math> bars</b></p> <ul style="list-style-type: none"> <li>Required development length =  <math>47.0 \times 12 = 564 \text{ mm}</math> is available beyond the column face.</li> </ul>
<p><b>Transfer of force at column base -Column A</b></p>	<p><b>Transfer of force at column base Column B</b></p>
<ul style="list-style-type: none"> <li>Limiting bearing stress at</li> </ul> <p>IS 456 Page 65 , CL34.4</p> <p><b>34.4 Transfer of Load at the Base of Column</b></p> <p>The compressive stress in concrete at the base of a column or pedestal shall be considered as being transferred by bearing to the top of the supporting pedestal or footing. The bearing pressure on the loaded area shall not exceed the permissible bearing stress in direct compression multiplied by a value equal to <math>\sqrt{\frac{A_1}{A_2}}</math> but not greater than 2;</p> <p>where</p> <p><math>A_1</math> = supporting area for bearing of footing, which in sloped or stepped footing may be taken as the area of the lower base of the largest frustum of a pyramid or cone contained wholly within the footing and having for its upper base, the area actually loaded and having side slope of one vertical to two horizontal; and</p> <p><math>A_2</math> = loaded area at the column base.</p>	<ul style="list-style-type: none"> <li>Limiting bearing stress at</li> </ul> <p>Permissible bearing stress = <math>0.45 f_{ck} \sqrt{\frac{A_1}{A_2}}</math></p> <p>[<math>A_1 = 2000^2</math> , <math>A_2 = 400^2 \text{ mm}^2</math> ]</p> <p><math>\sqrt{\frac{A_1}{A_2}} = 5 &gt; 2</math>, <math>\sqrt{\frac{A_1}{A_2}} = 2</math></p> <p><math>= 0.45 \times 20 \times 2 = 18 \text{ MPa}</math></p> <p>Permissible bearing resistance = <math>18 \times 400^2 = 2880 \text{ kN}</math></p> <p><math>2880 \text{ kN} &gt; 1800 \text{ kN}</math> , Hence safe</p>

For working stress method of design the permissible bearing stress on full area of concrete shall be taken as  $0.25 f_{ck}$ ; for limit state method of design the permissible bearing stress shall be  $0.45 f_{ck}$ .

$$\text{Permissible bearing stress} = 0.45 f_{ck} \sqrt{\frac{A_1}{A_2}}$$

$$A_1 = 2000^2 \text{ (2000mm is footing width)}$$

$$A_2 = 350 \times 350 =$$

$$= \sqrt{\frac{A_1}{A_2}} < 2$$

$$= 5.71 > 2, \text{ Take } \sqrt{\frac{A_1}{A_2}} = 2$$

$$\begin{aligned} \text{Permissible bearing stress} &= 0.45 \times 20 \times 2 \\ &= 18 \text{ N/mm}^2 \end{aligned}$$

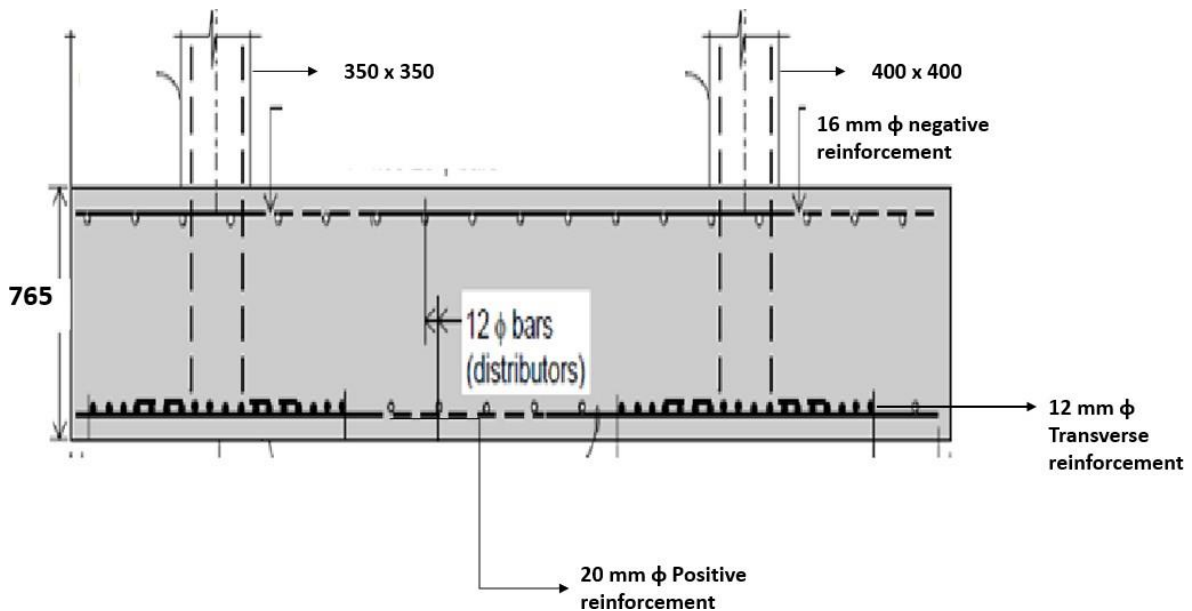
Permissible bearing resistance or force

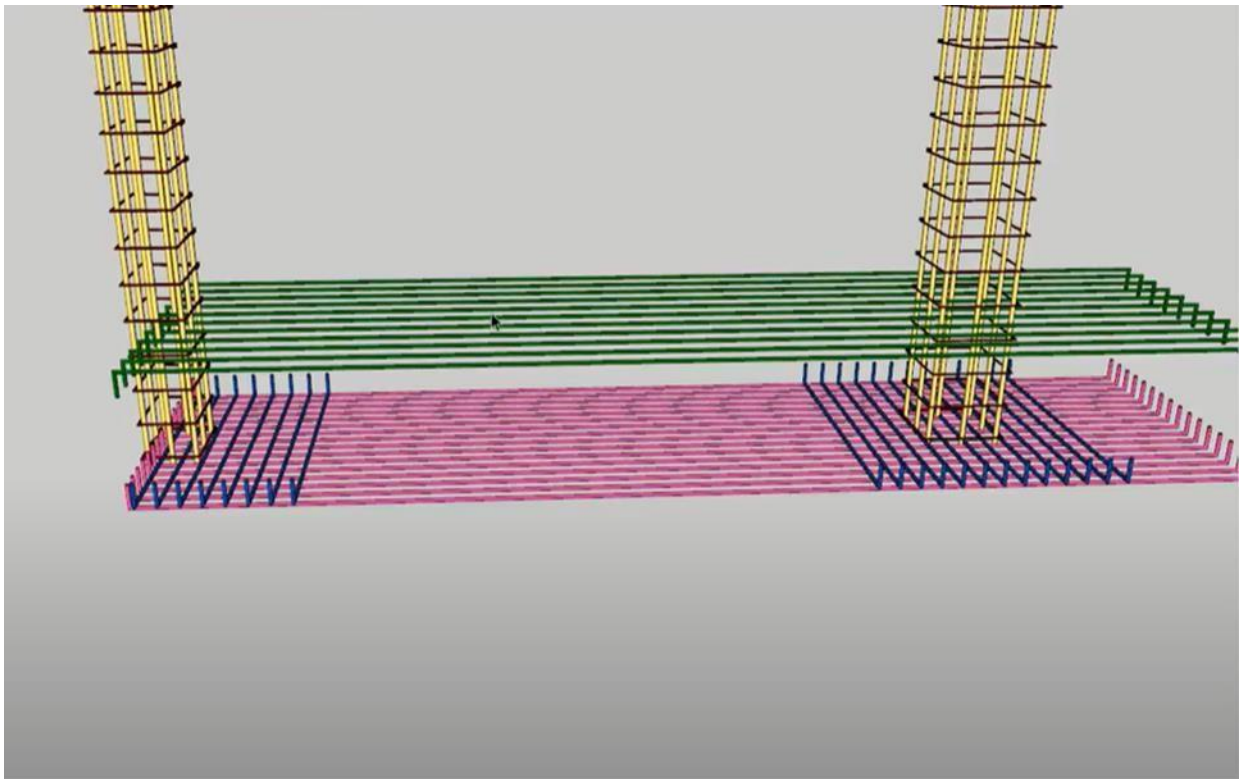
$$F_{br} = \text{Permissible bearing stress} \times \text{column area}$$

$$= 18 \times 350^2 = 2205 \times 10^3 \text{ N} = 2205 \text{ kN}$$

$2205 > 1050 \text{ kN}$ , Hence safe.

### Reinforcement detailing





OR

Design a cantilever retaining wall to retain an earth embankment 4m high above ground level. The density of earth is  $18\text{kN/m}^3$  and its angle of repose is  $30^\circ$ . The embankment is horizontal at top. The S.B.C. of soil is  $200\text{kN/m}^2$ . The coefficient of friction between soil and concrete is 0.5. Adopt M-20 and Fe415 grades. Draw C/S elevation of retaining wall. (50 Marks)

2. Design a cantilever retaining wall to retain soil of earth of height 4.0 m above ground level. The soil is having safe bearing capacity and density are respectively  $200\text{kN/m}^2$  and  $18\text{kN/m}^3$ . Design all components of retaining walls with all stability checks. Also draw detailed drawing. Use M 20 and Fe 415. Angle of Repose,  $\Phi = 30^\circ$ , Coefficient of friction between concrete and soil  $\mu = 0.5$ .

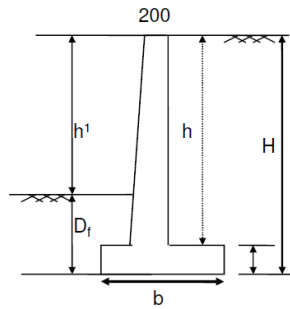
### Solution

Height of earth fill,  $h' = 4.0$  m, Safe bearing Capacity,  $\text{SBC} = 200 \text{ kN/m}^2$ , Density of soil,  $\gamma = 18 \text{ kN/m}^3$ , co-efficient of friction between concrete and soil,  $\mu = 0.5$ , angle of repose  $\phi = 30^\circ$

We need to fix the total height of retaining wall,  $H = h' + D_f$

- Depth of foundation,  $D_f$  – (Height of wall below Ground level)

Using Rankine's formula: find depth of foundation



$$D_f = \frac{SBC}{\gamma} \left[ \frac{1 - \sin \phi}{1 + \sin \phi} \right]^2 = \frac{SBC}{\gamma} k_a^2$$

$$\text{Active earth pressure coefficient } k_a = \frac{1 - \sin 30^\circ}{1 + \sin 30^\circ} = \frac{1}{3}$$

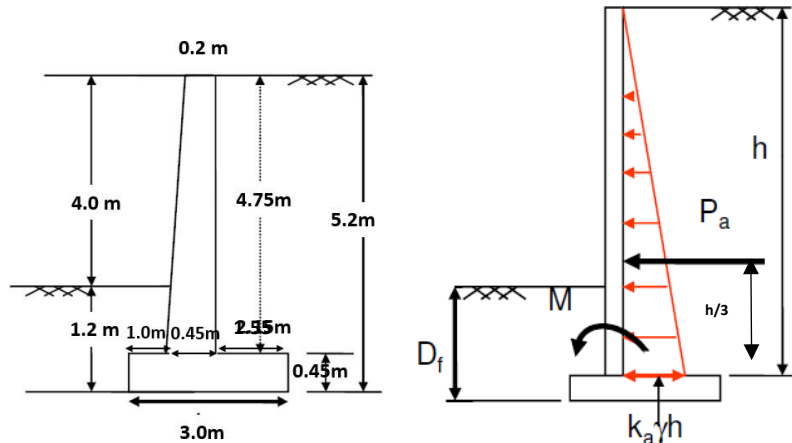
$$\text{Passive earth pressure coefficient } k_p = \frac{1}{k_a} = 3$$

$$= \frac{200}{18} \left[ \frac{1 - \sin 30^\circ}{1 + \sin 30^\circ} \right]^2 \mathbf{1.23 \text{ m} \approx \mathbf{1.2 \text{ m}}$$

Therefore total height of retaining wall  $H = 4.0 + 1.2 = 5.2 \text{ m}$

- Proportioning of wall

- Thickness of base slab** = (1/10 to 1/14) H = 1 / 10 × 5.2 to 1 / 14 × 5.2 = 0.52m to 0.37m, say 0.45 m - **450 mm**
- Width of base slab** = b = (0.5 to 0.6) H = 0.5 × 5.2 or 0.6 × 5.2 = 2.6 m to 3.12 m say **3.0 m**
- Toe projection** = (1/3 to 1/4) b = 1 / 3 × 3.0 or 1/4 × 3.0 = 1.0 m to 0.75 m say **1.0 m** (your wish !!)
- Provide **450 mm** thickness for the stem at the base (overall depth D) and **200 mm** at the **top**



- Design of stem

To find Maximum bending moment at the junction

Height of stem,  $h = 5.2 - 0.45 = 4.75 \text{ m}$

Active earth pressure acting on stem slab,  $P_a = \frac{1}{2} k \times \gamma \times h^2 = \frac{1}{2} k \times \gamma \times h^2$

$P_a = 1/2 \times 1/3 \times 18 \times 4.75 \times 4.75 = 67.68 \text{ kN}$

Total Bending moment at any height,  $M = P_a \times \frac{h}{3}$

$M = 14.25 \times \frac{4.75}{3} = 107.16 \text{ kN-m}$

$M_u = 1.5 \times M = 160.74 \text{ kN-m}$

Taking 1m length of wall,

We have overall depth at base or thickness of stem slab as,  $D = 450 \text{ mm}$

Check for effective depth “d”

$$M_{u,lim} = 0.36 \frac{x_{u,max}}{d} \left( 1 - 0.42 \frac{x_{u,max}}{d} \right) b d^2 f_{ck}$$

Put  $M_{u,lim} = 160.74 \times 10^6$ ,  $b = 1000 \text{ mm}$ ,  $f_{ck} = 20 \text{ N/mm}^2$

$x_{u,max} / d = 0.48$ , Fe 415, IS 456 2000

$160.74 \times 10^6 = 0.36 \times 0.48 \times (1 - 0.42 \times 0.48) \times 1000 \times d^2 \times 20$

$d = 241.3 \text{ mm} \approx 242 \text{ mm}$



effective cover = clear cover + bar diameter/2 (assuming 12 mm  $\phi$  bars)

$$= 40 + 12/2 = 46 \approx 50 \text{ mm}$$

d = Overall depth – effective cover = 450 – 50 = 400 mm  $\gg$  242 mm, hence safe

- Area of steel for stem slab

$$M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)$$

d = 400 mm, b = 1000 mm,  $M_u = 160.74 \times 10^6$  Nmm,  $f_y = 415$  N/mm<sup>2</sup>,  $f_{ck} = 20$  N/mm<sup>2</sup>

$A_{st} = 1184.86$  mm<sup>2</sup>

$A_{st, \min} = 0.0012 \times b \times D = 0.0012 \times 1000 \times 450 = 540$  mm<sup>2</sup>

$A_{st} > A_{st, \min}$ , hence Ok.

- Main steel

Provide 16 mm  $\phi$  bars as main steel

$$\text{Spacing required, } s = \frac{1000 \times \frac{\pi}{4} \times 16^2}{1184.86} = 169.69 \text{ mm} \approx 170 \text{ mm or } 160 \text{ mm (Your wish!!)}$$

Main steel #16 mm  $\phi$  @ 170 mm c/c < 300 mm or 3 times effective depth “d” (Check!!!)  
IS 456 2000

- Distribution steel or  $A_{st, \min}$

= 0.12% Gross Area =  $0.0012 \times 450 \times 1000 = 540$  mm<sup>2</sup>

Use 10 mm  $\phi$  bars, spacing required

$$\text{Spacing required, } s = \frac{1000 \times \frac{\pi}{4} \times 10^2}{540} = 145.4 \text{ mm} \approx 140 \text{ mm or } 150 \text{ mm (Your wish !!)}$$

Distribution bars #10 mm  $\phi$  @ 150 mm c/c < 450 mm and 5 times effective depth “d” ok  
(check!!!) IS 456 2000

- Development length  $L_d$

$$L_d = 47 \Phi_{\text{bar}} = 47 \times 16 = 752 \text{ mm} = 0.752 \text{ m}$$

- Curtailment of bars

Curtail 50% steel from top,  $A_{st} = \frac{50}{100} \times 1184.86 = 590$  mm<sup>2</sup>

$\left(\frac{h_1}{h}\right)^2 = \frac{1}{z}$ ,  $\left(\frac{h_1}{h}\right)^2 = \frac{1}{z}$ ,  $\frac{h_1^2}{5.2z} = \frac{1}{z}$ ,  $h_1 = 3.67$  m, is the curtailment length or cutting length

5.2

Actual point of cut off or cutting position =  $3.67 - L_d = 3.67 - 0.752 = 2.91$  m is the cutting length from top.

$$\text{Spacing required, } s = \frac{1000 \times \frac{\pi}{4} \times 16^2}{590} = 340.7 \text{ mm} \approx 340 \text{ mm As IS 456 spacing} < \underline{300 \text{ mm or}}$$

3 times effective depth “d” (Check!!!)

$s = 340 \text{ mm} > 300 \text{ mm}$

Instead of this you have to provide every alternate bars at 300 mm c/c.

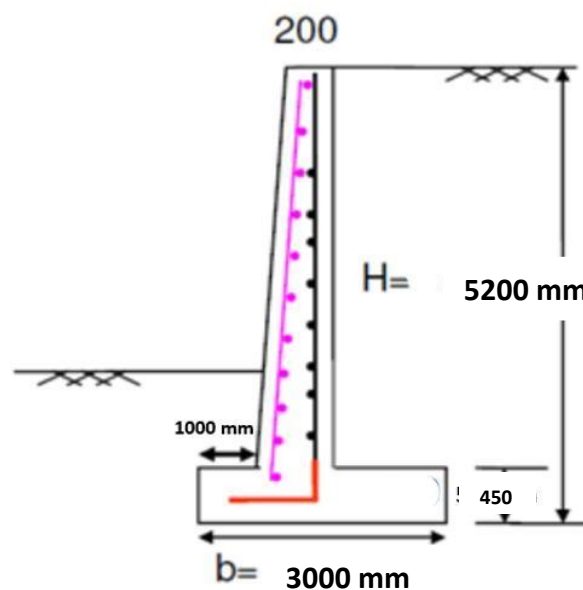
Spacing of bars 16 mm  $\phi$  @ 300 mm c/c. Hence it is ok.

### Secondary steel for stem at front (Temperature steel)

$$0.12\% \text{ Gross Area} = 0.12 \times 450 \times 1000/100 = 540$$

mm<sup>2</sup>

#10 @ 150 mm c/c < 450 mm and 5d Hence it is ok.



- Check for shear for stem slab

Max. Shear Force at Stem Junction,  $P_a = 67.68 \text{ kN}$  (Lateral earth pressure)

Ultimate Shear Force =  $V_u = 1.5 \times \text{[redacted]}$  kN

Nominal shear stress =  $\tau_v = V_u/bd = 101.52 \times 10^3 / (1000 \times 400) = 0.25 \text{ N/mm}^2$

To find  $\tau_c$ , calculate  $p_t = \frac{100 A_{st}}{b \times d} = \frac{100 \times 1184}{1000 \times 400} = 0.295 \%$

Use IS:456-2000, Page 73, Table 19,  $p_t = 0.295 \%$ , M 20

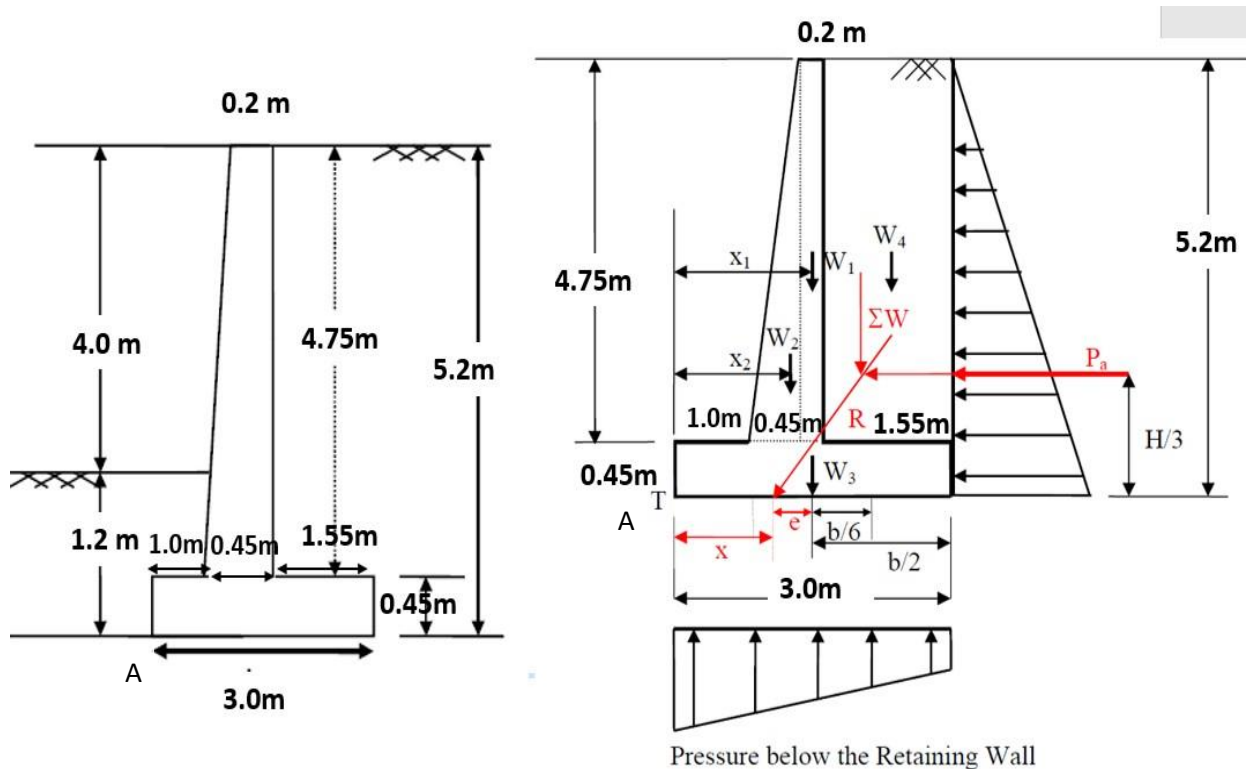
$$\tau_c = 0.379 \text{ N/mm}^2 \approx 0.38 \text{ N/mm}^2$$

Compare  $\tau_v$  and  $\tau_c$ , [redacted]

$\tau_v < \tau_c$  Hence safe in shear. No need of shear reinforcement.

### Stability analysis – 1. To find factor of safety against overturning

Calculations of **Resisting Moment**  $\Sigma M_R$  – **Self weight of wall and weight of earth fill retained by heel slab**



Load	Magnitude, kN	Distance from A, m	Bending moment about A kN-m
Stem W1	$0.2 \times 4.75 \times 1 \times 25 = 23.75$	$(1.0 + 0.25 + 0.2/2) = 1.35$	32.06
Stem W2	$\frac{1}{2} \times 0.25 \times 4.75 \times 1 \times 25 = 14.84$	$1.0 + \frac{2}{3} \times 0.25 = 1.16$	17.21
Base slab W3	$3.0 \times 0.45 \times 1 \times 25 = 33.75$	$3.0/2 = 1.5$	50.63
Back fill, W4	$1.55 \times 4.75 \times 1 \times 18 = 132.525$	$1.0 + 0.45 + 1.55/2 = 2.225$	294.857
<b>Total</b>	<b><math>\Sigma W = 204.86</math> kN</b>		<b><math>\Sigma M_R = 394.76</math> kN-m</b>

Calculations of Overturning Moment  $M_o$  – Lateral earth pressure about the base slab

Load	Magnitude, kN	Distance from A, m	Bending moment about A kN-m
Hori. earth pressure = $P_H$	$P_H = \frac{1}{2} \times \frac{1}{3} \times 18 \times 5.2^2 = 81.12$ kN	$H/3 = 5.2/3$	<b><math>M_o = 140.61</math></b>

**Stability checks:**

1. **Check for overturning:**

As per IS: 456:2000, (Factor of Safety) overturning should satisfy condition that  $\Sigma M_R / M_o > 1.55$

$\Sigma M_R = 394.757$  kNm,  $M_o = 140.61$  kNm

(F.S) overturning =  $\Sigma M_R / M_o = 2.80 > 1.55$  Hence it is safe

## 2. Check for

Sliding:  $\Sigma W = 204.86 \text{ kN}$

$P_H = 81.12 \text{ kN}$  ( Horizontal earth pressure)

As per IS: 456:2000, (F.S) sliding should satisfy condition that  $\mu \Sigma W / P_H \geq 1.55$

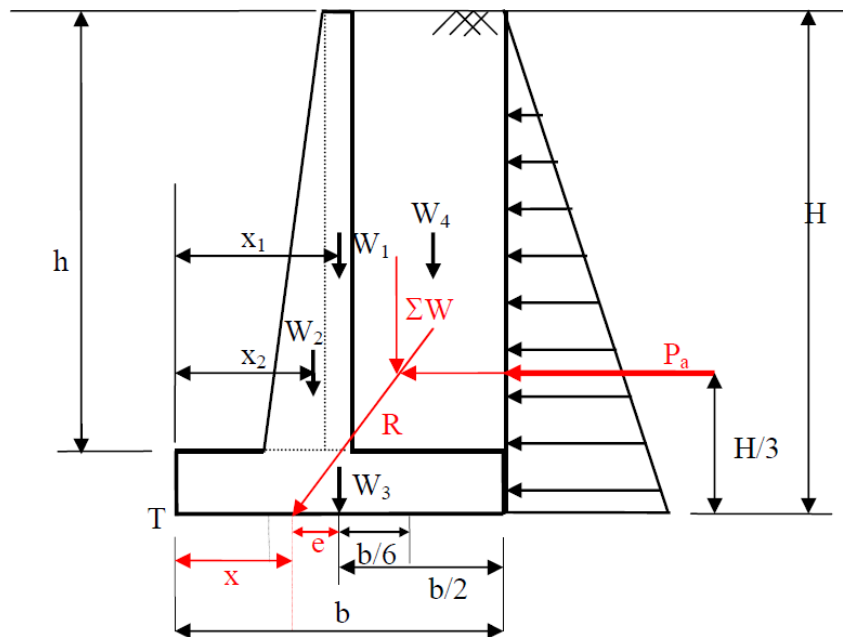
$$\frac{\mu \Sigma W}{P_H} = \frac{0.5 \times 204.86}{81.04} = 1.26$$

(F.S) sliding =  $1.26 \leq 1.55$  Hence it is not safe against sliding. Shear key is needed.

## 3. Check for subsidence: (Max. pressure at the toe should not exceed the safe bearing capacity of the soil under working condition)

Let the resultant cut the base at distance 'x' from toe T,

$x = \Sigma M / \Sigma W$ , where  $\Sigma M =$  Net moments about toe =  $\Sigma M_R - M_O = 394.757 - 140.61 = 254.15 \text{ kNm}$



$$x = \frac{254.15}{204.86} = 1.24 \text{ m}, b = 3 \text{ m}$$

- Eccentricity  $e = b/2 - x = 3/2 - 1.24 = 1.5 - 1.24 = 0.26 \text{ m} < b/6$ ,  $0.26 < 0.5$
- $e = 0.26 \text{ m}$
- (Eccentricity of force should not exceed one sixth of base)

Here  $e < b/6$ . Hence it is safe.

## Pressure below the base slab

$$\text{Max. pressure} = P_{\max} = \frac{\Sigma W}{b} \left[ 1 + \frac{6e}{b} \right]$$

$103.79 \text{ kN/m}^2 < \text{SBC}$ , safe bearing capacity

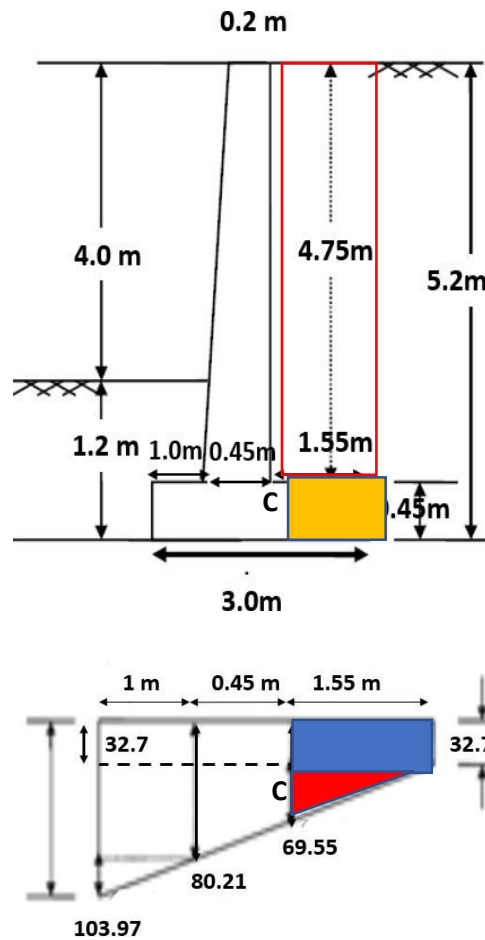
$$\text{Min. pressure} = P_{\min} = \frac{\sum W}{b} \left[ 1 - \frac{6e}{b} \right]$$




$32.7 \text{ kN/m}^2 > \text{zero}$ , So there is no tension or separation developed at base slab, Hence it is safe

Both values of pressure are lesser than SBC (  $200 \text{ kN/m}^2$  ) . Hence it is safe.


### Design of Heel Slab

Calculations of Moment about heel slab C



Load	Magnitude, kN	Distance from C, m	BM, Mc, kN-m
Backfill or earthfill 	$1.55 \times 4.75 \times 1 \times 18 = 132.52$	$1.55/2 = 0.775$	102.703
Heel slab 	$0.45 \times 1.55 \times 25 \times 1 = 17.43$	$1.55/2 = 0.775$	13.51
Upward Pressure distribution, (below heel slab) rectangle 	$-32.7 \times 1.55 = -50.68$	$1.55/2 = 0.775$	-39.28



Upward Pressure distribution, Triangle 	$-\frac{1}{2} \times (69.55 - 32.7) \times 1.55 = -28.55$	$\frac{1}{3} \times 1.55 = 0.516$	-14.73
<b>Total Load at junction C</b>	70.72	<b>Total BM at Junction C</b>	<b><math>\Sigma M_C = 62.18</math></b>

$$\Sigma M_C = 62.18 \text{ kNm}$$

$$M_u = 1.5 \times 62.18 = 93.27 \text{ kNm}$$

$$M_u = 0.87 f_y A_{st} d \left( 1 - \frac{A_{st} f_y}{bd f_{ck}} \right)$$

$$M_u = 62.18 \times 10^6 \text{ Nmm}, b = 1000 \text{ mm}, d = 400 \text{ mm}, f_{ck} = 20 \text{ N/mm}^2, f_y = 415 \text{ N/mm}^2$$

$$A_{st} = 669.04 \text{ mm}^2$$

Use 12 mm  $\phi$  bars ( it is base slab) ( You can choose 12 mm also)

$$\text{Spacing required, } s = \frac{1000 \times \frac{\pi}{4} \times 12^2}{669.04} = 170$$

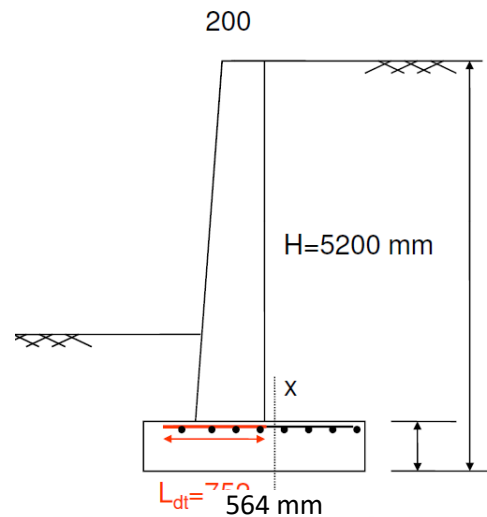
Main steel #12 mm  $\Phi$  @170 mm c/c < 300 mm and 3d ok. Hence it is safe.

### Development length

$$L_d = 47 \phi_{\text{bar}} = 47 \times 12 = 564 \text{ mm}$$

### Distribution steel

#10 mm  $\Phi$  @ 140 mm c/c < 450 mm and 5d ok



### Check for shear at junction (Tension)

Critical section for shear is at the face as it is subjected to tension.

Maximum shear  $= V = 70.72 \text{ kN}$ ,  $V_U, \text{ max} = 70.72 \times 1.5 = 106.08 \text{ kN}$

$$\tau_v = \frac{V_U}{b \times d} = \frac{106.08 \times 10^3}{1000 \times 400} = 0.26$$

$$pt = \frac{100 \times 669.04}{1000 \times 400} = 0.167\%$$

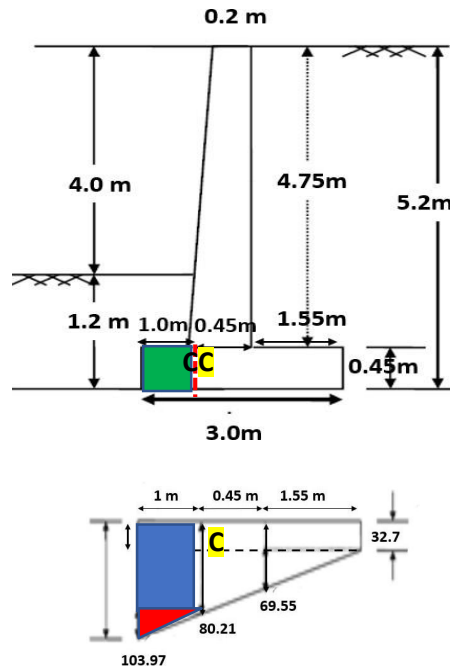
Use IS:456-2000, Page 73, Table 19,  $pt = \%$

$$\tau_c = 0.297 \text{ N/mm}^2$$

Compare  $\tau_v$  and  $\tau_{uc}$ ,  $\tau_v < \tau_c$  Hence it is safe in shear.

## Design of toe

To find the maximum bending moment



Load	Magnitude, kN	Distance from C, m	BM, $M_C$ , kN-m
Self wt Toe slab <span style="color: green;">■</span>	$1.0 \times 0.45 \times 25 = 11.25$	$1.0/2$	5.625
Upward Pressure distribution, rectangle	$-80.21 \times 1.0 = -80.21$	$1.0/2$	-40.10
Upward Pressure distribution, Triangle <span style="color: red;">▲</span>	$-\frac{1}{2} \times (103.97 - 80.21) \times 1.0 = -11.88$	$\frac{2}{3} \times 1 = 0.66$	-7.92
<b>Total Load at junction</b>		<b>Total BM at junction</b>	<b><math>\Sigma M_C = -42.4</math></b>

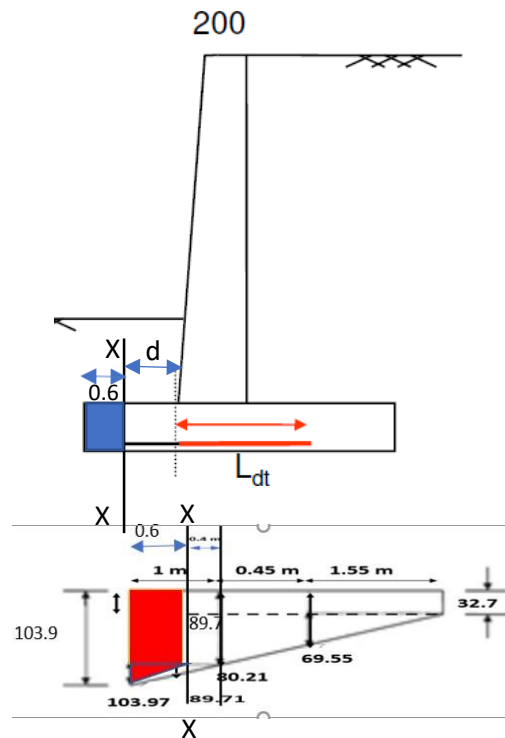
$$M_u = 1.5 \times 42.4 = 63.75 \text{ kN-m,}$$

$A_{st} = 452.02 \text{ mm}^2$ , since Area of steel is less, we have used 10 mm  $\Phi$  bars

Provide Main steel #10 @ 150 mm c/c < 300 mm and 3d ok

Development length:

$$L_d = 47 \phi_{bar} = 47 \times 10 = 470 \text{ mm}$$



Check for shear: at  $d$  from junction,  $d = 400$  mm

Net shear force at the section XX

$$V = -(103.9 + 89.7) / 2 \times 0.6 + 0.45 \times 0.6 \times 25 = -51.33 \text{ kN}$$

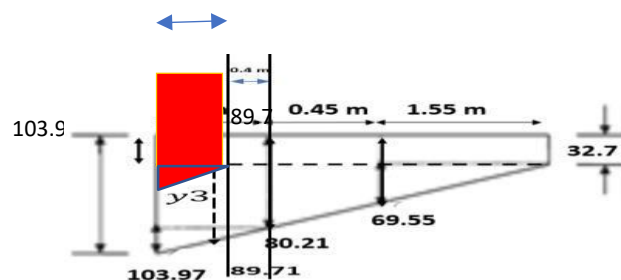
$$V_{U,max} = 51.33 \times 1.5 = 76.9 \text{ kN}$$

$$\zeta_v = 76.9 \times 1000 / (1000 \times 400) = 0.19 \text{ MPa,}$$

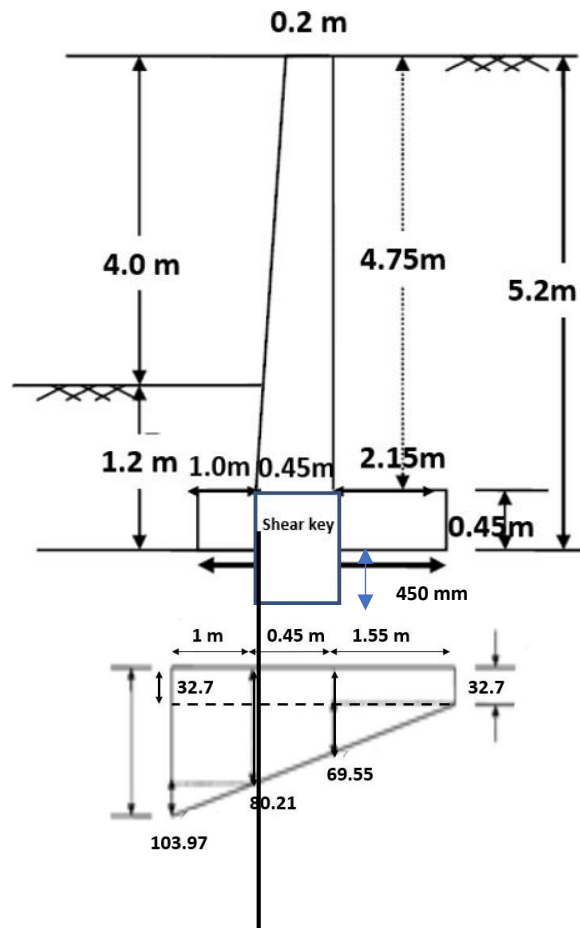
$$\text{find } p_t = 100 \times A_{st} / (1000 \times 400) = 0.113$$

From IS:456-2000,  $\zeta_c = 0.28$  MPa

$\zeta_v < \zeta_c$ , Hence safe in shear.



- Design of Shear key



Assume width and height of shear key as 450 mm. Let  $P_p$  be Total passive force developed in the front of shear key.

$$P_p = \text{Pressure at junction of toe} \times \text{width of shear key} \times \text{coefficient of passive pressure} \times 1\text{m}$$

$$= 80.21 \times 0.45 \times 3 \times 1 = 108.28\text{kN}$$

Factor of safety against sliding

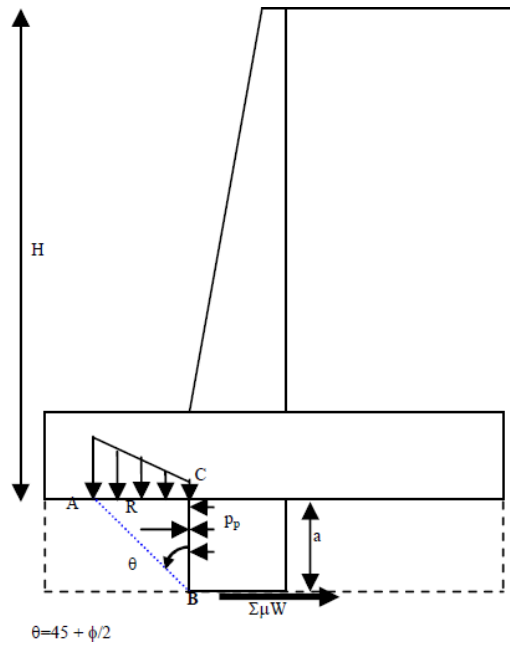
$$\text{FOS} = \frac{P_p + \mu \sum W}{P_A} = \frac{108.28 + 0.6 \times 204.86}{81.12} = 2.85 \geq 1.55$$

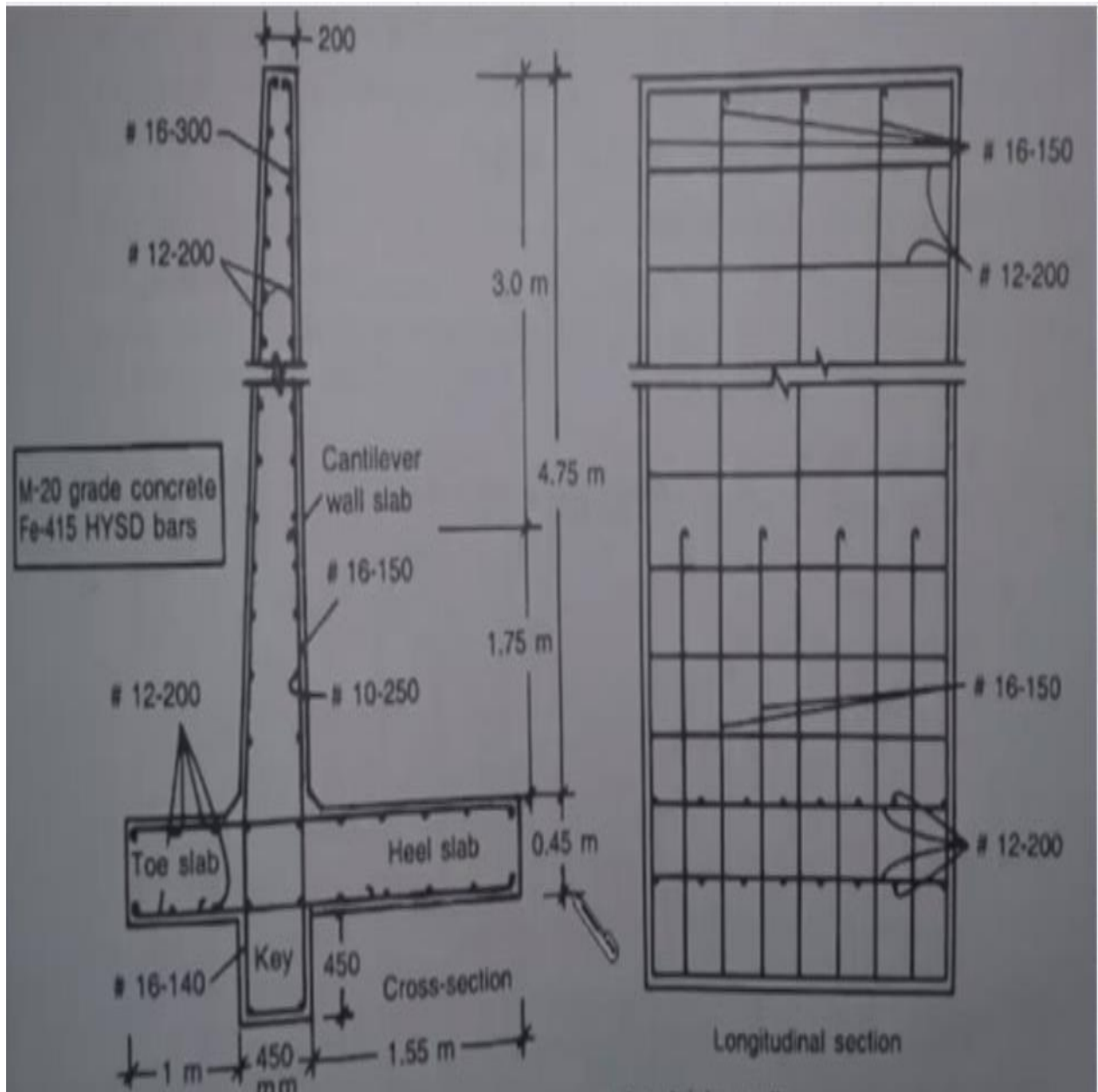
Hence safe

Provide 0.3 % of cross-sectional area for shear key as reinforcement

$$= \frac{0.3}{100} \times 450 \times 1000 = 1350\text{mm}^2$$

Provide 16mm dia @ 140mm/c





Reinforcement detailing of Retaining wall with shear key

**Module-2**

- 3 A line diagram of a roof truss with internal loads and forces in each members are shown in Fig.Q.3. Design the various members of the roof truss along with their end connection with bolt using property class 5.6 black bolts. Also design the bearing plate at the support for the reaction and anchor bolts for an uplift force of 15kN. Draw elevation of truss greater than half span. (50 Marks)

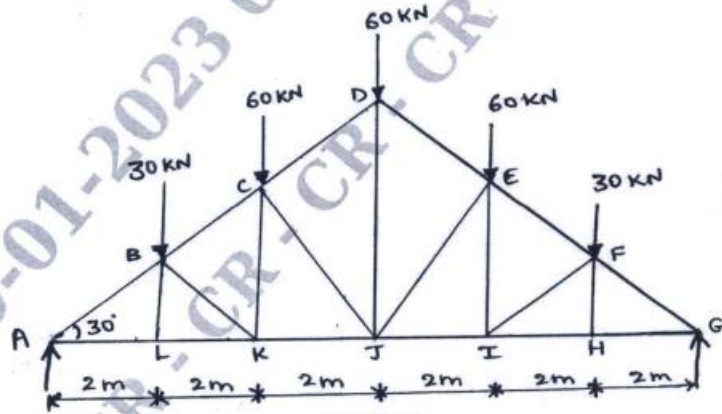


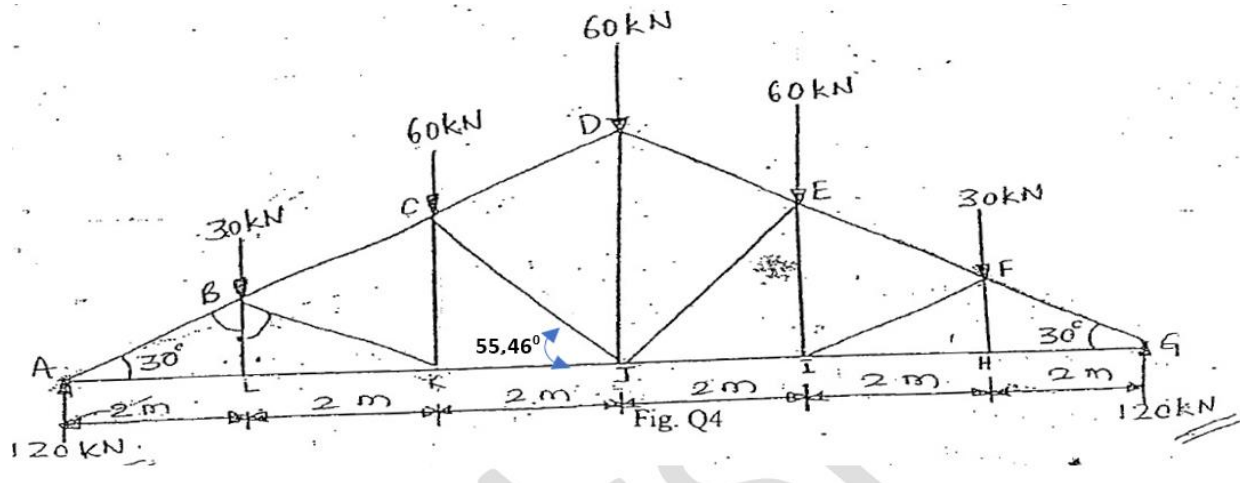
Fig.Q.3

Tabulation of member forces

Members	Length (m)	Force (kN)	Nature of Force
AB, GF	2.31	240.00	Compression
BC, FE	2.31	210.00	Compression
CD, ED	2.31	160.04	Compression
AL, GH	2.00	207.84	Tension
LK, HI	2.00	207.84	Tension
KJ, IJ	2.00	181.82	Tension
BL, FH	1.154	0.00	-
BK, FI	2.31	30.00	Compression
CK, EI	2.31	15	Tension
CJ, EJ	3.05	66.05	Compression
DJ	3.46	66.00	Compression



- Design a roof truss with tabulation of member forces as shown in Figure. Design various member of roof truss along with their end connections with a gusset plate of 10 mm thick. The truss rest on 300 x 500 mm column made of M 20 grade concrete. Design the support bearing base plate for a reaction of 120 kN and anchor bolt for an uplift force of 18 kN. Draw to suitable scale
  - Elevation of truss greater than half span
  - Enlarged view of support joint
  - Elevation of upper joint of truss.



Member	Force	Nature of force (N)
AB, GF	240 kN	Compression, C
BC, FE	210 kN	C
CD, ED	160 kN	C
AL, GH	208 kN	Tension, T
LK, HI	208 kN	T
KJ, IJ	182 kN	T
BL, FH	0	-
BK, FI	30 kN	C
CK, EI	18 kN	T
CJ, EJ	66 kN	C
DJ	60 kN	C

#### Important points to be noted

- Top and Bottom chord – Two angles or Double angles
  - Inner members - Single angles
  - Use at least two bolts
  - Gusset plate thickness should be uniform
- Design of Top chord members (AB, BC, CD) (Compression members)**  
 AB, BC, CD are top chord member, maximum load is 240 kN  
 Factored load or Force =  $240 \times 1.5 = 360$  kN

Length of members AB/ BC/ CD

$$\cos 30 = 2/ AB , AB = 2/ \cos 30 = 2.3 \text{ m} = 2300 \text{ mm}$$

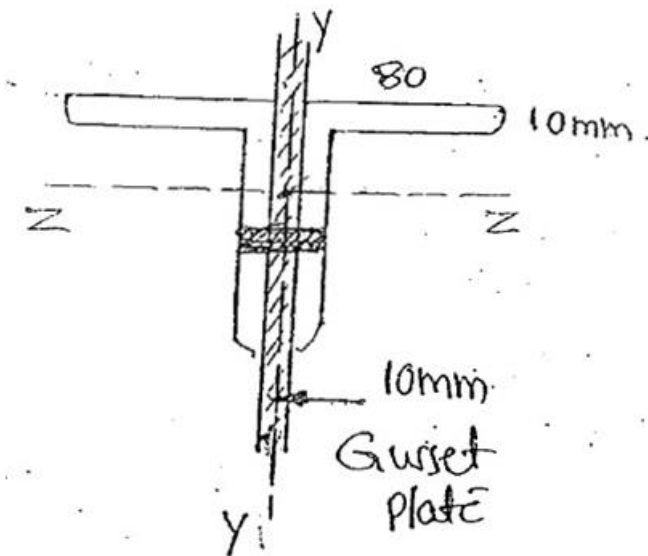
Assume design compressive stress,  $f_{cd} = 110 \text{ N/mm}^2$

(Assume  $f_{cd} = 40 - 120 \text{ N/mm}^2$  based on load and experience)

$$\text{Gross area, } A_g = \frac{\text{Force}}{f_{cd}} = \frac{360 \times 10^3}{110} = 3272.7 \text{ mm}^2 = 32.73 \text{ cm}^2$$

- Select double angle section from steel table

From Steel Table 6, Page 18, Try 2 ISA 80 x 80 x 10 mm



Area =  $30.10 \text{ cm}^2$  ,  $r_{xx} = 2.41 \text{ cm}$  Taking gusset plate of 10 mm thickness,

$$r_{yy} = 3.73 \text{ cm}$$

$$r_{min} = 2.41 \text{ cm or } 24.1 \text{ mm}$$

- Effective length of section - Page 48, CL7.5.2.1, IS 800- 2007

### 7.5.2 Double Angle Struts

**7.5.2.1** For double angle discontinuous struts, connected back to back, on opposite sides of the gusset or a section, by not less than two bolts or rivets in line along the angles at each end, or by the equivalent in welding, the load may be regarded as applied axially. The effective length,  $KL$ , in the plane of end gusset shall be taken as between 0.7 and 0.85 times the distance between intersections, depending on the degree of the restraint provided. The effective length,  $KL$ , in the plane perpendicular to that of the end gusset, shall be taken as equal to the distance between centres of intersections. The calculated average compressive

48

Effective length for Top chord members (AB) ,  $L_{eff} = 0.85 \times L = 0.85 \times 2.3 = 1.955 \text{ m}$

$$\text{Slenderness ratio, } \lambda = \frac{L_{eff}}{r_{min}} = \frac{1955}{24.1} = 81.12$$

Since it is a built-up member it will come under buckling class “c” (IS 800 – 2007, Page 44, Table 10). If it is class “c”, Use Table 9(c)

From Table 9 (c) Page 42 – IS 800 2007, through interpolation, for  $\lambda = 81.12$  we get design Compressive Stress,  $f_{cd} = 134.32 \text{ N/mm}^2$

(Page 34) Design compressive strength,  $P_c = f_{cd} \times A_g = 134.32 \times 3010$   
 $= 404.3 \text{ kN} > 360 \text{ kN}$

So selected section is safe.

- **Connections**

Using M 22 Property Class 5.6 bolts

- **Shear strength of bolts - Page 75 CL10.3.3 , IS 800 2007**

Assume fully threaded bolts, number of shear planes  $n_n = 2$  ,  $n_s = 0$  ( no shank portion)

$$A_{nb} = 0.78 \times \frac{\pi}{4} \times 22^2 = 296.5 \text{ mm}^2 , A_{sb} = 0, f_b = 500 \text{ N/mm}^2, \gamma_{mb} = 1.25$$

$$V_{dsb} = \frac{V_{nsb}}{\gamma_{mb}} = \frac{f_{ub}}{\sqrt{3} \gamma_{mb}} (n_n A_{nb} + n_s A_{sb})$$

$$= \frac{500}{\sqrt{3} \times 1.25} (2 \times 296.5) = 136.94 \text{ kN} \dots (1)$$

- **Bearing strength of bolts - Page 75 CL10.3.4 , IS 800 2007**

Pitch,  $p = 2.5 \times d = 2.5 \times 22 = 55 \text{ mm}$

Edge distance  $e = 1.7 \times d_o = 1.7 \times 24 = 40.8 \approx 45 \text{ mm}$  ( $d_o$  is the dia of bolt hole,  $(22 + 2)$ )  
 where  $k_b =$  smaller of  $\frac{e}{3d_0}$ ,  $\frac{p}{3d_0} - 0.25$ ,  $\frac{f_{ub}}{f_u}$ , and 1.0

$f_{ub} = 500 \text{ N/mm}^2, f_u = 410 \text{ N/mm}^2, t = 10 \text{ mm}, d = 22 \text{ mm}$

$$k = \frac{45}{3 \times 24} = 0.63, k = \frac{55}{3 \times 24} - 0.25 = 0.513, \frac{f_{ub}}{f_u} = \frac{500}{410} = 1.22, 1.0$$

$$V_{dpb} = 2.5 \times 0.513 \times 22 \times 10 \times \frac{410}{1.25} = 92.54 \text{ kN} \dots (2)$$

Bolt value = Minimum of (1) and (2) = 92.54 kN

No of bolts =  $\frac{360}{92.54} = 3.8 \approx 4$

Hence provide 2 ISA 80 × 80 × 10 mm with 4 bolts

- Design of Bottom chord members (AL, LK, KJ)-Tension members

Taking Max Force = 208 kN

Factored Tensile Force  $T_{dg} = 208 \times 1.5 = 312$  kN

Tensile strength due to gross section yielding, Page 32, CL 6.2 (IS 800)

$$T_{dg} = \frac{A_g f_y}{\gamma_{m0}}$$

$$\text{Gross area, } A_g = \frac{312 \times 10^3 \times 1.1}{250} = 1372.8 \text{ mm}^2$$

Since it is bottom member, increase the area by 30% =  $1.3 \times 1372.8 = 1784.64 \text{ mm}^2 = 17.84 \text{ cm}^2$

ISA 8080	80 x 80	6.0	14.6	143.2	18.58	112.0	19.2	2.46
		8.0	19.2	188.4	24.42	145.0	25.2	2.44
		10.0	23.6	231.5	30.10	175.4	31.0	2.41
		12.0	28.0	274.7	35.62	203.8	36.6	2.39

Try 2 ISA 80× 80 × 6 mm ( two angles back to back) with 10 mm gap

$$A_g = 18.58 \text{ cm}^2 = 1858 \text{ mm}^2$$

- **Connections**

Use M 22, class 5.6 ( Same bolt diameter for all connection design)

- **Shear strength of bolts**

Assume fully threaded bolts, number of shear planes  $n_n = 2$ , ( double angle),  $n_s = 0$  ( no shank portion)

$$A_{ns} = 0.78 \times \frac{\pi}{4} \times 22^2 = 296.5 \text{ mm}^2, A_{sb} = 0$$

$$V_{dsb} = \frac{V_{nsb}}{\gamma_{mb}} = \frac{f_{ub}}{\sqrt{3} \gamma_{mb}} (n_n A_{nb} + n_s A_{sb})$$

$$= \frac{500}{\sqrt{3} \times 1.25} (2 \times 296.5) = 136.94 \text{ kN} \dots (1)$$

- **Bearing strength of bolts**

$$V_{dpb} = 2.5 k_b d t \frac{f_u}{\gamma_{mb}}$$

$$V_{dpb} = 2.5 \times 0.513 \times 22 \times 6 \times \frac{410}{1.25} = 55.53 \text{ kN} \dots (2)$$

Bolt value = 55.53 kN (least of (1) or (2))

$$\text{No of bolts} = \frac{312}{55.53} = 6$$

Hence provide 2 ISA 80 × 80 × 6 mm with 6 bolts

(In case of High Strength Friction Grip Bolts (HSFG) - Shear capacity only needsto be calculated by CL 10.4.3, page 76 and then calculate no of bolts based on shearcapacity) No need of calculating "bearing strength of bolts".

$V_{dsf} = V_{nsf} / \gamma_{mf}$   
 $V_{nsf}$  = nominal shear capacity of a bolt as governed by slip for friction type connection, calculated as follows:

$$V_{nsf} = \mu_f n_e K_h F_o$$

where

$\mu_f$  = coefficient of friction (slip factor) as specified in Table 20 ( $\mu_f = 0.55$ ),

$n_e$  = number of effective interfaces offering frictional resistance to slip,

$K_h$  = 1.0 for fasteners in clearance holes,  
 = 0.85 for fasteners in oversized and short slotted holes and for fasteners in long slotted holes loaded perpendicular to the slot,

= 0.7 for fasteners in long slotted holes loaded parallel to the slot,

$\gamma_{mf}$  = 1.10 (if slip resistance is designed at service load),

= 1.25 (if slip resistance is designed at ultimate load),

$F_o$  = minimum bolt tension (proof load) at installation and may be taken as  $A_{nb} f_o$ ,

$A_{nb}$  = net area of the bolt at threads, and

$f_o$  = proof stress (=  $0.70 f_{ub}$ ).

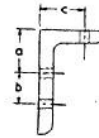
## To find partial safety factor

Table 5 Partial Safety Factor for Materials,  $\gamma_m$

(Clause 5.4.1)

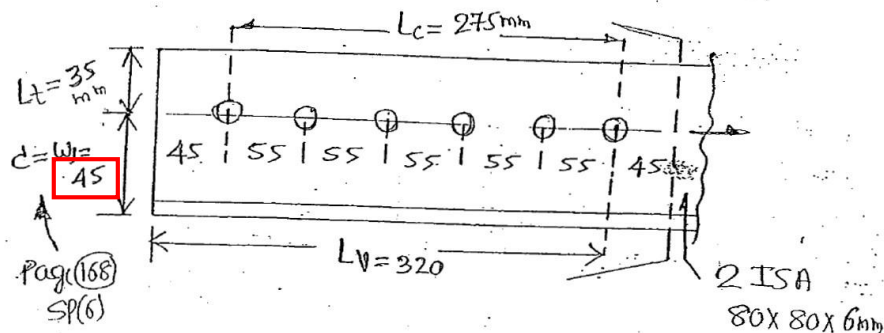
Sl No.	Definition	Partial Safety Factor	
		Shop Fabrications	Field Fabrications
i)	Resistance, governed by yielding, $\gamma_{m0}$	1.10	
ii)	Resistance of member to buckling, $\gamma_{m1}$	1.10	
iii)	Resistance, governed by ultimate stress, $\gamma_{m2}$	1.25	
iv)	Resistance of connection:		
a)	Bolts-Friction Type, $\gamma_{m3}$	1.25	1.25
b)	Bolts-Bearing Type, $\gamma_{m4}$	1.25	1.25
c)	Rivets, $\gamma_{m5}$	1.25	1.25
d)	Welds, $\gamma_{m6}$	1.25	1.50

TABLE XXXI RIVET GAUGE DISTANCES IN LEGS OF ANGLES



Leg Size	Double Row of Rivets		Single Row of Rivets c	Maximum Rivet Size for Double Row of Rivets
	a	b		
mm	mm	mm	mm	mm
200	75	85	115	27
150	55	65	90	22
130	50	55	80	20
125	45	55	75	20
115	45	50	70	12
110	45	45	65	12
100	40	40	60	12
95	—	—	55	—
90	—	—	50	—
80	—	—	45	—
75	—	—	40	—
70	—	—	40	—
65	—	—	35	—
60	—	—	35	—
55	—	—	30	—
50	—	—	28	—
45	—	—	25	—
40	—	—	21	—
35	—	—	19	—
30	—	—	17	—
25	—	—	15	—
20	—	—	12	—

SP 6, Page 168



Longitudinal section of a double angle with bolts

- Check for rupture (tension member)



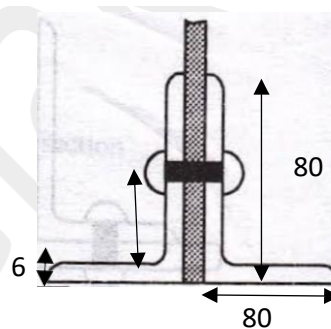
Page 33 CL 6.3.3 IS 800 - 2007

The rupture strength of an angle connected through one leg is affected by shear lag. The design strength,  $T_{dn}$ , as governed by rupture at net section is given by:

$$T_{dn} = 0.9 A_{nc} f_u / \gamma_{m1} + \beta A_{go} f_y / \gamma_{m0}$$

where

$$\beta = 1.4 - 0.076 (w/t) (f_y/f_u) (b_s/L_c) \leq (f_u \gamma_{m0} / f_y \gamma_{m1}) \geq 0.7$$



$w$  = outstand leg width or **width of unconnected leg** = 80 mm

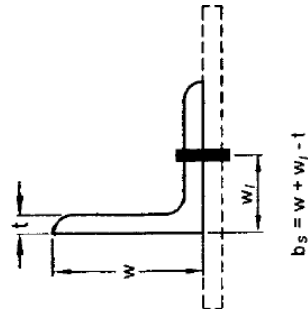
$L_c$  = Distance between the outermost bolts in the end joint measured along the load direction = 275 mm

$t$  = 6mm

$f_u$  = 410  $N/mm^2$ , Ultimate strength of material

$f_y$  = 250  $N/mm^2$ , Yield strength of material





$$b_s = w + w_l - t = 80 + 45 - 6 = 119 \text{ mm}$$

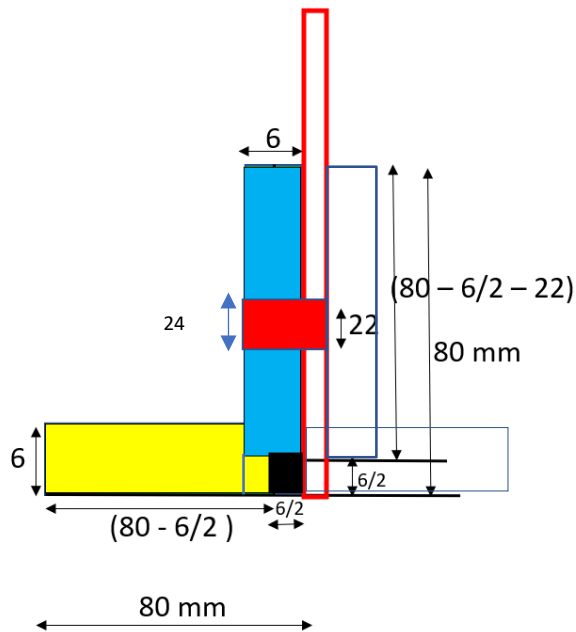
$$\beta = 1.4 - 0.076 (w/t) (f_y/f_u) (b_s/L_c)$$

$$= 1.132$$

Also find  $\frac{\gamma_{m0} f_u}{f_y \gamma_{m1}} = 1.44$

As per IS 800 - 2007,  $\beta = 1.132 \geq 0.7 \leq 1.44$

Hence take  $\beta = 1.132$



Angle section attached to gusset plate

**Gross area of outstanding or unconnected leg (without bolt)**

$$t =$$

go

$$(80 - \frac{6}{2}) 6 = 462 \text{ mm}^2$$

$$(B - \frac{t}{2}) t =$$

**Net area of connected leg (subtract area of bolt hole)**

$$(A - d_o - \frac{t}{2}) t =$$

$$(80 - 24 - \frac{6}{2}) 6 = 318 \text{ mm}^2 \text{ where diameter of bolt hole, } d_o = 22 + 2 = 24 \text{ mm}$$

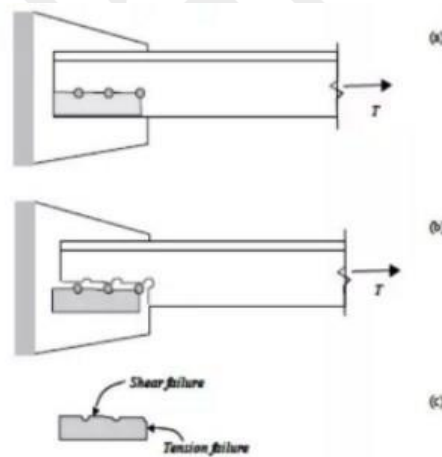
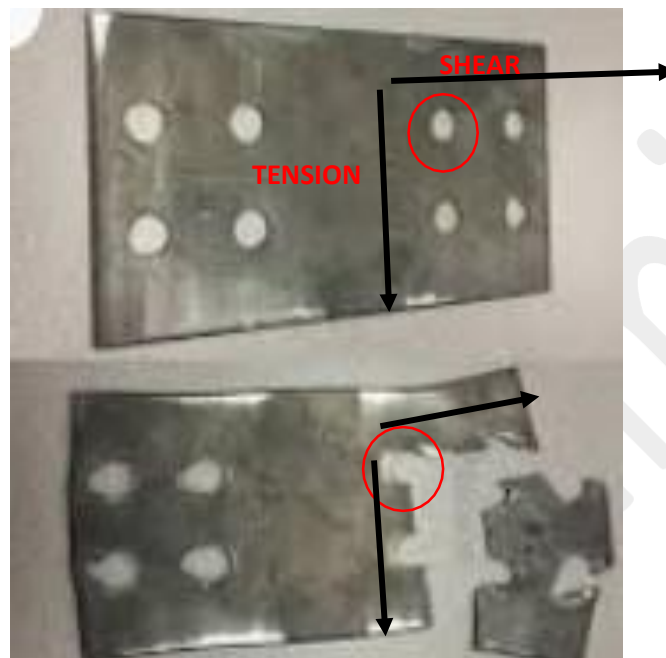
$$T_{dn} = 0.9 A_{nc} f_u / \gamma_{m1} + \beta A_{go} f_y / \gamma_{m0}$$

For double angle section multiply the value of  $T_{dn}$  by “2”

$$\text{For double angle } T_{dn} = 2 \times \left( 0.9 \times 318 \times \frac{410}{1.25} + 1.132 \times 462 \times \frac{250}{1.1} \right)$$

$$= 425.47 \text{ kN} > 312 \text{ kN}, \text{ It is safe.}$$

- Check for block shear



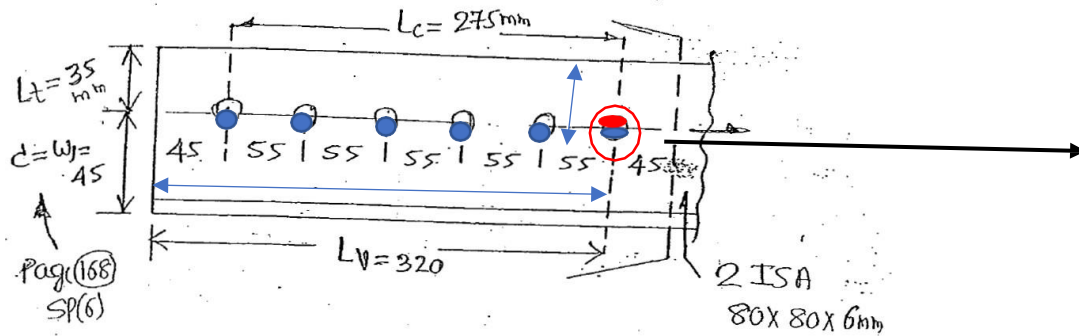
Page 33 CL 6.4.1

The block shear strength,  $T_{db}$  of connection shall be taken as the smaller of,

$$T_{db} = [ A_{vg} f_y / (\sqrt{3} \gamma_{m0}) + 0.9 A_{tn} f_u / \gamma_{m1} ]$$

or

$$T_{db} = (0.9 A_{vn} f_u / (\sqrt{3} \gamma_{m1}) + A_{tg} f_y / \gamma_{m0} )$$



Length of shearing action,  $L_v = 45 + 5 \times 55 = 320 \text{ mm}$

Length of tensile action,  $L_t = 35 \text{ mm}$

Gross area in shear parallel to force,  $A_{vg} = L_v \times t = 320 \times 6 = 1920 \text{ mm}^2$

Net area in shear parallel to force,  $A_{vn} = A_{vg} - 5.5 \times d_o \times t$

$$= 1920 - 5.5 \times 24 \times 6 = 1128 \text{ mm}^2$$

Gross area in tension perpendicular to force,  $A_{tg} = L_t \times t = 35 \times 6 = 210 \text{ mm}^2$

Net area in tension perpendicular to force,  $A_{tn} = 210 - 0.5 \times 24 \times 6 = 138 \text{ mm}^2$

$$T_{db} = [A_{vg} f_y / (\sqrt{3} \gamma_{m0}) + 0.9 A_{tn} f_u / \gamma_{m1}]$$

For two angles,  $T_{db} = 2 \times [1920 \times \frac{250}{\sqrt{3} \times 1.1} + 0.9 \times 138 \times \frac{410}{1.25}] = 585.3 \text{ kN} > 312 \text{ kN}$

$$T_{db} = (0.9 A_{vn} f_u / (\sqrt{3} \gamma_{m1}) + A_{tg} f_y / \gamma_{m0})$$

For two angles,  $T_{db} = 2 \times [0.9 \times 1128 \times \frac{410}{1.25 \times \sqrt{3}} + 210 \times \frac{250}{1.1}] = 479.9 \text{ kN} > 312 \text{ kN}$

Hence 2 ISA 80 × 80 × 6 mm is safe .

- Design of “Inner Tension members”

Member Force **CK** = 18 kN

Factored Force,  $T_{dg} = 1.5 \times 18 = 27 \text{ kN}$

Tensile strength due to gross section yielding, Page 32, CL 6.2 (IS 800)

$$T_{dg} = \frac{A_g f_y}{\gamma_{m0}}$$

$$A_g = 118.8 \text{ mm}^2 = 1.18 \text{ cm}^2$$

Use Steel table and Take single angle section ( it is a inner member)

Table 1, Page 2 , Choose ISA 50 X 50 X 6 mm ,

$$\text{Area} = 5.68 \text{ cm}^2 = 568 \text{ mm}^2$$

- Connection – Bolts - M 12 Class 5.6

Diameter of bolt hole  $d_o = 12 + 1 = 13 \text{ mm}$

- Shear strength of bolts

Assume fully threaded bolts, number of shear planes  $n_n = 1$  (single angle section) ,  $n_s = 0$  ( no shank portion)

$$A_{nb} = 0.78 \times \frac{\pi}{4} \times 12^2 = 88.2 \text{ mm}^2, A_{sb} = 0, f_{ib} = 500 \text{ N/mm}^2$$

$$V_{dsb} = \frac{V_{nsb}}{\gamma_{mb}} = \frac{f_{ub}}{\sqrt{3} \gamma_{mb}} (n_n A_{nb} + n_s A_{sb})$$

$$= \frac{500}{\sqrt{3} \times 1.25} (1 \times 88.2) = 20.36 \text{ kN} \dots (1)$$

• Bearing strength of bolts

$$V_{dpb} = 2.5 k_b d t \frac{f_u}{\gamma_{mb}}$$

Pitch,  $p = 2.5 \times d = 2.5 \times 12 = 30 \text{ mm}$ ,  $t = 6 \text{ mm}$

Edge distance  $e = 1.7 \times d_o = 1.7 \times 13 = 22.1 = 25 \text{ mm}$

where  $k_b =$  smaller of  $\frac{e}{3d_o}$ ,  $\frac{p}{3d_o} - 0.25$ ,  $\frac{f_{ub}}{f_u}$ , and 1.0

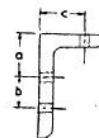
Smallest of :  $k_b = \frac{25}{3 \times 13} = 0.64$ ,  $k_b = \frac{30}{3 \times 13} - 0.25 = 0.519$ ,  $\frac{f_{ub}}{f_u} = \frac{500}{410} = 1.22$ , 1.0

$$V_{dpb} = 2.5 \times 0.519 \times 12 \times 6 \times \frac{410}{1.25} = 30.64 \text{ kN} \dots (2)$$

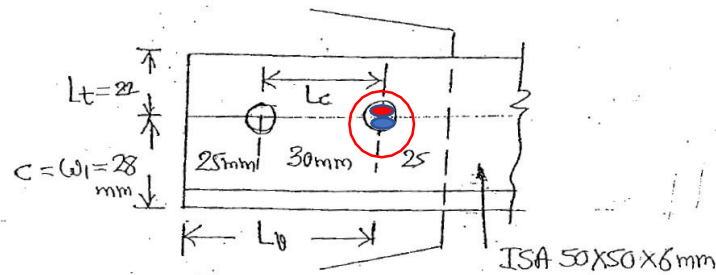
Bolt value = 20.36 kN (LEAST OF (1) and (2))

No of Bolts =  $27 / 20.36 = 2$  (Minimum no of bolts = 2)

TABLE XXXI RIVET GAUGE DISTANCES IN LEGS OF ANGLES



Leg Size	Double Row of Rivets		Single Row of Rivets	Maximum Rivet Size for Double Row of Rivets
	a	b		
mm	mm	mm	mm	mm
200	75	85	115	27
150	55	65	90	22
130	50	55	80	20
125	45	55	75	20
115	45	50	70	12
110	45	45	65	12
100	40	40	60	12
95	—	—	55	—
90	—	—	50	—
80	—	—	45	—
75	—	—	40	—
70	—	—	40	—
65	—	—	35	—
60	—	—	35	—
55	—	—	30	—
50	—	—	28	—



- Check for rupture

w, width of unconnected leg = 50 mm, t = 6 mm, distance between end bolts, L<sub>c</sub> = 30 mm,

$$b_s = w + w_t - t$$

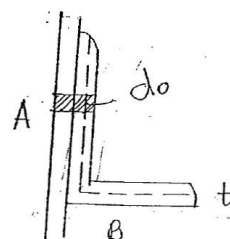
$$b_s = 50 + 28 - 6 = 72 \text{ mm}$$

The rupture strength of an angle connected through one leg is affected by shear lag. The design strength, T<sub>dn</sub>, as governed by rupture at net section is given by:

$$T_{dn} = 0.9 A_{nc} f_u / \gamma_{m1} + \beta A_{go} f_y / \gamma_{m0}$$

where

$$\beta = 1.4 - 0.076 (w/t) (f_y/f_u) (b_s/L_c) \leq (f_u \gamma_{m0} / f_y \gamma_{m1}) \geq 0.7$$



$$A_{go} = \text{Gross area of unconnected leg} = (B - \frac{t}{2}) t = (50 - \frac{6}{2}) 6 = 282 \text{ mm}^2$$

$$A_{nc} = \text{Net area of connected leg} = (A - d_o - \frac{t}{2}) t = (50 - 13 - \frac{6}{2}) 6 = 204 \text{ mm}^2$$

Also calculate  $\beta = 1.4 - 0.076 \times (50 / 6) \times (250 / 410) \times (72 / 30) = 0.47 \leq 0.7$

$\gamma_{m0} f_u = 1.44$ , The value of  $\beta = 0.47 \leq 1.44$ .

Let us take  $\beta = 0.7$ , since calculated value of  $\beta$  is less than 0.7

$$\text{For single angle } T_{dn} = 1 \times (0.9 \times 204 \times \frac{410}{1.25} + 0.7 \times 282 \times \frac{250}{1.1}) =$$

$$= 105.08 \text{ kN} > 27 \text{ kN, it is safe.}$$

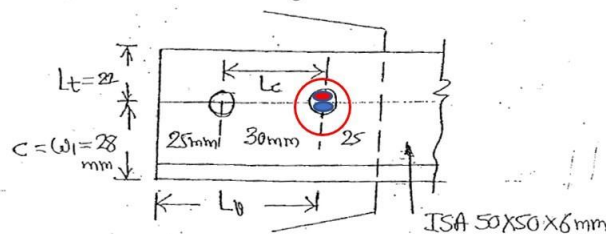
- Check for block shear

The block shear strength,  $T_{db}$  of connection shall be taken as the smaller of,

$$T_{db} = [ A_{vg} f_y / (\sqrt{3} \gamma_{m0}) + 0.9 A_{tn} f_u / \gamma_{m1} ]$$

or

$$T_{db} = (0.9 A_{vn} f_u / (\sqrt{3} \gamma_{m1}) + A_{tg} f_y / \gamma_{m0} )$$



$$A_{vg} = L_v \times t = (30 + 25) \times 6 = 330 \text{ mm}^2 \quad A_{vn} = 330 - 1.5 \times 13 \times 6 = 213 \text{ mm}^2$$

$$A_{tg} = L_t \times t = 22 \times 6 = 132 \text{ mm}^2 \quad A_{tn} = 132 - 0.5 \times 13 \times 6 = 93 \text{ mm}^2$$

$$T_{db} = [ A_{vg} f_y / (\sqrt{3} \gamma_{m0}) + 0.9 A_{tn} f_u / \gamma_{m1} ]$$

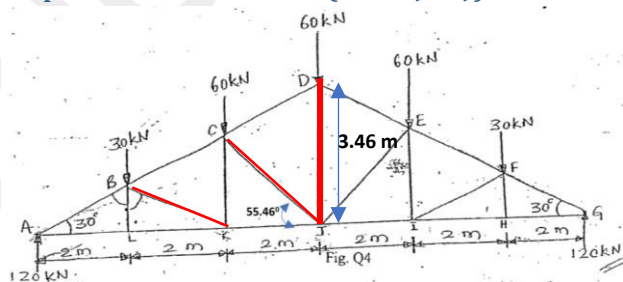
$$\text{For single angles, } T_{db} = 1 \times [ 330 \times \frac{250}{1.1 \times \sqrt{3}} + 0.9 \times 93 \times \frac{410}{1.25} ] = 70.75 \text{ kN} > 27 \text{ kN}$$

$$T_{db} = (0.9 A_{vn} f_u / (\sqrt{3} \gamma_{m1}) + A_{tg} f_y / \gamma_{m0} )$$

$$\text{For two angles, } T_{db} = 1 \times [ 0.9 \times 213 \times \frac{410}{1.25 \times \sqrt{3}} + 132 \times \frac{250}{1.1} ] = 66.30 \text{ kN} > 27 \text{ kN}$$

Hence ISA 50 × 50 × 6 mm is safe .

- Design of Inner compression members (BK, CJ, DJ)



Taking Maximum Force = 66.0 kN

Factored Force = 1.5 × 66.0 = 99. kN

Take the member which is having maximum length ie DJ = 3.46 m

Maximum Length = 3.46 m

Assume  $f_{cd} = 50 \text{ N/mm}^2$

(Assume  $f_{cd} = 40 - 120 \text{ N/mm}^2$  based on load and experience)

$$\text{Gross Area, } A_g = \frac{\text{Factored Force}}{f_{cd}} = \frac{99.07 \times 10^3}{50} = 1981.4 \text{ mm}^2 = 19.81 \text{ cm}^2$$

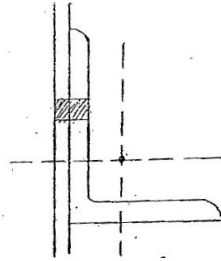
From Steel Table, Try Single ISA 100 x 100 x 10 mm (Table 1, Page 4)

$$\text{Area} = 19.03 \text{ cm}^2$$

$$r_{xx} = r_{yy} = 3.05 \text{ cm} = 30.5 \text{ mm}$$

$$r_{uu} = 3.85 \text{ cm}, r_{vv} = 1.94 \text{ cm}$$

$$\text{So } r_{\min} = 1.94 \text{ cm} = 19.4 \text{ mm}$$



Here load is acting through only one leg it will be subjected to torsional buckling  
Using Page 48, Table 12 and using CL 7.5.1.2 Loaded through one leg ( IS 800 2007)

### 7.5.1.2 Loaded through one leg

The flexural torsional buckling strength of single angle loaded in compression through one of its legs may be evaluated using the equivalent slenderness ratio,  $\lambda_e$ , as given below:

$$\lambda_e = \sqrt{k_1 + k_2 \lambda_{vv}^2 + k_3 \lambda_{\phi}^2}$$

where

$k_1, k_2, k_3$  = constants depending upon the end condition, as given in Table 12,

$$\lambda_{vv} = \frac{\left(\frac{l}{r_{vv}}\right)}{\epsilon \sqrt{\frac{\pi^2 \epsilon}{250}}} \quad \text{and} \quad \lambda_{\phi} = \frac{(b_1 + b_2) / 2t}{\epsilon \sqrt{\frac{\pi^2 \epsilon}{250}}}$$

where

$l$  = centre-to-centre length of the supporting member,

$r_{vv}$  = radius of gyration about the minor axis,

$b_1, b_2$  = width of the two legs of the angle,

$t$  = thickness of the leg, and

$\epsilon$  = yield stress ratio  $(250/f_y)^{0.5}$ .



**Table 12 Constants  $k_1$ ,  $k_2$  and  $k_3$**

Sl No.	No. of Bolts at Each End Connection	Gusset/Connecting Member Fixity <sup>1)</sup>	$k_1$	$k_2$	$k_3$
(1)	(2)	(3)	(4)	(5)	(6)
i)	$\geq 2$	Fixed	0.20	0.35	20
		Hinged	0.70	0.60	5
ii)	1	Fixed	0.75	0.35	20
		Hinged	1.25	0.50	60

<sup>1)</sup> Stiffness of in-plane rotational restraint provided by the gusset/connecting member.

For partial restraint, the  $\lambda_c$  can be interpolated between the  $\lambda_c$  results for fixed and hinged cases.

Assuming bolts  $\geq 2$  and hinged end conditions with gusset plate,  $k_1 = 0.7$ ,  $k_2 = 0.6$ ,  $k_3 = 5$

Assuming Effective length,  $L_{eff} = 0.85 \times L = 0.85 \times 3460 = 2940 \text{ mm}$

$\varepsilon = 1$ ,  $E = 2 \times 10^5 \text{ N/mm}^2$   $b_1 = b_2 = 100 \text{ mm}$ ,  $t = 10 \text{ mm}$

$$\lambda_{vv} = \frac{\left(\frac{I}{r_{vv}}\right)}{\varepsilon \sqrt{\frac{\pi^2 E}{250}}}$$

$$\lambda_{vv} = \frac{L_{eff}}{r_{vv}} = 1.705$$

$$\lambda_{\phi} = \frac{(b_1 + b_2)/2t}{\varepsilon \sqrt{\frac{\pi^2 E}{250}}}$$

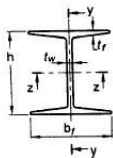
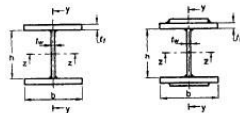

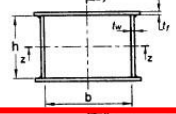
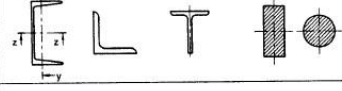
$$\lambda_{\phi} = 0.1125, k_1 = 0.7, k_2 = 0.6, k_3 = 5$$

$$\lambda_c = \sqrt{k_1 + k_2 \lambda_{vv}^2 + k_3 \lambda_{\phi}^2}$$

Equivalent slenderness ratio,  $\lambda_c = 1.589$

From Table 10, Page No 44, Choose the buckling class based on type of section

**Table 10 Buckling Class of Cross-Sections**  
(Clause 7.1.2.2)

Cross-Section (1)	Limits (2)	Buckling About Axis (3)	Buckling Class (4)
	$h/b_f > 1.2$ ; $t_f \leq 40$ mm	z-z y-y	a b
	$40 \leq \text{mm} < t_f \leq 100$ mm	z-z y-y	b c
	$h/b_f \leq 1.2$ ; $t_f \leq 100$ mm $t_f > 100$ mm	z-z y-y z-z y-y	b c d d
	$t_f \leq 40$ mm $t_f > 40$ mm	z-z y-y z-z y-y	b c c d
	Hot rolled Cold formed	Any Any	a b
	Generally (except as below) Thick welds and $b/t_f < 30$ $h/t_w < 30$	Any z-z y-y	b c c
		Any	c

Since it is single angle section, choose buckling class as 'c'. Based on Buckling class, find  $\alpha$  from Table 7 Page No 35 as 0.49.

**Table 7 Imperfection Factor,  $\alpha$**   
(Clauses 7.1.1 and 7.1.2.1)

Buckling Class	a	b	c	d
$\alpha$	0.21	0.34	0.49	0.76

From Page 34, CL 7.1.2.1, IS 800-2007

$$\phi = 0.5 [1 + \alpha (\lambda - 0.2) + \lambda^2]$$

Put  $\alpha = 0.49$ ,  $\lambda_c = 1.589$ ,  $\Phi = 2.102$

$$f_{cd} = \frac{f_y / \gamma_{m0}}{\phi + [\phi^2 - \lambda^2]^{0.5}}$$

(Single angle section) Design compressive stress,  $f_{cd} = \frac{250}{1.1(2.102 + (2.102^2 - 1.589^2)^{0.5})} = 108.47 \text{ N/mm}^2$

The design compressive strength of a member is given by:

Load  $P_d = A \times f_{cd} = 1903 \times 108.47 = 206.42 \text{ kN} > 99.07 \text{ kN}$

Hence it is safe.

Design of connection using M22, class 5.6 ( same diameter bolt)

Benefit- No need to do shear strength calculations, (for single angle,  $n_n = 1$ )

- Shear strength of bolts

Assume fully threaded bolts, number of shear planes  $n_n = 1$  ( Single angle section),  $n_s = 0$  (no shank portion)

$$A_{ns} = 0.78 \times \frac{\pi}{4} \times 22^2 = 296.5 \text{ mm}^2, A_{sb} = 0$$

$$V_{dsb} = \frac{V_{nsb}}{\gamma_{mb}} = \frac{f_{ub}}{\sqrt{3} \gamma_{mb}} (n_n A_{nb} + n_s A_{sb})$$
$$= \frac{500}{\sqrt{3} \times 1.25} (1 \times 296.5) = 68.47 \text{ kN} \dots (1)$$

- Bearing strength of bolts

$$V_{dpb} = 2.5 k_b d t \frac{f_u}{\gamma_{mb}}$$

Pitch,  $p = 2.5 \times d = 2.5 \times 22 = 55 \text{ mm}$

Edge distance  $e = 1.7 \times d_o = 1.7 \times 24 = 40.8 = 45 \text{ mm}$

where  $k_b =$  smaller of  $\frac{e}{3d_o}$ ,  $\frac{p}{3d_o} - 0.25$ ,  $\frac{f_{ub}}{f_u}$ , and 1.0

$$k_b = \frac{45}{3 \times 24} = 0.63, k_b = \frac{55}{3 \times 24} - 0.25 = 0.513, \frac{f_{ub}}{f_u} = \frac{500}{410} = 1.22, 1.0$$

$$V_{dpb} = 2.5 \times 0.513 \times 22 \times 10 \times \frac{410}{1.25} = 92.54 \text{ kN} \dots (2)$$

Bolt value = Minimum of (1) and (2) = 68.47kN

$$\text{No of bolts} = \frac{99}{68.47} = 2$$

- Design of Bearing Plate

Support Reaction = 120 kN

Factored Reaction =  $1.5 \times 120 = 180 \text{ kN}$

CL 34.4, Page 56 IS 456 2000

Using M 20 concrete, Bearing pressure of concrete =  $0.45 f_{ck} = 0.45 \times 20 = 9 \text{ N/mm}^2$

**IS 456 : 2000**

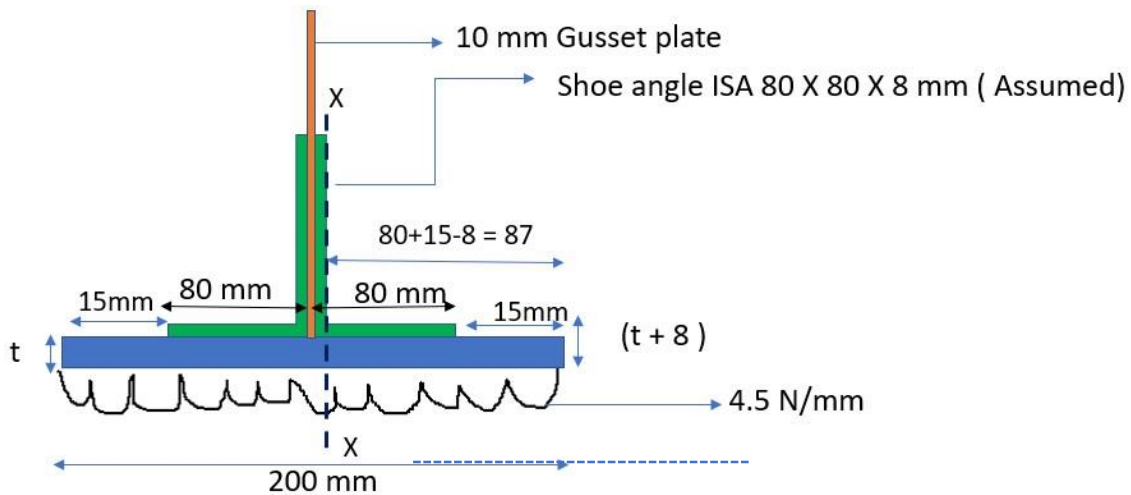
**For working stress method of design the permissible bearing stress on full area of concrete shall be taken as  $0.25 f_{ck}$ ; for limit state method of design the permissible bearing stress shall be  $0.45 f_{ck}$ .**

$$\text{Area of Bearing plate} = \frac{180 \times 10^3}{9} = 20000 \text{ mm}^2$$

$$\text{Provide square plate} = \sqrt{20000} = 141.4 \text{ mm}$$

Provide 200 mm  $\times$  200 mm as base dimensions

- Calculation of thickness 't' of the bearing plate



Upward pressure,  $q_o = \text{Support Reaction} / \text{Size of plate} = 180 \times 10^3 / (200 \times 200) = 4.5 \text{ N/mm}^2$

For 1mm, it is 4.5 N/mm

Moment at section XX (near the inner face of angle section),  $M_{xx} = 4.5 \times 87 \times \frac{87}{2} =$

17030.25 Nmm

Take  $b = 1 \text{ mm}$ ,  $d = (t + 8)$

Using Bending formula  $M_{xx} = \sigma_b \times Z = \frac{f_y}{\gamma_{mo}} \times b \times \frac{d^2}{6}$  (Z is section modulus)

$$17.03 \times 10^3 = \frac{250}{1.1} \times 1 \times \frac{(t+8)^2}{6}$$

$$t = 13.20 \text{ mm} \approx 14 \text{ mm}$$

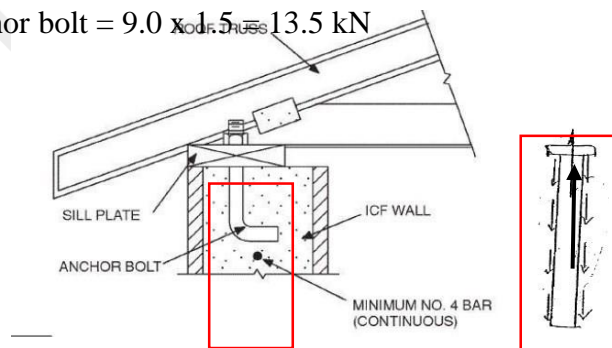
Bearing plate  $200 \times 200 \times 14 \text{ mm}$

### • Design of Anchor Bolts

Given Uplift force = 18 kN at each support

Assuming two Anchor bolts at each support, Force on each anchor bolt =  $18/2 = 9.0 \text{ kN}$

Factored force on anchor bolt =  $9.0 \times 1.5 = 13.5 \text{ kN}$



From CL 26.2.1.1 Bond stress for M 20 ( RCC column M 20) = **1.2** N/mm<sup>2</sup>, increase by **60%**  
 (Page 43, IS 456 2000) = 1.2 x 1.6

IS 456 : 2000

**26.2.1.1** Design bond stress in limit state method for plain bars in tension shall be as below:

Grade of concrete	M 20	M 25	M 30	M 35	M 40 and above
Design bond stress, $\tau_{bd}$ , N/mm <sup>2</sup>	1.2	1.4	1.5	1.7	1.9

For deformed bars conforming to IS 1786 these values shall be increased by 60 percent.

For bars in compression, the values of bond stress for bars in tension shall be increased by 25 percent.

The values of bond stress in working stress design, are given in B-2.1.

2) In the compression zone, from the mid depth of the beam.

b) *Stirrups*—Notwithstanding any of the provisions of this standard, in case of secondary reinforcement, such as stirrups and transverse ties, complete development lengths and anchorage shall be deemed to have been

To find length of Anchor bolt

Let us assume 18 mm diameter Anchor Bolt

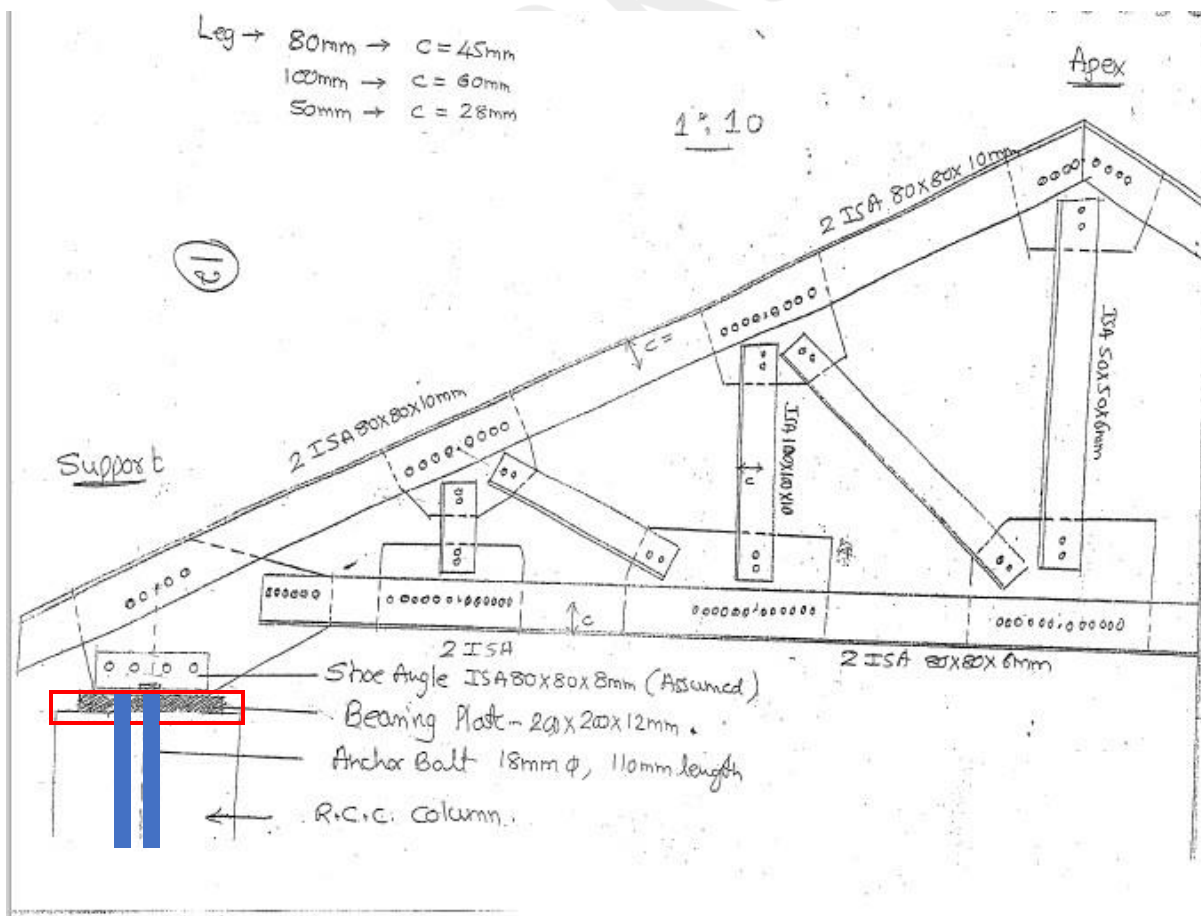
Factored Uplift Force = Force developed around the anchor bolt

$$= (\text{Circumference} \times \text{length}) \times \text{Bond stress}$$

$$13.5 \times 10^3 = \pi \times 18 \times \text{length} \times (1.2 \times 1.6)$$

$$\text{length} = 124.34 \text{ mm} \approx 125 \text{ mm}$$

Provide 2, 18 mm  $\Phi$  anchor bolts of length 125 mm at each end.



- OR**
- 4 Design a simply supported gantry girder to carry an electrically operated travelling crane with the following data:  
 Span of crane bridge = 25m  
 Column spacing = span of gantry girder = 8m  
 Wheel Base = 3.5m  
 Crane capacity = 200kN  
 Weight of crane bridge = 150kN  
 Weight of Trolley = 75kN  
 Min Hook Distance = 1.0m  
 Weight of Rail = 0.30kN/m  
 Height of Rail = 105mm  
 Also draw sectional elevation.
- (50 Marks)

\*\*\*\*\*

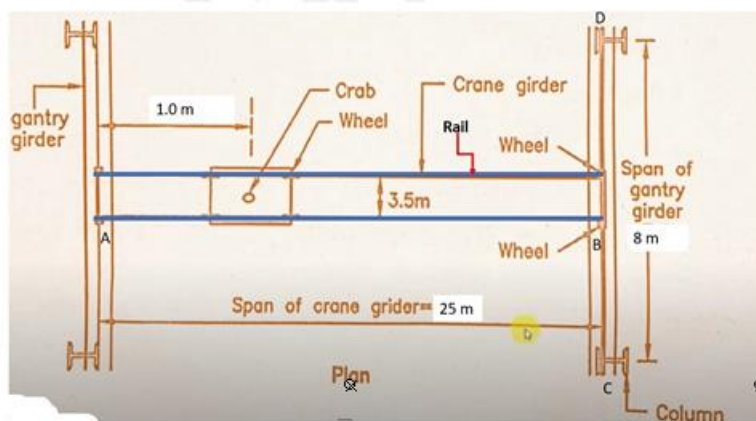
### DESIGN OF GANTRY GIRDER

**1. Design a gantry girder to be used in an industrial building carrying an electrically operated overhead travelling crane, for the following data**

- Centre to Centre between distance between gantry rails or span of crane girders **25 m**
- Centre to Centre distance between columns (span of gantry girder) **8 m**
- Crane capacity **200 kN**
- Self-weight of the crane girder excluding trolley **150 kN** at centre or  $(150/25 = 6\text{kN/m})$
- Self-weight of the crab or trolley, electric motor, hook, etc. **75 kN**
- Approximate minimum approach of the crane hook to the gantry girder **1.0 m**
- Wheel base **3.5 m**
- Self-weight of rail section **300 N/m = 0.3kN/m**
- Diameter of crane wheels **150 mm**
- Height of rail **105 mm**
- Steel is of grade **Fe 410**. Design also the field welded connection if required.
- The support bracket connection need not be designed.

*Solution* For Fe 410 grade of steel:  $f_u = 410 \text{ MPa}$ ,  $f_y = f_{yw} = f_{yf} = 250 \text{ MPa}$

Design Maximum load transferred from crane girder to gantry girder

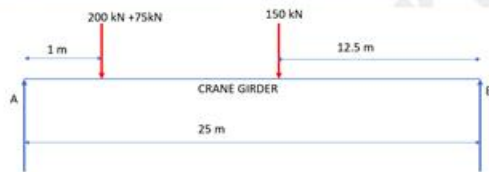
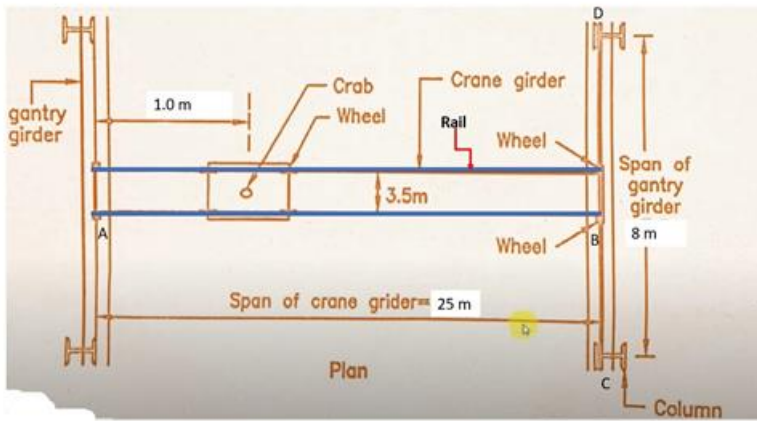


The maximum reaction/load in crane girder occurs when the crab or trolley along with hook if it istowards left or

right of crane girder with a minimum hook distance of 1.0 m.

The Free body diagram is shown as below.





To find max Reaction  $R_A$ , take moment at B,  $\sum M_B = R_A \times 25 - 150 \times 12.5 - 275 \times 24 = 0$

$R_A = 339 \text{ kN}$  is the reaction developed at the end of Crane girder

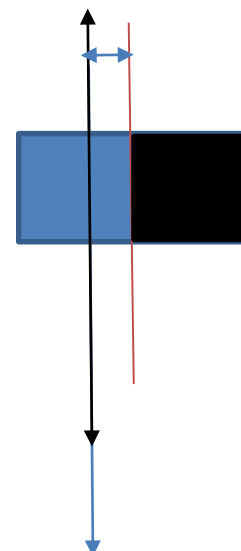
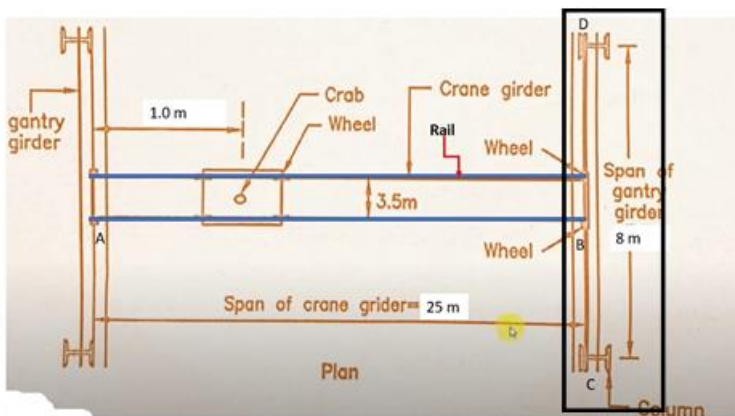
Since there are two wheels at each end of crane bridge, **load on each wheel of trolley or crab** =  $339/2 = 169.5 \text{ kN}$

As per IS: 875 (part 2)-1987, for Electrically operated crane (EOT crane), ( CL 6.1, Page 15)

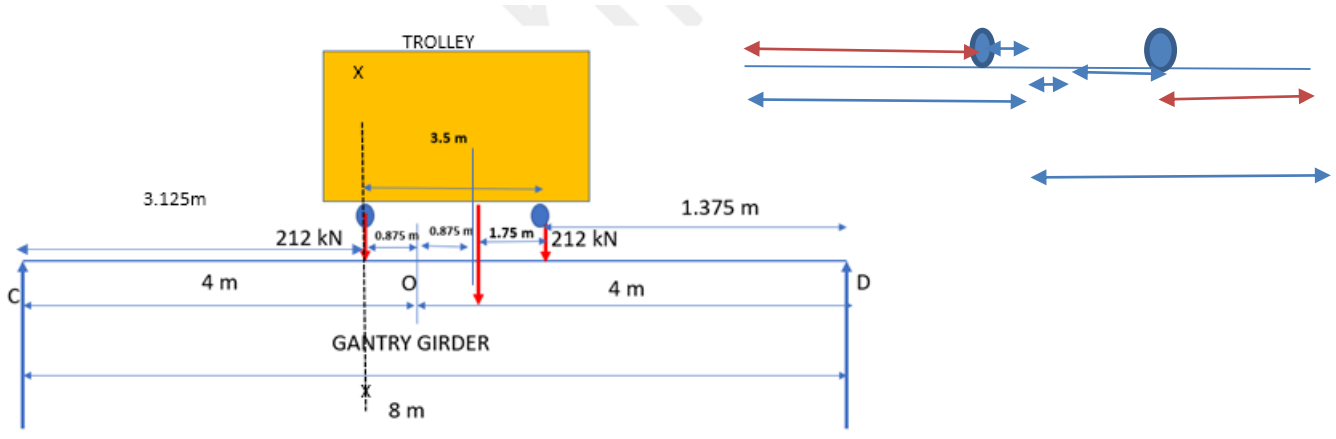
Vertical forces transferred through rails = 25% of **maximum static** wheel load ( 25% means 1.25)

Load on each wheel =  $1.25 \times 169.5 = 211.87 \text{ kN} \approx 212 \text{ kN}$

- Design Maximum bending moments on gantry girder







The arrangement of wheel loads for Maximum bending moment is at **Centre of gravity of the wheelloads(trolley)** and one of the wheel load are **equidistant from the Centre of the gantry girder**

$$\sum M_D = R_c \times 8 - 212 \times 1.375 - 212 \times (1.375 + 1.75 + 1.75) = 0, R_c = 165.63 \text{ kN}$$

$$B M \text{ under a wheel load or Maxi B M at XX} = R_c \times 3.125 = 165.63 \times 3.125 = 517.6 \text{ kNm}$$

$$\text{Factored B M} = 517.6 \times 1.5 = 776.4 \text{ kNm}$$

B M and S F due to self-weight or dead load

Assume self-weight of gantry girder as 1.6kN/m

Self-weight of rail = 300N/m = 0.3kN/m ( given )

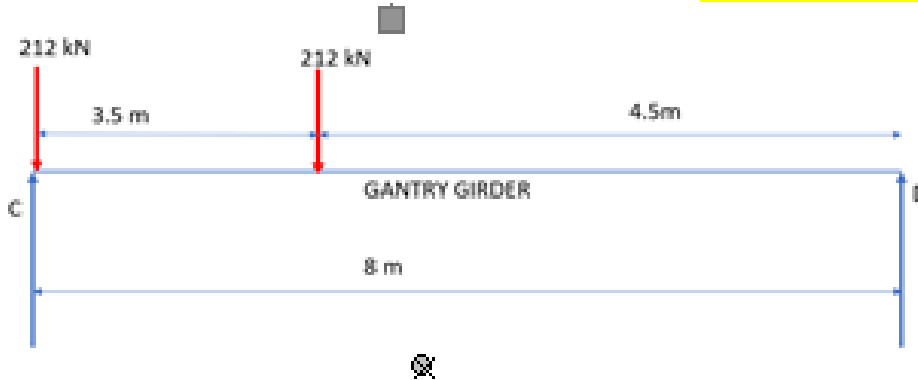
Total self-weight = 1.6 + 0.3 = 1.9 kN/m, factored self-weight = 1.5 x 1.9 = 2.85 kN/m

$$B M \text{ due to self-weight or dead load} = 2.85 \times 8^2/8 = 22.8 \text{ kNm}$$

$$S F \text{ due to self-weight or dead load} = 2.85 \times 8/2 = 11.4 \text{ kN}$$

- **Design Maximum shear force**

The shear due to the wheel load is maximum when **one of the wheels is at the support**



$$\sum M_D = R_c \times 8 - 212 \times 8 - 212 \times 4.5 = 0,$$

$$R_c = 331.25$$

$$\text{Factored S F} = 331.25 \times 1.5 = 496.88 \text{ kN}$$

- **Lateral load and its moment**

Lateral load is developed due to the application of brakes or sudden acceleration of the trolley

As per IS: 875 (part 2)-1987, Lateral load or Horizontal forces transverse to the rails = 10% of the weight of the crab or trolley and the weight lifted on the crane(crane capacity)

$$\text{Lateral force or Horizontal force} = 10 / 100 \times (200 + 75) = 27.5 \text{ kN}$$

$$\text{Lateral load acting on each wheel (there are 4 wheels)} = 27.5 / 4 = 6.875 \approx 7 \text{ kN}$$

$$\text{Factored lateral load} = 7 \times 1.5 = 10.5 \text{ kN}$$

Bending Moment due to lateral load

$$\text{For 212 kN, B.M} = 776.4 \text{ kNm}$$

Then for 10.5 kN, B.M due to lateral load = ?

$$212/10.5 = 776.4 / ? , ? =$$

$$\frac{212}{10.5} = \frac{776.4}{?}$$

Bending moment due to lateral load? = 38.46kNm

Total Design bending moment (Live Load + Lateral load + Dead Load) = 776.4 + 38.46 + 22.8 = 837.76 kNm

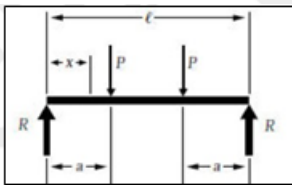
Total Design shear force (Live Load + Lateral load + Dead Load) = 496.88 + 11.4 + 7 = kN

- preliminary trial section

The trial section is selected based on deflection criteria, IS 800 2007, Table 6 Page 31

Table 6 Deflection Limits

Type of Building	Deflection	Design Load	Member	Supporting	Maximum Deflection
(1)	(2)	(3)	(4)	(5)	(6)
Industrial Buildings	Vertical	Live load/ Wind load	Purlins and Girts	Elastic cladding	Span/150
				Brittle cladding	Span/180
		Live load	Simple span	Elastic cladding	Span/240
				Brittle cladding	Span/300
		Live load	Cantilever span	Elastic cladding	Span/120
				Brittle cladding	Span/150
		Live load/ Wind load	Rafter supporting	Profiled Metal Sheeting	Span/180
				Plastered Sheeting	Span/240
		Crane load (Manual operation)	Gantry	Crane	Span/500
					Span/750
Crane load (Electric operation up to 50 t)	Gantry	Crane	Span/1000		
			Span/1000		
No cranes	Column	Elastic cladding	Height/150		
		Masonry/Brittle cladding	Height/240		
Crane + wind	Gantry (lateral)	Crane (absolute)	Span/400		
			Relative displacement between rails supporting crane	10 mm	



$$\Delta_{max} \text{ (at center)} = \frac{Pa}{24EI} (3l^2 - 4a^2)$$



Maxi deflection for electrically operated crane,  $\delta_{max deflection} = \text{Span of gantry girder}/750$   
 $= 8000/750 = 10.67 \text{ mm}$

But Maxi deflection = Deflection due to Dead load + Deflection due to Live load

Let us assume deflection due to dead load  $\delta_{dead load}$  as 1 mm (Since it is steel structure self-weight is very less, we are assuming very less deflection also)

Deflection due to live load  $\delta_{live load}$  under two equal concentrated loads (two equal wheel loads)

Taking  $\Delta_{max} = \delta_{live load}$

$$\delta_{live load} = \frac{Pa}{24EI} (3l^2 - 4a^2)$$

Put  $P = 212 \text{ kN}$ ,  $E = 2 \times 10^5 \text{ N/mm}^2$ ,  $a = 2.25 \text{ m}$ ,  $l = 8 \text{ m}$

$$\delta_{max deflection} = \delta_{dead load} + \delta_{live load}$$

$$10.67 = 1.00 + \frac{Pa}{24EI} (3l^2 - 4a^2)$$

We need to calculate Moment of Inertia  $I$  for the selecting the section for gantry girder

$$10.67 = 1.00 + \frac{212\,000 \times 2250}{24 \times 2 \times 10^5 \times I} (3 \times 8000^2 - 4 \times 2250^2)$$

$$I = 1765 \times 10^6 \text{ mm}^4$$

Increase the value of  $I$  by 30% =  $1.3 \times 1765 \times 10^6 = 2294.5 \times 10^6 \text{ mm}^4 = 229450 \text{ cm}^4$

Take steel table and select a built-up section from Page 42 and Table 12 based on  $I_{xx}$  value

Selecting ISWB 500 @ 95.2kg/m, ISMC 400 @ 49.4kg/m as section for Gantry girder (Table 12, Page 42) for  $I_{xx} = 230194 \text{ cm}^4 = 2301.9 \times 10^6 \text{ mm}^4$

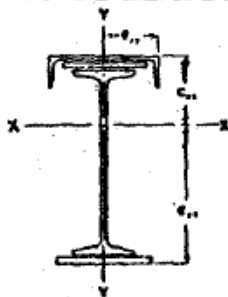


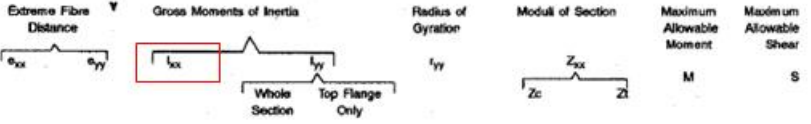
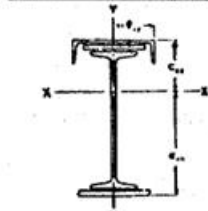
TABLE 12 (Contd.)

**SINGLE JOIST WITH CHANNEL AND PLATES ON THE FLANGES (GIRDERS)**

Joist Designation	w	Composed of		Weight per Metre (W)		Sectional Area a	Centre of Gravity C <sub>xx</sub>	Mean Thickness of Flanges			
		Channel Designation	Plate Width x Thickness	Plate Width x Thickness	kg			N	Top	Bottom	
											Kg N
ISWB 500	95.2 933.9	ISMC 400	49.4 484.6	320 x 10.0	320 x 20.0	219.9	2157.2	280.15	24.40	25.8	31.5
				12.0	25.0	237.5	2329.9	302.55	25.63	27.4	36.5
				16.0	32.0	265.1	2600.6	337.75	26.95	30.6	43.5
				20.0	40.0	295.3	2896.9	376.15	28.37	33.8	51.5
ISMC 350	42.1 413.0	250 x 10.0	320 x 20.0	12.0	25.0	207.1	2031.7	263.88	25.74	25.7	31.5
				16.0	32.0	223.6	2193.5	284.88	27.06	27.2	36.5
				20.0	40.0	249.1	2443.7	317.28	28.52	30.0	43.5
				277.0	2717.4	352.88	30.06	32.9	51.5		
ISMC 400	49.4 484.6	—	320 x 10.0	12.0	169.7	1664.8	216.15	22.81	17.8	21.5	
				16.0	174.7	1713.8	222.55	23.65	23.5		
				184.7	1811.9	235.35	25.21	27.5			
ISMC 350	42.1 413.0	—	320 x 10.0	12.0	162.4	1593.1	206.88	23.69	21.5	21.5	
				16.0	167.4	1642.2	213.28	24.54	18.6	23.5	
				177.5	1741.3	226.08	26.11	27.5	27.5		
ISMB 450	72.4 710.2	ISMC 300	35.8 351.2	250 x 10.0	127.9	1254.7	162.91	20.93	16.3	20.4	
				12.0	131.8	1293.0	167.91	21.71	22.4		
				16.0	139.7	1370.5	177.91	23.14	26.4		
ISMC 250	30.4 296.2	—	250 x 10.0	12.0	122.4	1200.7	155.94	21.71	17.5	20.4	
				16.0	126.3	1239.0	160.94	22.49	22.4		
				134.2	1316.5	170.94	23.93	26.4			
ISMC 225	25.9 254.1	—	200 x 10.0	12.0	114.0	1118.3	145.28	21.57	18.0	23.0	
				16.0	117.2	1149.7	149.28	22.25	25.0		
				123.5	1211.5	157.28	23.51	29.0			

TABLE 12 (Contd.)

**SINGLE JOIST WITH CHANNEL AND PLATES ON THE FLANGES (GIRDERS)**



cm	cm	cm <sup>4</sup>	cm <sup>4</sup>	cm <sup>4</sup>	cm	cm <sup>3</sup>	cm <sup>3</sup>	kg.m x 10 <sup>3</sup>	kg x 10 <sup>3</sup>
29.46	20.00	152781.2	26262.6	19307.2	9.68	6262.3	5185.5	81.7	46.8
28.93		170717.9	28174.1	19853.4	9.85	6661.7	5900.4	92.9	
28.21		195797.6	31177.8	20945.7	9.61	7377.1	6923.9	109.1	
28.49		230184.6	34454.6	22038.1	9.57	8114.6	8079.2	127.2	
28.07	17.50	143834.0	19759.2	12803.9	8.65	5587.0	5124.9	80.7	
27.45		159945.7	21385.0	13064.4	8.66	5911.3	5826.3	91.8	
27.09		184963.8	23917.3	13585.3	8.66	6485.6	6827.6	107.5	
26.75		212768.9	26522.6	14106.2	8.67	7077.7	7954.5	125.3	
29.05	20.00	106172.6	20801.3	16576.5	9.81	4654.5	3654.9	57.6	
28.41		111454.1	21347.4	16576.5	9.79	4712.8	3922.9	61.8	
27.25		121362.3	22439.7	16576.6	9.76	4815.0	4452.9	70.1	
28.12	17.50	101911.5	15726.5	11501.7	8.72	4301.4	3624.5	57.1	
27.47		106854.4	16272.6	11501.8	8.73	4354.4	3889.8	61.3	
26.30		116100.2	17364.9	11501.9	8.76	4447.3	4413.8	69.5	
25.83	15.00	62983.5	8496.7	6779.4	7.22	3008.6	2438.8	38.4	40.0
25.25		66244.2	8759.1	6779.5	7.22	3051.8	2623.1	41.3	
24.22		72359.1	9279.9	6779.6	7.22	3127.4	2987.2	47.0	
25.00	12.50	60394.7	5952.9	4233.7	6.18	2781.6	2416.0	38.1	
24.42		63446.4	6213.3	4233.8	6.21	2820.9	2596.3	40.9	
23.38		69152.5	6734.1	4233.8	6.28	2889.5	2958.0	46.6	
25.07	11.25	55138.5	4195.3	3111.5	5.37	2556.1	2199.5	34.6	
24.59		57604.5	4328.6	3111.5	5.38	2589.5	2342.2	36.9	
23.73		62272.6	4595.3	3111.6	5.41	2649.1	2623.8	41.3	

Selecting ISWB 500 @ 95.2kg/m, ISMC 400 @ 49.4kg/m as section for Gantry girder( Table 12,Page 43)

$A = 376.15 \text{ cm}^2 = 37615 \text{ mm}^2$ ,  $r_{yy} = 95.7 \text{ mm}$ ,  $C_{xx} = 283.7 \text{ mm}$  ( from Top),  $e_{xx} = 284.9 \text{ mm}$  ( FromBottom)

### Sectional properties of I section and Channel section used in Gantry Girder

ISWB 500 @ 95.2kg/m (Table 4, Page 14 Steel Table)	ISMC 400 @ 49.4kg/m (Table 5, Page 16, Steel Table)
$A = 121.22 \text{ cm}^2$	$A = 62.93 \text{ cm}^2 = 6293 \text{ mm}^2$
$h = 500 \text{ mm}$	$h = 400 \text{ mm}$
$b = 250 \text{ mm}$	$b = 100 \text{ mm}$
$t_f = 14.7 \text{ mm}$	$t_f = 15.3 \text{ mm}$
$t_w = 9.9 \text{ mm}$	$t_w = 8.6 \text{ mm}$
$r_{xx} = 207.7 \text{ mm}, r_{yy} = 49.6 \text{ mm}$	$C_{yy} = 24.2 \text{ mm}$

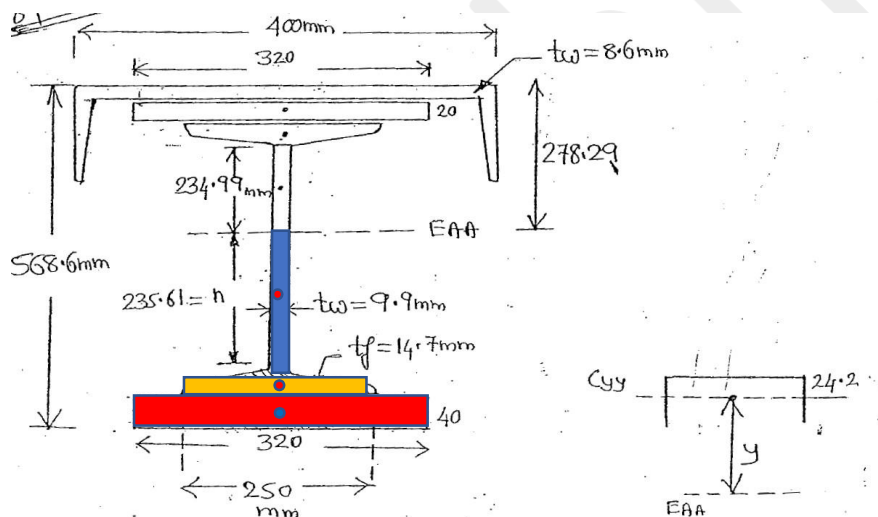
#### Location of Equal area axis

**Equal area axis** is the location of the axis which results in equal compressive and tensile forces when all fibres in a section have reached yield stress

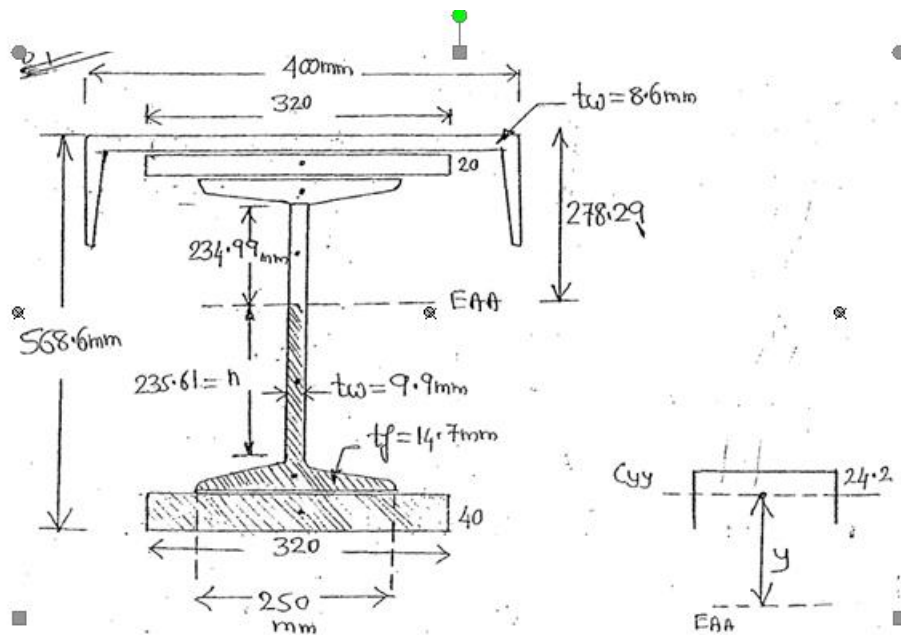
To locate Equal area axis 'n', Area of shaded portion =  $\frac{1}{2}$  total area of section

$$9.9 \times n + 250 \times 14.7 + 320 \times 40 = \frac{1}{2} \times (37615)$$

$$n = 235.61 \text{ mm}$$

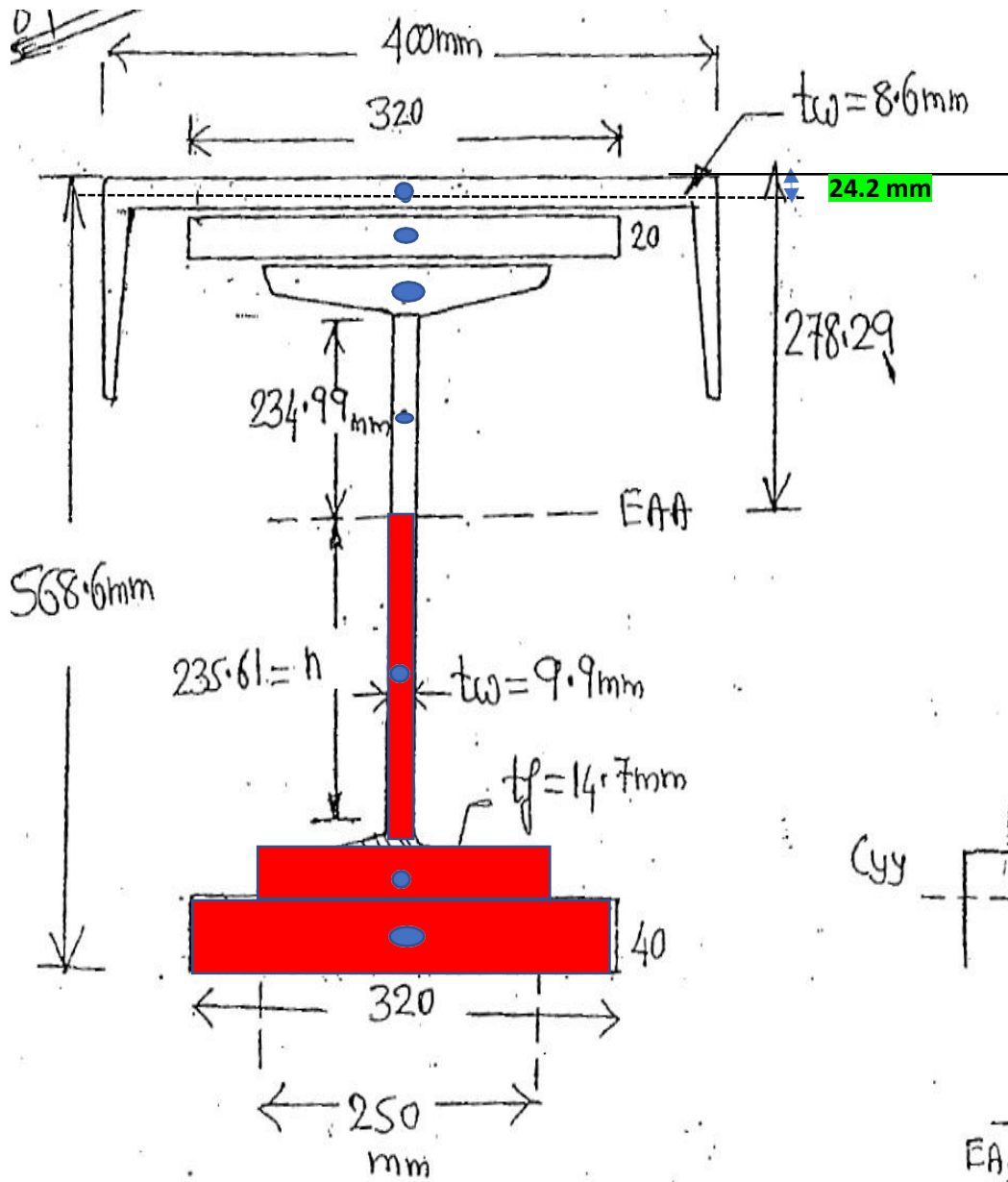


- **Plastic modulus Z<sub>p</sub>** – It is sum of areas of compression and tension zones multiplied by corresponding distance of the centroid of the compressive and tension area from the equal area axis



**Calculation of Plastic modulus  $Z_p = \sum a \times \bar{y}$**

The distances  $\bar{y}$  are measured from EAA axis to centroidal axis of the section



Shaded area (□)	Centroidal distance from EAA (mm)	$Z_p$ (mm <sup>3</sup> ) (□ × mm)	Unshaded area (□) mm <sup>2</sup>	Centroidal distance from EAA (mm)	$Z_p$ (mm <sup>3</sup> ) (□ × mm)
 320 × 40	mm + mm + mm - mm		 mm × mm	$\frac{\text{mm}}{\text{mm}}$	
 mm × mm	mm + mm mm		 250 × mm	mm + 14.7/2	
 mm × mm	$\frac{\text{mm}}{\text{mm}}$		320 × mm	mm + mm + mm	
		$= \sum \square \times$ mm			



			Area of Channel section (From steel table)	□□□. □□ - □□. □	
<b>Total</b>	$= \sum \square \times \bar{\square}$	=		+	$= \sum \square \times \bar{\square}$
<b>Plastic Modulus</b> $Z_p = \sum \square \times \bar{\square}$	$= \square. \square \square \times \square \square \square \square \square$				

- Check for Moment of resistance

Page 54, CL 8.2.2 IS 800 2007

### 8.2.2 Laterally Unsupported Beams

Resistance to lateral torsional buckling need not be checked separately (member may be treated as laterally supported, *see 8.2.1*) in the following cases:

- Bending is about the minor axis of the section,
- Section is hollow (rectangular/ tubular) or solid bars, and
- In case of major axis bending,  $\lambda_{LT}$  (as defined herein) is less than 0.4.

The design bending strength of laterally unsupported beam as governed by lateral torsional buckling is given by:

$$M_d = \beta_b Z_p f_{bd}$$

where

$\beta_b = 1.0$  for plastic and compact sections.

$\beta_b = Z_e/Z_p$  for semi-compact sections.

$Z_p, Z_e =$  plastic section modulus and elastic section modulus with respect to extreme compression fibre.

$f_{bd} =$  design bending compressive stress, obtained as given below [*see Tables 13(a) and 13(b)*]

$\alpha_{LT}$ , the imperfection parameter is given by:

$\alpha_{LT} = 0.21$  for rolled steel section

$\alpha_{LT} = 0.49$  for welded steel section

We need to calculate  $f_{bd}$  based on the values of  $f_{cr,b}$  and imperfection factor  $\alpha_{LT}$

$\alpha_{LT}$ , the imperfection parameter is given by:

$$\alpha_{LT} = 0.21 \text{ for rolled steel section}$$

$$\alpha_{LT} = 0.49 \text{ for welded steel section}$$

$$f_{cr,b} = \frac{1.1 \pi^2 E}{(L_{LT}/r_y)^2} \left[ 1 + \frac{1}{20} \left( \frac{L_{LT}/r_y}{h_f/t_f} \right)^2 \right]^{0.5}$$

$E = 2 \times 10^5 \text{ N/mm}^2$ ,  $L_{LT} = 8000 \text{ mm}$  (span of gantry girder), [ $r_{yy} = 95.7 \text{ mm}$ ,  $t_f = 33.8 \text{ mm}$  (Top flange mean thickness) (for Girder from Steel table, Table 12, Page 43)]

$h_f =$  centre to centre distance between the flanges = Overall depth of girder  $- \frac{1}{2}$  (Top and bottom mean flange thickness of girder)

$$= 568.6 - \frac{1}{2} \times (33.8 + 51.5)$$

$$= 525.9 \text{ mm}$$

$$f_{cr,b} = 485.65 \text{ N/mm}^2$$

Find  $f_{bd}$ , Table 13 (a) Page 55, IS 800- 2007, for  $f_{cr,b} = 485.65 \text{ N/mm}^2$ ,  $\alpha_{LT} = 0.21$ ,  $f_y = 250 \text{ N/mm}^2$

Table 13(a) Design Bending Compressive Stress Corresponding to Lateral Buckling,  $f_{bd}$ ,  $\alpha_{LT} = 0.21$   
(Clause 8.2.2)

$f_{cr,b}$	$f_y$																		
	200	210	220	230	240	250	260	280	300	320	340	360	380	400	420	450	480	510	540
10 000	181.8	190.9	200	209.1	218.2	227.3	236.4	254.5	272.7	290.9	309.1	327.3	345.5	363.6	381.8	409.1	436.4	463.6	490.9
8 000	181.8	190.9	200	209.1	218.2	227.3	236.4	254.5	272.7	290.9	309.1	327.3	345.5	363.6	381.8	409.1	436.4	463.6	490.9
6 000	181.8	190.9	200	209.1	218.2	227.3	236.4	254.5	272.7	290.9	309.1	327.3	345.5	363.6	381.8	409.1	436.4	463.6	490.9
4 000	181.8	190.9	200	209.1	218.2	227.3	236.4	254.5	272.7	290.9	309.1	327.3	345.5	363.6	381.8	409.1	436.4	463.6	490.9
2 000	181.8	190.9	200	209.1	218.2	227.3	236.4	254.5	272.7	290.9	309.1	327.3	345.5	363.6	381.8	409.1	436.4	463.6	490.9
1 000	169.1	179.5	186	196.5	202.9	209.1	219.8	229.1	245.5	261.8	275.1	291.3	300.5	323.6	332.2	355.9	370.9	384.8	412.4
900	169.1	179.5	186	194.5	200.7	204.5	215.1	231.6	242.7	258.9	272	291.3	300.5	316.4	328.4	339.5	366.5	380.2	392.7
800	167.3	177.5	184	190.3	196.4	206.8	212.7	224	240	258.9	268.9	284.7	293.6	301.8	324.5	335.5	349.1	370.9	387.8
700	163.6	171.8	182	188.2	192	202.3	208	226.5	237.3	250.2	259.6	278.2	286.7	294.5	305.5	327.3	340.4	352.4	363.3
600	161.8	168	176	181.9	194.2	197.7	203.3	218.9	226.4	244.4	253.5	261.8	276.4	287.3	294	306.8	322.9	333.8	343.6
500	161.8	166.1	172	179.8	185.5	188.6	200.9	208.7	218.2	232.7	244.2	248.7	259.1	269.1	274.9	286.4	296.7	301.4	314.2
450	158.2	164.2	168	173.5	183.3	186.4	191.5	206.2	215.5	224	231.8	242.2	248.7	258.2	263.5	274.1	279.3	292.1	294.5
400	150.9	162.3	166	169.4	174.5	184.1	186.7	196	204.5	215.3	222.5	229.1	238.4	243.6	248.2	257.7	261.8	264.3	274.9

Through interpolation, the value of  $f_{bd} = 187.96 \text{ N/mm}^2$

$$M_d = \beta_b Z_p f_{bd}$$

Design bending strength or Moment of resistance,  $M_d = 1 \times 9.05 \times 10^6 \times 187.96 = 1701.6 \times 10^6 \text{ N mm} = 1701.6 \text{ kNm}$

$1701.6 \text{ kNm} > 839.4 \text{ kNm}$

It is safe.

- **Check for Shear resistance**

Page 59, CL 8.4, IS 800 2007

- **8.4 Shear**

The factored design shear force,  $V$ , in a beam due to external actions shall satisfy

$$V \leq V_d$$

where

$$\begin{aligned} V_d &= \text{design strength} \\ &= V_n / \gamma_{m0} \end{aligned}$$

where

$$\gamma_{m0} = \text{partial safety factor against shear failure (see 5.4.1).}$$

The nominal shear strength of a cross-section,  $V_n$ , may be governed by plastic shear resistance (see 8.4.1) or strength of the web as governed by shear buckling (see 8.4.2).

**8.4.1** The nominal plastic shear resistance under pure shear is given by:

$$V_n = V_p$$

where

$$V_p = \frac{A_v f_{yw}}{\sqrt{3}}$$

$A_v$  = shear area, and

$f_{yw}$  = yield strength of the web.

- **Check for Shear resistance**

Page 59, CL 8.4, IS 800 2007

- **8.4 Shear**

The factored design shear force,  $V$ , in a beam due to external actions shall satisfy

$$V \leq V_d$$

where

$$\begin{aligned} V_d &= \text{design strength} \\ &= V_n / \gamma_{m0} \end{aligned}$$

where

$$\gamma_{m0} = \text{partial safety factor against shear failure (see 5.4.1).}$$

The nominal shear strength of a cross-section,  $V_n$ , may be governed by plastic shear resistance (see 8.4.1) or strength of the web as governed by shear buckling (see 8.4.2).

**8.4.1** The nominal plastic shear resistance under pure shear is given by:

$$V_n = V_p$$

where

$$V_p = \frac{A_v f_{yw}}{\sqrt{3}}$$

$$\begin{aligned} A_v &= \text{shear area, and} \\ f_{yw} &= \text{yield strength of the web.} \end{aligned}$$

$$\text{Design shear strength } V_d = \frac{V_n}{\gamma_{m0}} = \frac{A_v f_{yw}}{\sqrt{3} \gamma_{m0}}$$

$$A_v \text{ is shear area} = h \times t_w = 568.6 \times 9.9 = 5629 \text{ mm}^2$$

$$V_d = \frac{5629 \times 250}{\sqrt{3} \times 1.1} = 738 \text{ kN} > 515.28 \text{ kN, it is safe.}$$

- **Check for web crippling**

- Web crippling causes local crushing failure of web due to large bearing stresses under reactions at supports or concentrated loads.
- This occurs due to stress concentration because of the bottle neck condition at the junction between flanges and web.
- It is due to the large localized bearing stress caused by the transfer of compression from relatively wide flange to narrow and thin web.

Use CL 8.7.4, Page 67, IS 800-2007

### 8.7.4 Bearing Stiffeners

Bearing stiffeners should be provided for webs where forces applied through a flange by loads or reactions exceeding the local capacity of the web at its connection to the flange,  $F_w$ , given by:

$$F_w = (b_1 + n_2)t_w f_{yw} / \gamma_{m0}$$

where

$b_1$  = stiff bearing length (see 8.7.1.3),

$n_2$  = length obtained by dispersion through the

flange to the web junction at a slope of 1 : 2.5 to the plane of the flange.

$t_w$  = thickness of the web, and

$f_{yw}$  = yield stress of the web.

Let us assume bearing length width,  $b_1 = 100$  mm,

$n_2 = 2.5 \times (\text{thickness of bottom plate} + \text{thickness of flange})$

$$= 2.5 \times (40 + 14.7) = 136.75 \text{ mm}$$

$$F_w = (100 + 136.75) \times 9.9 \times 250 / 1.1 = 532.68 \text{ kN} > 515.28 \text{ kN (Shear Force)}$$

- **Check for buckling of web**

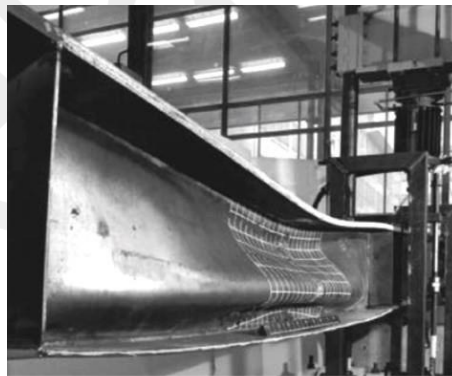
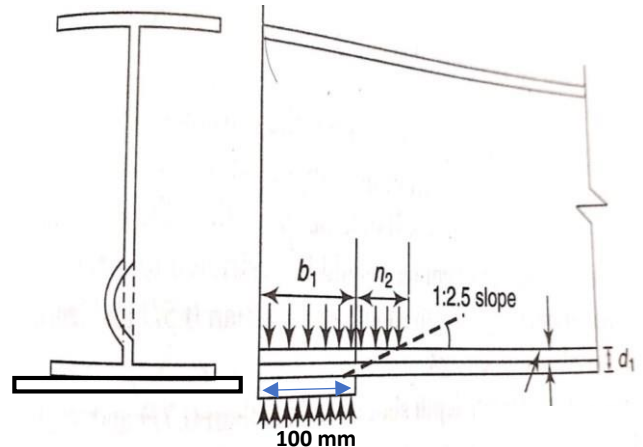
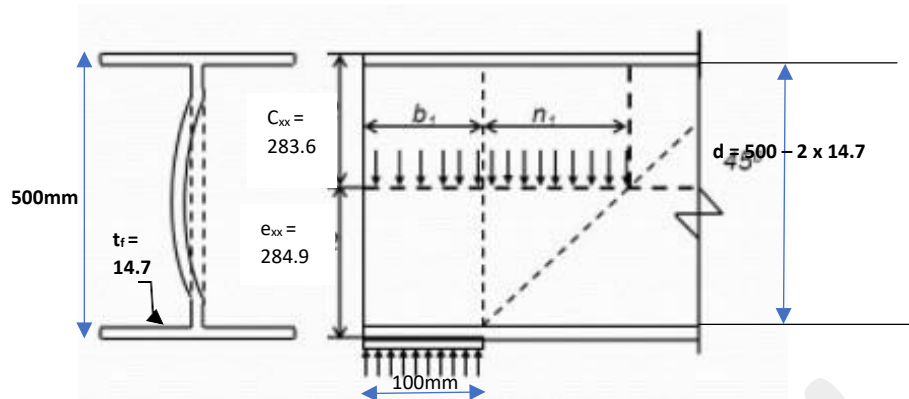


Figure 2. Vertical web buckling.

- The web of the beam is thin and can buckle under reactions and concentrated loads with the web behaving like a short column fixed at the flanges.
- The unsupported length between the fillet lines for I sections and the vertical distance between the flanges or flange angles in built up sections can buckle due to reactions or concentrated loads. This is called web buckling.



Buckling strength of web,  $F_{wb} = (b_1 + n_1) t_w f_{cd}$

Breadth of bearing stiffener,  $b_1 = 100$  mm,

Dispersion of  $45^\circ$  line at the mid depth of section  $n_1 = e_{xx} = 284.9$  mm,

Thickness of web,  $t_w = 9.9$  mm

To find design compressive stress  $f_{cd}$ , we need to calculate Slenderness ratio  $\lambda$

Slenderness ratio,  $\lambda = \frac{L_{eff}}{r_{min}}$

where  $L_{eff}$  is the effective length of the strut (compression) taken as,  $L_{eff} = 0.7 \times d$

where 'd' is the depth of the web portion (strut) between the flanges =  $500 - 2 \times 14.7 = 470.6$  mm

$r_{min} = r_{yy} = 95.7$  mm (girder)

$$\lambda = \frac{0.7 \times 470.6}{95.7} = 3.44$$

Since it is a built-up member it will come under buckling class "c" (IS 800 – 2007, Page 44, Table 10).

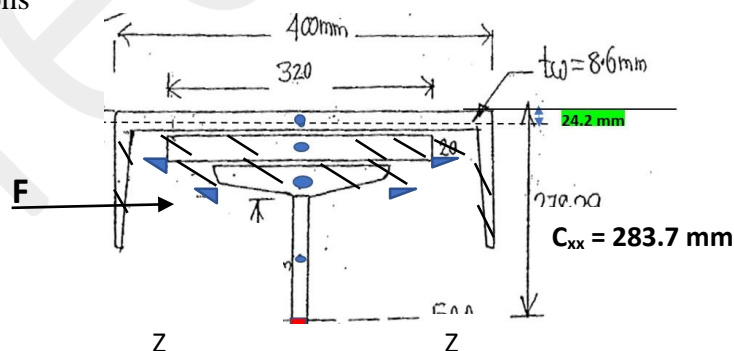
Since it is class "c", Use Table 9(c)

From Table 9 (c) Page 42 – IS 800 2007, for  $\lambda = 3.44$  we get design Compressive Stress,  $f_{cd} = 227$  N/mm<sup>2</sup>

Hence Buckling strength of web,  $F_{wb} = (100 + 284.9) \times 9.9 \times 227 = 865$  kN > 515.28 kN ( Shear Force)

### • Connections

Using welded connections



$$\text{Force at the junction} = F = \frac{V a \bar{y}}{I_z}$$

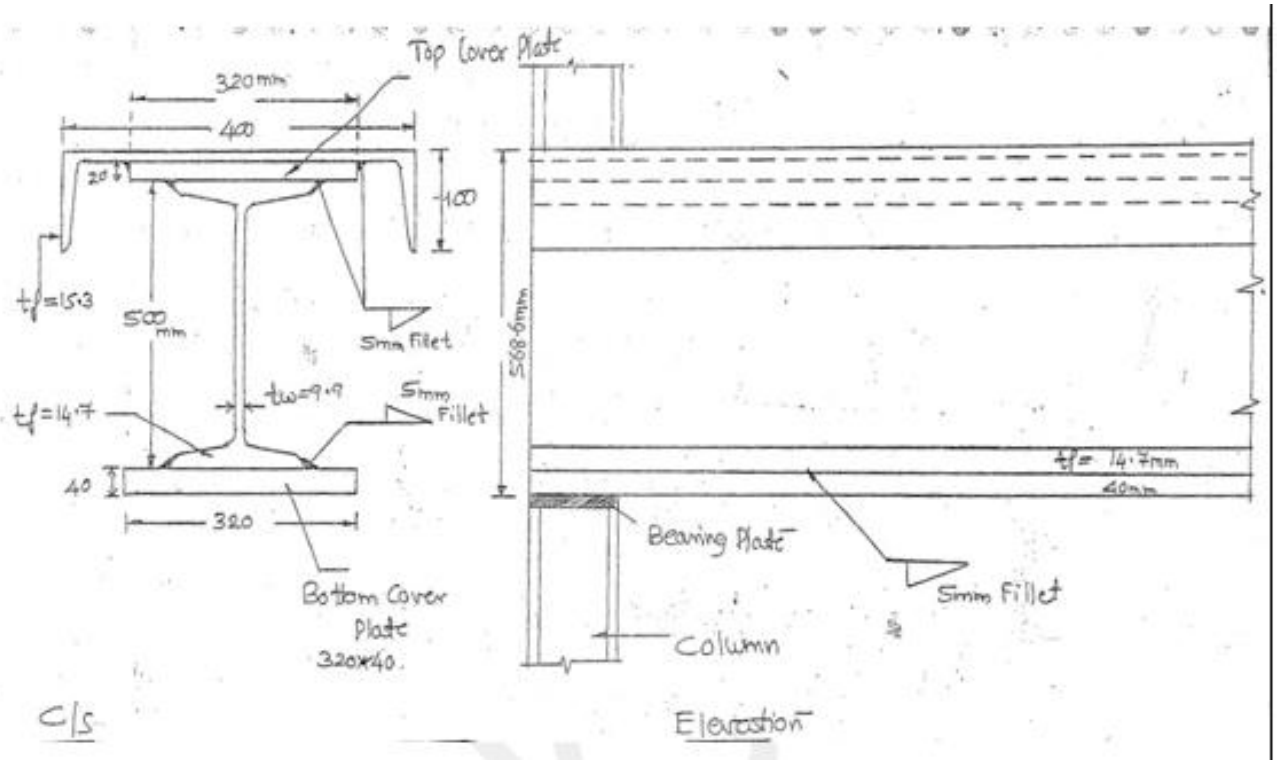
Where  $a \bar{y}$  is the area of shaded portion multiplied by centroidal distances measured from individual section

$$= 515.28 \times \frac{[6293 \times (283.7 - 24.2) + (320 \times 20 \times (283.7 - 8.6 - 10))]}{2301.9 \times 10^6} = 718.7 \text{ kN/m}$$



$$\text{Strength of weld} = 2 \times \frac{[0.7 \times s \times 1 \text{ mm} \times 410]}{\sqrt{3} \times 1.25}, s = 2.71 \text{ mm}$$

Provide 5 mm size weld



### Optional

Bracket connections for Gantry girder

$$\text{No. of bolts} = n = \sqrt{\frac{6M}{lpR}}$$

Where l = number of bolt lines = 4

Assume 20 mm diameter bolts

p is the pitch = 2.5 x diameter of bolt = 2.5 x 20 = 50 mm

R is the bolt value = 60.38 KN

Moment  $M = P \times e$

$P = \text{Max. SF in Gantry Girder} = 515.28 \text{ kN}$

Assume,  $e = 200 \text{ mm}$

$M = (515.28 \times 10^3) \times (200) = 103056 \times 10^3 \text{ N-mm}$

$$= n = \sqrt{\frac{6M}{lpR}} = \sqrt{\frac{6 \times 103056 \times 10^3}{4 \times 50 \times 515280}} = 8$$

Use 8 number of bolts for bracket connections